



Application of flux footprint equations from Kljun et al. (2015) to field eddy-covariance systems for footprint characteristics into flux network datasets

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Abstract. Gas fluxes passing through an eddy-covariance (EC) system's measurement volume reflect the outgassing rate of these molecules from an upwind area known as the "flux footprint". While sources/sinks of these molecules may be uniform over a flat field, their spatial contribution to the measured fluxes is not. Thus, understanding the contribution to measured fluxes and the spatial quantification of sources/sinks from the measured fluxes requires footprint analysis. Such analysis yields flux footprint characteristics, which commonly include upwind maximum footprint location, upwind fetch containing certain percentages of measured flux (70%, 80%, 90%), and the percent of flux from a user-defined upwind fetch of interest. These characteristics are included in the datasets of flux networks such as ChinaFlux, AmeriFlux, and FluxNet. Ideally, the characteristics are calculated in real-time and on-site by EC systems in the field, but this has often not been the case due to the calculations being computationally challenging. For field applications, this study develops the equations and algorithms for these characteristics from analytical crosswind-integrated flux footprint equations. The development shows that in-field computation is made feasible by the following means: using time-efficient algorithms, taking advantage of the nondimensional nature of the footprint equations of Kljun et al. (2015), implementing practical limits on numerical integration, and developing a differential-based estimation of boundary layer height for each EC interval. Accuracy of in-field calculations is maintained by the selection of footprint equations based on boundary-layer conditions and considerations of integration methods and computation techniques. This computational approach may also be applied to footprint analyses over complex terrain, nonuniform sources/sinks, or in cases where other footprint equations are used. The most popular application of footprint analysis is to optimize the EC sensor height for maximization of measured fluxes from an area of interest. This optimization using the nondimensional footprint equations is discussed, which leads to a practical



methodology. This work serves as a technical reference for users or developers of EasyFlux programs, widely used in
35 Campbell Scientific EC systems globally.

1 Introduction

An eddy-covariance system for flux measurements, including a gas analyzer (e.g., an infrared CO₂–H₂O analyzer) and three-dimensional (3D) sonic anemometer, is mounted at its measurement height z_m on a field tower (Munger et al., 2012). The gas analyzer and sonic anemometer are configured for their sensing surfaces to enclose the outmost boundary of the
40 “measurement volume” (see IRGASON in Fig. 1a), through which passive gas, sensible heat, and momentum fluxes are measured. These measured passive gas fluxes (e.g., CO₂) through the measurement volume are stochastically transferred by boundary-layer turbulent flows (Horst and Weil, 1992) from their sources or to their sinks over an area called the flux footprint. As such, atmospheric conditions and the spatial relation of the measurement volume to the sources/sinks determine
45 the molecular number of a measured passive gas flux from or to a particular unit area over the flux footprint field. In other words, the flux contribution varies spatially (Fig. 1a). This is the case even when the rate of source emission or sink absorption is spatially uniform (Hsieh et al., 2000). However, given that in common instances this rate may be spatially nonuniform, for practical cases over heterogeneous or sporadic sources/sinks, flux footprint equations are needed for evaluation of sources/sinks (Leclerc and Foken, 2014).

In a boundary-layer turbulent flow field, a flux footprint equation $f(x, y)$ is spatially defined in a wind coordinate system,
50 with x in a direction against streamwise wind, y horizontally across streamwise wind, and z orthogonal to x and y (Fig. 1 in Schmid, 1994). To easily relate a wind coordinate (x, y) to its ground location, the horizontal coordinate of a flux tower base is assigned as the origin $(0, 0)$ in both the wind and ground coordinate systems. In this way, given a wind direction in reference to the ground location of a flux tower, any location at (x, y) can be trigonometrically related to its ground location. In the wind coordinate system, $f(x, y)$ is understood to be a probability distribution of contributions (Kormann and Meixner,
55 2001) from the spatially uniform sources/sinks of a passive gas over a topographically flat field to its turbulent flux passing through the “measurement volume” of an eddy-covariance flux system. Thus, for uniform sources/sinks of a passive gas over a flat fetch, a footprint value at a particular ground location indicates a relative contribution of passive gas from this location to the measurement volume centered at $(0, 0, z)$, where z is the aerodynamic height equal to z_m minus d (zero-plane displacement height). The greater the footprint value, the more contribution from that location.

60 The flux footprint $f(x, y)$ can be integrated across wind (i.e., along y) to yield a crosswind-integrated flux footprint $f_y(x)$, resulting in a skewed bell-shape curve along the x -axis only (Fig. 1a). The curve represents a probability distribution. A common convention in literature is to present flux footprints for positive fluxes, where a gas is emitted from its upwind sources. However, footprints apply to negative fluxes with sinks as well. Regardless of the flux direction, all gas molecules in boundary-layer turbulent flows are transported through a stochastic process (Lumley and Panofsky, 1964; Horst, 1979),
65 and accordingly, the flux footprint curve for measured flux from upwind sources along the positive x (Fig. 1) should be

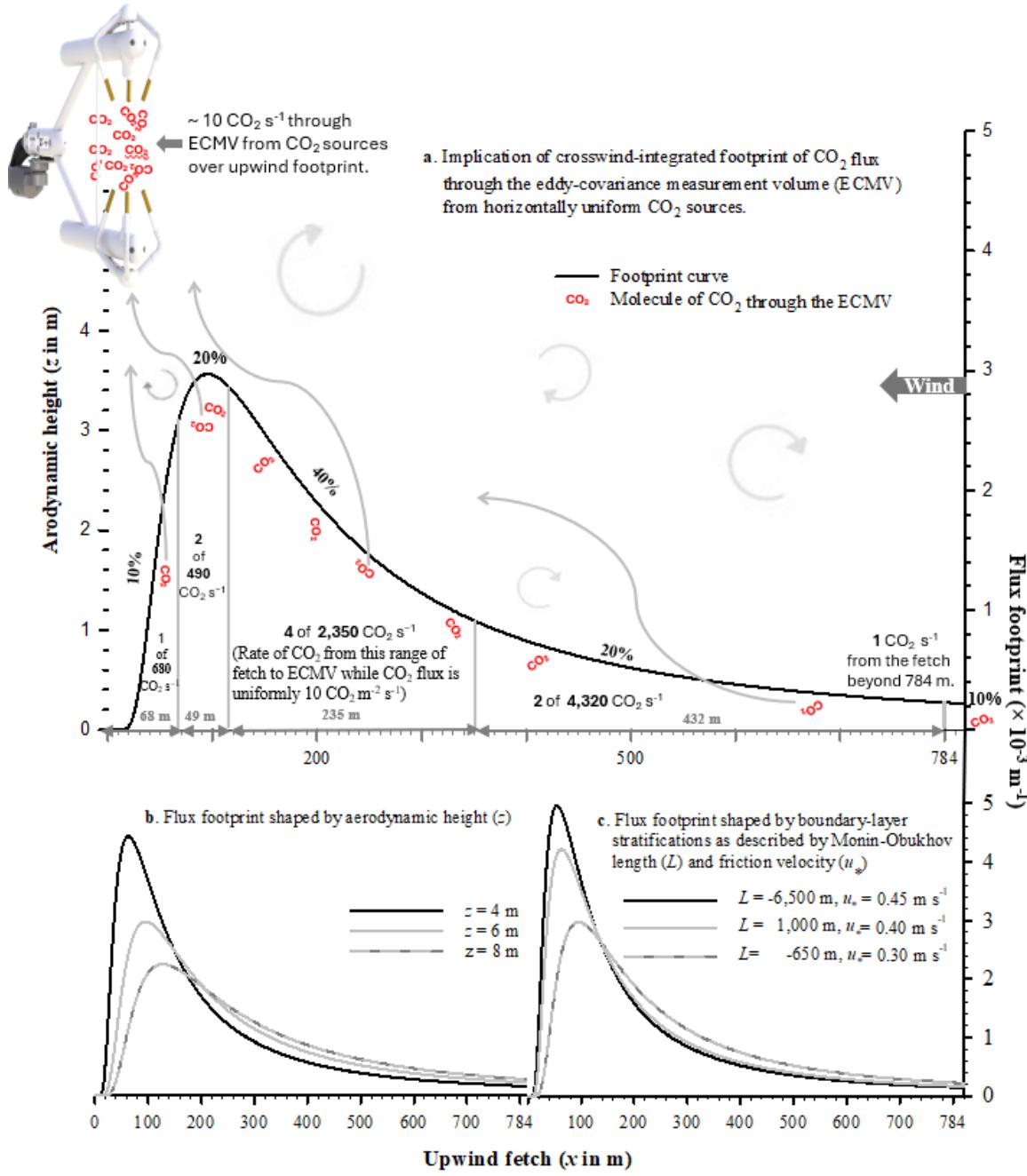


Figure 1: A flux footprint equation is displayed as a crosswind-integrated flux footprint, where the curve along the x -axis shows the contribution of molecules originating at a particular value of x and for all values of y . a. Crosswind-integrated flux footprint in a case where CO_2 flux sources are known to be uniform and emitted at a rate of $10 \text{ CO}_2 \text{ molecules m}^{-2} \text{ s}^{-1}$ under convective conditions. The shape of the curve is affected by changes in sensor aerodynamic height z (i.e., height of center of measurement volume minus the zero-plane displacement height (panel b) and by changes in atmospheric boundary-layer stability, as determined from the Monin-Obukhov length L and friction velocity u_* (panel c). All curves in this figure are computed using Eq. (22) in Kormann and Meixner (2001). For these curves, unless noted inside panels, z is 6 m, L is -650 m , wind speed is 5 m s^{-1} , u_* is 0.3 m s^{-1} , and the van Karman constant is 0.41.



symmetric with its counterpart for measured fluxes to downwind sinks along the negative x . This symmetry should be true
75 around the z -axis in the x and y domain, too (Schmid, 1997). Because of this symmetry and for simplification, only the cases
of upwind flux footprint for a positive flux from its upwind sources are conventionally presented in literature (e.g., Schmid,
2002). Such a conventional presentation is followed by this study for figures, equations, and algorithms.

As shown in Fig. 1 for upwind flux footprint curves, even in cases where gas flux over a vast flat field is uniform, the
flux footprint varies with upwind distance away from the measurement volume. It is also shaped by the aerodynamic height
80 of the measurement volume (Fig. 1b, Hsieh et al., 2000; Raupach et al., 1991) and by boundary-layer conditions related to
thermodynamic stratifications in air flows (Fig. 1c, Kormann and Meixner, 2001).

As a probability distribution, $f(x, y)$ can be used to derive a mean of passive gas sources/sinks $Q(x, y)$ over a 2-dimensional (2D) field (Snedecor and Cochran, 1989) because both are related to the flux through the measurement volume
85 $F(0, 0, z)$ (Kormann and Meixner, 2001):

$$85 \quad F(0, 0, z) = \int_{\mathfrak{R}} Q(x, y) f(x, y) dx dy \quad (1, \text{model})$$

where \mathfrak{R} denotes an integration domain. Indeed, $f(x, y)$ may be thought of as a transfer function of the gas flux of $Q(x, y)$
over an extended 2D field to the flux at the measurement volume $F(0, 0, z)$ (Kljun et al., 2015). Accordingly, although
developed based on horizontally uniform sources/sinks of a passive gas, $f(x, y)$ is also applicable to the description of the
90 transfer process of passive gas flux signals from nonuniform sources/sinks, represented by $Q(x, y)$ (see Chapter 8 in Leclerc
and Foken, 2014; Göckede et al., 2004).

The ultimate objective from a measured flux $F(0, 0, z)$ is to evaluate $Q(x, y)$ over the ecosystems targeted for
measurement. For horizontally nonuniform sources/sinks over flat terrain, $Q(x, y)$ varies with x and y . In this case, $f(x, y)$ is
imperative for $Q(x, y)$ to be evaluated from $F(0, 0, z)$, which is an advanced application of $f(x, y)$ still under development
95 (Leclerc and Foken, 2014). In cases where $Q(x, y)$ is constant for horizontally uniform sources/sinks of measured gas over
flat terrain, $F(0, 0, z)$ must be representative to this constant due to the right side of model (1) to be this constant because the
integration of $f(x, y)$ alone over its full domain is equal to a unit (Snedecor and Cochran, 1989). For most flux measurements,
this scenario is assumed, thus, for scenarios where $Q(x, y)$ is constant, $f(x, y)$ is less significant.

However, modern flux network datasets, most of which are from sites of assumed horizontally uniform
sources/sinks over flat terrain, report footprint characteristics including the upwind maximum footprint location
100 (*FETCH_MAX*, i.e., distance at which the sources/sinks contribute most to the measured flux) and the upwind fetch within
which the sources/sinks contribute a given percentage to the measured flux (e.g., *FETCH_70* for 70%, *FETCH_80* for 80%,
and *FETCH_90* for 90%). Additionally, EasyFlux outputs the interest fetch (*FP_FETCH_INTRST*, i.e., the integrated flux
contribution from a defined fetch of interest). These footprint characteristics are increasingly becoming essential variables in
many datasets from international networks (e.g., AsiaFlux, <https://www.asiaflux.net>; FLUXNET, <http://fluxnet.org>; and



105 ICOS, www.icos-infrastructure.eu), national networks (e.g., AmeriFlux, <http://ameriflux.lbl.gov> and ChinaFlux, <http://www.chinaflux.org>), regional networks (e.g., NYS Mesonet, <https://www.nysmesonet.org/>), and individual eddy-covariance flux stations. In these networks and stations, thousands of Campbell Scientific eddy-covariance flux systems have been deployed based on instruments such as the IRGASON (integrated open-path infrared CO₂–H₂O analyzer and 3D sonic anemometer), CPEC300 series (EC155 closed-path infrared CO₂–H₂O analyzer with CSAT3A), and TGA (Trace Gas Analyzer) with CSAT3B (Campbell Scientific Inc., UT, USA). Each of these systems is controlled and measured by a datalogger (e.g., CR6, CR1000X, or Granite9, Campbell Scientific Inc., UT, USA), which also processes, transfers, and stores data.

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115 Each datalogger operates a program from the EasyFlux series, which handles instructions for system control, field measurements, and data transfers (e.g., to FTP site or Campbell Cloud). And most importantly, the EasyFlux program processes raw high-frequency (e.g., up to 20 Hz) measurements into fully corrected fluxes every user-specified output interval (e.g., 30 min). Other required variables, including footprint characteristics from the analytical crosswind-integrated flux footprint equations of Kormann and Meixner (2001) or Kljun et al. (2004; 2015), are also output each interval. The recent implementation of the equations from Kljun et al. (2015) is a new update, as previously the equations from Kljun et al. (2004) were used. The applicability of this update is important because of its consideration of various boundary-layer 120 stabilities. Due to this advancement, EasyFlux series programs released hereafter are programmed Kljun et al. (2015) as its default option for flux footprint characteristics, although Kormann and Meixner (2001) for these is still available as an alternative.

125 The primary goal of this study is to develop efficient algorithms for applying Kljun et al. (2015) in a datalogger, thus allowing for in-field computations of footprint characteristics every output interval. And since the resulting algorithms have been implemented in recent versions of EasyFlux datalogger programs, this paper also serves as a reference for the users and developers of Campbell Scientific eddy-covariance flux stations who wish to know technical details about the flux footprint characteristic outputs. But first, to comprehend the algorithms related to Kljun et al. (2015), we briefly summarize the development of their flux footprint equations.

2 Brief the development of flux footprint equations by Kljun et al. (2015)

130 Using the backward Lagrangian stochastic particle dispersion model (LPDM-B), Kljun et al. (2015) simulated the flux footprint for a vast range of values for z , going between 1 and 1,000 m in boundary-layer conditions ranging from strongly convective through neutral to strongly stable, and a large range of values for roughness length z_0 , including values for sparse forest canopies (Fig. 1 in Kljun et al., 2015). The vast range in flux footprint sizes (e.g., up to 270 km for only 80% footprint) manifests that it is not practical for a limited number of analytical $f(x, y)$ equations to meet the needs for all boundary-layer 135 flow fields at field scales. However, if the variables in $f(x, y)$ are made dimensionless, $f(x, y)$ could be independent of the dimensions of boundary-layer flow fields.



Ideally, $f(x, y)$ contours for all flow fields converge to a single shape or narrow ensemble, regardless of the magnitude of the field dimensions or the boundary-layer conditions. Thus, the single shape may be regarded as dimensionless and applicable to any field size and in any condition of atmospheric stability. With this aim, Buckingham II dimensional analysis (Stull, 1988) is an approach of Kljun et al. (2015) to formulate the universal model for this contour. The data from the LPDM-B simulations include a vast range of boundary-layer flows, as characterized by the combinations of z , z_0 , and boundary-layer stabilities, and thus are a good source of statistical samples for determination of the model parameters.

2.1 Buckingham II dimensional analysis of flux footprint

In a case where \mathfrak{R} is infinitely small, model (1) can be written as

$$145 \quad F(0, 0, z) = f(x, y)Q(x, y)\Delta x\Delta y, \quad (2)$$

which is equivalent to

$$f(x, y) \frac{F(0, 0, z)}{Q(x, y)\Delta x\Delta y}, \quad (3)$$

where F and Q have the same units given in mass/molecules $\text{m}^{-2} \text{ s}^{-1}$. Accordingly, $f(x, y)$ has units of m^{-2} since that would be the units of $1/\Delta x\Delta y$ if x and y are in m

150 The flux footprint characteristics in AmeriFlux (2018) datasets include *FETCH_MAX*, *FETCH_70*, *FETCH_80*, and *FETCH_90*, which are all measured in terms of an upwind fetch in m. Within a fetch, the relative contribution to the measured gas flux from horizontally uniform sources/sinks of a passive gas over a flat field is an accumulation of $f(x, y)$ after integration across wind, defined as $f_y(x)$. For the computations of flux footprint characteristics, as addressed in this study, only $f_y(x)$ is needed, although $f(x, y)$ may still be desired for flux footprint maps in two dimensions (Kormann and Meixner, 155 2001; Kljun et al., 2004; 2015). If the independent dispersion of a passive gas across wind is described by $D(y)$, $f_y(x)$ forms a 2D flux footprint $f(x, y)$ given as (Horst and Weil, 1992):

$$160 \quad D(y)f_y(x) = f(x, y). \quad (4)$$

Although the explicit equation of $D(y)$ is omitted here, it is a probability distribution (Pasquill and Smith, 1983) whose integration over y is equal to a unit. Because $f_y(x)$ is not dependent on y , the integration of Eq. (4) with respect to y yields

$$f_y(x) = \int_{-\infty}^{\infty} f(x, y)dy \quad (5)$$

Thus $f_y(x)$ has the same dimension as $f(x, y)dy$, which is m^{-1} . Its dimension is fundamental to nondimensionalization of $f_y(x)$ using Buckingham II dimensional analysis (Stull, 1988).

2.2 Buckingham II dimensionless combinations

$f_y(x)$ is a function of upwind fetch (x in m), varying with z in m, mean wind speed $\bar{u}(z)$ in m s^{-1} , friction velocity u_* in



165 m s^{-1} , z_0 in m, and the planetary boundary layer height h in m. In sections 3 and 4 of Kljun et al. (2015), these dimensional variables are used for their Eq. (4) to (14) to formulate each dimensionless combination Π that will be used to nondimensionalize $f_y(x)$, as briefed below.

The first choice is z because both the extent and magnitude of the footprint are most strongly dependent on it (Hsieh et al., 2000). The higher the measurement volume at z , the farther the footprint stretches along the upwind fetch (Fig 1b).

170 Accordingly, the independent fetch variable x of $f_y(x)$ should be inversely nondimensionalized by z as combination Π_1 :

$$\Pi_1 = \frac{x}{z}. \quad (6)$$

Another effect is that the $f_y(x)$ curve on average has a lower value when z is higher and the footprint is stretched along the upwind fetch, (Schmid, 1997). Therefore, $f_y(x)$ should be positively nondimensionalized by z as combination Π_2 :

$$\Pi_2 = zf_y(x). \quad (7)$$

175 According to a common finding that turbulent fluxes decline approximately linearly through the planetary boundary layer from surface value to zero at h (e.g., Stull 1988), z and h can be nondimensionalized as combination Π_3 :

$$\Pi_3 = 1 - \frac{z}{h}. \quad (8)$$

As a transfer function in turbulent boundary layer flows, the flux footprint is directly affected by $\bar{u}(z)$, u_* , and z_0 . Well-known nondimensional wind shear ϕ_m explicitly and implicitly includes these three variables (Kaimal and Finnigan, 1994), 180 given by:

$$\phi_m = \frac{kz}{u_*} \frac{\partial \bar{u}(z)}{\partial z}, \quad (9)$$

where k is the von Karman constant (0.41). If the derivative is replaced by its approximation at z , ϕ_m becomes

$$\phi_m \approx \frac{kz}{u_*} \frac{\bar{u}(z) - \bar{u}(z_0 + d)}{z - (z_0 + d)} \approx k \frac{\bar{u}(z)}{u_*}. \quad (10)$$

From Kaimal and Finnigan (1994) and Högström (1996), ϕ_m is influenced by z_0 because:

$$185 \quad k \frac{\bar{u}(z)}{u_*} = \ln \left(\frac{z - d}{z_0} \right) - \psi_m \quad (11)$$

where ψ_m is the integrated form of nondimensional wind shear (Kaimal and Finnigan, 1994), which accounts for the effects of stability (z/L , where L is Monin-Obukhov length). If ϕ_m is thought of as nondimensional wind speed at z , reflecting a combined effect of u , u_* , and z_0 , it follows to use it as combination Π_4 :

$$\Pi_4 = k \frac{\bar{u}(z)}{u_*} \quad (12)$$

190 Unlike Kljun et al. (2004) which uses z_0 explicitly, combination Π_4 here includes z_0 implicitly.



2.3 Nondimensional upwind fetch (X^*)

The footprint of the measurement volume of an eddy covariance flux systems at a given z extends farther when h is higher (i.e., positively proportional to Π_3) and shrinks when wind is stronger (i.e., inversely proportional to Π_4). Accordingly, Kljun et al. (2015) formed nondimensional upwind fetch as:

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$$X^* = \Pi_1 \Pi_3 \Pi_4^{-1} = \frac{x}{z} \left(1 - \frac{z}{h}\right) \left(k \frac{\bar{u}(z)}{u_*}\right)^{-1} \quad (13)$$

2.4 Nondimensional crosswind-integrated flux footprint (F_y^*)

Because the integration of the flux footprint over its full range is always equal to 1, individual footprint values are on average lower when the footprint has a longer range, and higher when the footprint has a shorter one. Therefore, Π_3 and Π_4 interact inversely. Accordingly, the nondimensional crosswind-integrated flux footprint can be formulated as:

200
$$F_y^* = \Pi_2 \Pi_3^{-1} \Pi_4 = f_y(x) z \left(1 - \frac{z}{h}\right)^{-1} \left(k \frac{\bar{u}(z)}{u_*}\right). \quad (14)$$

Even when $f_y(x)$ extends to very long ranges as shown in Fig. 1 of Kljun et al. (2015), F_y^* versus X^* converges to an ensemble of nondimensionalized crosswind-integrated flux footprints very similar in curve shape, peak location, and fetch extent (see Fig. 2 of Kljun et al. 2015).

2.5 Formulation and parameterization for F_y^*

205 For a given range of boundary-layer stabilities, the convergence of F_y^* versus X^* to a narrow ensemble provides the basis to formulate a universal model fitted to the ensemble of LPDM-B results. Additionally, Kljun et al. (2015) chose to describe the relationship of F_y^* to X^* using the product of a power function of X^* and an exponential function of X^* (see their Fig. 2). The product formulates a universal model for the non-dimensional crosswind-integrated flux footprint:

$$F_y^*(X^*) = a (X^* - d_0)^b \exp\left(-\frac{c}{X^* - d_0}\right), \quad (15, \text{model})$$

210 where a , b , c , and d_0 are parameters, and the subscript 0 is used to avoid confusion between the fourth parameter and the zero-plane displacement height, conventionally denoted by a d in boundary-layer meteorology. Because model (15) is a probability distribution, its four parameters satisfy a constraint where the integral of $F_y^*(X^*)$ over the X^* domain must be unity:

$$\lim_{\delta_0 \rightarrow d_0} \int_{\delta_0}^{+\infty} F_y^*(X^*) dX^* = 1, \quad (16)$$

215 where δ_0 is the lower limit of integration. Using an alternative variable,



$$t = \frac{c}{X^* - d_0}, \quad (17)$$

the integral of the right side of model (15) can be related to the Gamma function (Nemes, 2010) as

$$ac^{b+1} \int_0^{+\infty} t^{-b-2} \exp(-t) dt = ac^{b+1} \Gamma(-b-1) = 1 \quad (18)$$

With this constraint, the parameters in model (15) were statistically estimated using the data from LPDM-B simulations after 220 nondimensionalization. With a set of estimated parameters, model (15) was developed into a non-dimensional crosswind-integrated flux footprint equation.

As shown in Fig. 2 of Kljun et al. (2015), this equation represents the flux footprint across all field scales, with model (15) shown as the universal framework. The goodness-of-fit of this single F_y^* equation for the ensemble of nondimensionalized flux footprints for all simulated measurement heights, stability conditions, and roughness lengths is 225 evidenced from the model performance metrics in Table 3 of Kljun et al. (2015). The fit can be improved even more if model parameters are optimized as two sets as shown in Table A1 of Kljun et al. (2015), each of which represent F_y^* under convective ($z/L < 0$) or neutral/stable ($z/L \geq 0$) boundary-layer conditions. Thus, a pair of equations are formulated as a set for F_y^* :

$$F_y^*(X^*) = \begin{cases} \frac{2.930}{(X^* + 0.107)^{2.285}} \exp\left(-\frac{2.127}{X^* + 0.107}\right) & \frac{z}{L} < 0 \\ \frac{1.472}{(X^* - 0.169)^{1.996}} \exp\left(-\frac{1.480}{X^* - 0.169}\right) & \frac{z}{L} \geq 0 \end{cases} \quad (19)$$

230 This set of analytical crosswind-integrated flux footprint equations are adopted into the EasyFlux series of programs.

3 Applications of nondimensional crosswind-integrated flux footprint equations

In the EasyFlux series, the nondimensional crosswind-integrated flux footprint equations for $F_y^*(X^*)$ as shown in Eq. (19) are adopted to estimate the footprint characteristics over a flat field with horizontally uniform sources/sinks of passive gases. For example, *FETCH_70* is found by integrating Eq. (19) from a starting limit to X_{70}^* , It is the upper integration limit that 235 results in a cumulative footprint probability of 0.7. X_{70}^* is converted to field scale units (e.g., meters) using Eq. (13). Similarly, *FETCH_80* and *FETCH_90* may be found. For *FT_FETCH_INTRST*, which is the percentage of measured flux attributable to the area within a user-defined fetch distance, *fetch_intrst*. Also, through Eq. (13), this field distance is converted to X_{intrst}^* , to which Eq. (19) is integrated from its starting limit, yielding *FT_FETCH_INTRST*.



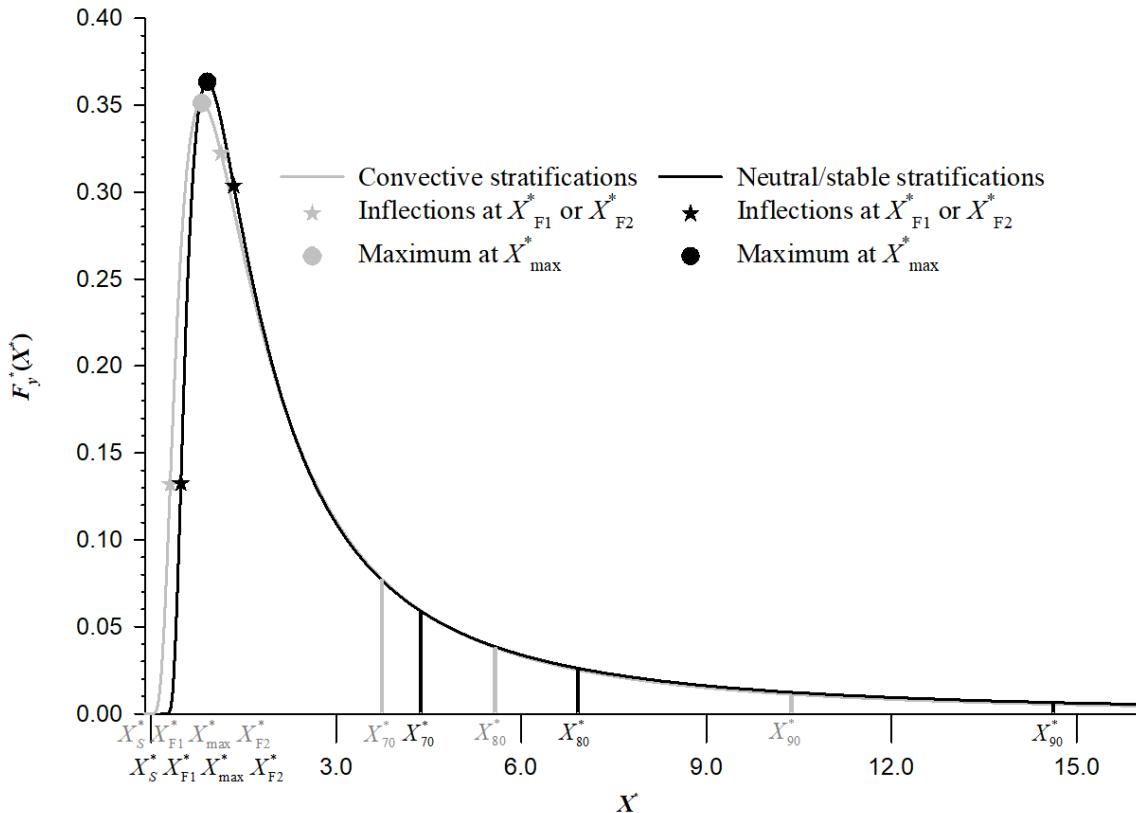
240 Since the integrations described above can be computationally intensive and difficult to do in the field, the following sections discuss approaches for calculating the footprint characteristics that eliminate or reduce in-field numerical integration.

3.1 *FETCH_MAX*

$F_y^*(X^*)$ yields skewed bell-shaped curves with respect to X^* (Fig. 2). The location of the maximum in terms of nondimensional upwind fetch X_{\max}^* is given by Eq. (20) of Kljun et al. 2015 (see derivation in Appendix A):

$$X_{\max}^* = d_0 - \frac{c}{b} \quad (20)$$

245 Its values for two ranges of atmospheric conditions are computed and shown in Table 1.



250 **Figure 2:** Nondimensional crosswind-integrated flux footprint $F_y^*(X^*)$ as a function of nondimensional upwind fetch X^* . A vertical bar at X_p^* , where subscript p indicates the percent of 70, 80, or 90, is the boundary at which the integration of $F_y^*(X^*)$, as shown in Eq. (19), from its starting limit X_S^* is equal to $p\%$. X_{\max}^* is the location of the maximum value of $F_y(X^*)$. X_{F1}^* and X_{F2}^* are the 1st and 2nd inflections on a $F_y(X^*)$ curve.



Table 1. The 1st inflection X_{F1}^* , maximum X_{\max}^* , and 2nd inflection X_{F2}^* on a nondimensional crosswind-integrated flux footprint curve $F_y^*(X)$ along the nondimensional upwind fetch X^* (z is aerodynamic height for flux measurements and L is Monin-Obukhov length.)

Atmospheric stability	X_{F1}^*	X_{\max}^*	X_{F2}^*
Convective $z/L < 0$	0.31026689 ^a	0.82385339	1.3374399
Neutral/stable $z/L \geq 0$	0.48210189	0.91048297	1.3388640

^a For each number, at least eight digits are kept for computations in single precision.

In Eq. (13), when X^* equals X_{\max}^* , x becomes *FETCH_MAX* and is given by:

$$255 \quad \text{FETCH_MAX} = kX_{\max}^* z \frac{\bar{u}(z)}{u_*} \left(1 - \frac{z}{h}\right)^{-1}. \quad (21)$$

Over an averaging interval (e.g., 30 min) in a field eddy-covariance flux system, $\bar{u}(z)$ and u_* are derived from its wind measurements, h can be either directly measured using a Lidar Ceilometer (e.g., SkyVue Pro, Campbell Scientific Inc., 2025) or alternatively estimated from its measurements using the algorithms in Appendix B, and z is the aerodynamic height which is calculated from z_m minus d , where d in a field eddy-covariance flux system can be entered by the system user as the first choice whereas it is automatically estimated inside the system from the height of canopies around the flux tower (Rosenberg et al., 1983; Oke, 1987; Kaimal and Finnigan, 1994). The sensor measurement height z_m and canopy height are also included among the station variables whose values are set by the system user onto an EasyFlux program before or while an eddy-covariance system is running (Campbell Scientific Inc., 2022).

3.2 FP_FETCH_INTRST

265 *FP_FETCH_INTRST* is the cumulative footprint probability within a specified upwind fetch, *fetch_intrst*. In EasyFlux series, *fetch_intrst* is one of the so-called station variables that are entered by a system user into the EasyFlux program before or while the station is running. Using Eq. (13), it can be used to compute its corresponding nondimensional form X_{intrst}^* . In this equation, at x equal to *fetch_intrst*, X^* is X_{intrst}^* and given by:

$$X_{\text{intrst}}^* = \frac{\text{fetch_intrst}}{z} \left(1 - \frac{z}{h}\right) \left(k \frac{\bar{u}(z)}{u_*}\right)^{-1}. \quad (22)$$

270 Accordingly, the footprint percentage of measured passive gas flux within *fetch_intrst* around a flux tower is an integration of $F_y^*(X^*)$ with respect to X^* from its starting limit X_s^* (near the flux tower) to X_{intrst}^* :

$$FP_FETCH_INTRST = 100 \int_{X_s^*}^{X_{\text{intrst}}^*} F_y^*(X^*) dX^*, \quad (23)$$



where X_s^* can be set to $d_0 + 10^{-7}$ because $F_y^*(X^*)$ is valid only when X^* is greater than d_0 and, in the 7th significant digit after decimal (i.e., single precision expression), $d_0 + 10^{-7}$ is a number near the smallest precise number greater than d_0 .

275 **3.2.1 Computation considerations**

As explicitly expressed in Eqs. (19) and (23), $F_y^*(X^*)$ may be numerically integrated at discrete, incremental values of X^* , starting at X_s^* and increasing until X_{inrst}^* is reached. The accuracy of numerical integration depends on the resolution of increments in X^* . The smaller the increment, the higher resolution and greater accuracy the result (Burden et. al., 2016). However, for a given range of X^* , smaller increments increase the number of iterations for numerical integration, which adds 280 more computational loads to a microprocessor of an in-field computation module, such as a CR6 or CR1000X datalogger (Campbell Scientific Inc. UT, USA), commonly used inside of an eddy-covariance flux system. Thus, the integration for field applications must be optimized to ensure integration accuracy with a minimized computational load.

As shown in Fig. 2 for Eq. (19), $F_y^*(X^*)$ has four identified turning points: the starting limit at X_s^* , the maximum at X_{max}^* , and the bilateral inflection points at F_{F1}^* and X_{F2}^* as well. Since the flux footprint curve changes more rapidly around 285 these points, the accuracy of numerical integration would include less uncertainties if these points were located at the boundaries for segments or zones of integration (Burden et. al., 2016). Additionally, as compared to the right tail of the flux footprint curve, the curve across the three zones from X_s^* to F_{F1}^* , F_{F1}^* to X_{max}^* , and X_{max}^* to X_{F2}^* is steeper in slope or changes more dramatically. Since one of the two end points of each zone is an inflection point, these zones will be called inflection zones for the purposes of this study.

290 Within a zone, an increment for numerical integration should be small for greater accuracy, and X_s^* , F_{F1}^* , X_{max}^* , and X_{F2}^* are used as boundaries. Beyond X_{F2}^* , an integration increment may be extended, creating lower resolution but reducing computations. As previously noted, X_s^* is defined based on d_0 , which is a parameter in model (15) and used as a constant in Eq. (19). X_{max}^* is given by Eq. (20). Derived in Appendix A, the first inflection point is located at

$$X_{F1}^* = \frac{-c \left[1 - (1-b)^{-\frac{1}{2}} \right]}{b} + d_0 \quad (24)$$

295 and the second one is located at

$$X_{F2}^* = \frac{-c \left[1 + (1-b)^{-\frac{1}{2}} \right]}{b} + d_0 \quad (25)$$

For F_{F1}^* and X_{F2}^* , their computed values are shown in Table 1, and their locations on the footprint curve are shown in Fig. 2.



3.2.2 Algorithm

As discussed previously, Eq. (22) is used to nondimensionalize the upwind fetch of interest to X_{intrst}^* and the numerical integration of Eq. (23) to X_{intrst}^* yields the footprint fraction of measured flux sourced from the upwind fetch of interest. For 300 the integration, we choose the Composite Simpson's Rule (Burden et al., 2016). Depending on X_{intrst}^* , the integration can cover, from left to right as shown in Fig. 2, one to three full inflection zones unless $X_{\text{intrst}}^* < X_{F1}^*$. To reduce the uncertainties in the accuracy over the range of integration, $F_y^*(X^*)$ in Eq. (23) should be integrated at higher resolution with smaller increments over these zones, but beyond them (i.e., $X^* > X_{F2}^*$), the integration can be performed at lower resolution with an 305 increased size of increments.

One thousand increments of X^* within an inflection zone are considered adequate, with increments smaller than 5.14×10^{-4} (Table 2). For the inflection zone in which X_{intrst}^* is located, only the portion of the zone up to X_{intrst}^* is numerically integrated in the field. In this way, the computational load for *FP_FETCH_INTRST* can be controlled to its minimum so that in-situ outputs are possible while the full inflection zones within X_{intrst}^* are numerically integrated in a lab at high resolution as 310 shown in Table 2.

Within an inflection zone that X_{intrst}^* is located and up to the value of X_{intrst}^* the resolution in Table 2 is used. For inflection zones lower than the zone in which X_{intrst}^* is located, no integration is required, as calculated constants for cumulative footprint in each zone may be used (see Table 2). In cases where X_{intrst}^* is beyond the second inflection point, the integration increment between X_{F2}^* and X_{intrst}^* is determined by $(X_{\text{intrst}}^* - X_{F2}^*)/n$ where n is typically 1000 or less in order to 315 limit the time needed for computation. The number of increments n for the lower resolution depends on the computation capacity of the microprocessor in a field eddy-covariance flux system. It should be noted that the numerical integration calculations also rely on inputs from real-time eddy-covariance sensor measurements, because as shown by Eq. (22), the evaluation of X_{intrst}^* requires $\bar{u}(z)$, u_* , and h , which are calculated from in-field high-frequency measurements.

3.2.3 Example

320 Given that an upwind fetch of interest is 500 m, z equals 5 m, and the conditions for scenario 3 in Table 1 of Kljun et al (2015) ($L = -650$ m, $u_* = 0.30$ m s $^{-1}$, and $h = 1,200$ m) with $\bar{u}(5)$ equal to 4.00 m s $^{-1}$, X_{intrst}^* from Eq. (22) is 18.216463. Because X_{intrst}^* is greater than X_{F2}^* (Table 1), using Eq. (23), the flux footprint percentage within this upwind fetch to the flux tower can be evaluated by:

$$FP_FETCH_INTRST = 100 \int_{X_S^*}^{X_{F2}^*} F_y^*(X^*) dX^* + 100 \int_{X_{F2}^*}^{X_{\text{intrst}}^*} F_y^*(X^*) dX^* \quad (26)$$



Table 2. The flux footprint values in inflection zones (Fig. 2) and their cumulative flux footprint values from the starting value of nondimensional upwind fetch X^* , denoted by $X_S^* = d_0 + 10^{-7}$ where d_0 is a parameter of nondimensional crosswind-integrated flux footprint equation $F_y^*(X^*)$ as shown in Model (15) and Eq. (19). The flux footprint values are numerically integrated for each inflection zone using the Composite Simpson's Rule on a $F_y^*(X^*)$ curve (Fig. 2). X_{F1}^* is the 1st inflection ahead of X_{\max}^* (the maximum location) and X_{F2}^* is the 2nd inflection behind X_{\max}^* . L is Monin-Obukhov length and z is the aerodynamic height for measurements.

Atmospheric stability	Zone	Ending	X_{F1}^*	X_{\max}^*	X_{F2}^*
		Range	$X_S^* \sim X_{F1}^*$	$X_{F1}^* \sim X_{\max}^*$	$X_{\max}^* \sim X_{F2}^*$
Convective $z/L < 0$	Integration resolution		4.1726699×10^{-4} ^a	5.1358650×10^{-4}	
	Zone footprint %		1.1321783	14.605774	16.788529
	Cumulative footprint % ^b			15.737952	32.526482
Neutral/stable $z/L \geq 0$	Integration resolution		3.1310199×10^{-4}	4.2838107×10^{-4}	
	Zone footprint %		0.87452260	12.597578	14.546249
	Cumulative footprint %			13.472100	28.018350

^a For each number, eight digits are kept for significance of computations in single precision at least.

^b Cumulative footprint in each zone column is the integration of $F_y^*(X^*)$ from $d_0 + 10^{-7}$ to the ending boundary of this zone.

The 1st term on the right side of this equation was evaluated in Table 2 as a constant 32.526482 %. For field applications, Eq. (26) for this case can be expressed as:

$$FP_FETCH_INTRST = 32.526482\% + 100 \int_{X_{F2}^*}^{18.216463} F_y^*(X^*) dX^* \quad (27)$$

Thus, in the field, numerical integration is required only on the 2nd term on the right side. If n is 1,000, the size of increments in X^* for numerical integration is given by:

$$\Delta X^* = \frac{X_{\text{intrst}}^* - X_{F2}^*}{n} = \frac{18.216463 - X_{F2}^*}{1000} = 1.6879023 \times 10^{-2} \quad (28)$$

Appendix C shows the algorithms used for numerical integrations of Eqs. (27) and (28) using the Composite Simpson's Rule. In this example, after integration FP_FETCH_INTRST is found to be 94.86%. By using the calculated cumulative footprints in Table 2 for full inflection zones to the left of X_{intrst}^* , by beginning numerical integration in the zone in which X_{intrst}^* is located, and by only performing integration up to the value of X_{intrst}^* , the number of iterations is confined to be no greater than n (Appendix C).



3.3 *FETCH_70*, *FETCH_80*, and *FETCH_90*

FETCH_p, where suffix p can be 70, 80, or 90, is the conversion of the corresponding nondimensional form X_p^* to its field scale, or dimensional form, using Eq. (13). Therefore, similarly to the derivation of Eq. (21), the conversion of *FETCH_p* from X_p^* is given by:

$$350 \quad \text{FETCH}_p = kX_p^* z \frac{\bar{u}(z)}{u_*} \left(1 - \frac{z}{h}\right)^{-1}. \quad (29)$$

Since the values of k , z , $\bar{u}(z)$, u_* , and h can be acquired in the same way as for Eq. (21), X_p^* is additionally needed. Whereas X_p^* is the nondimensional upwind fetch within which the horizontal uniform sources of gas flux contribute $p\%$ of the measured flux, it is mathematically expressed as:

$$100 \int_{X_S^*}^{X_p^*} F_y^*(X^*) dX^* = p. \quad (30)$$

355 The data in Table 2 indicate $X_p^* > X_{F2}^*$ while $p \geq 32.53\%$. For p equal to 70, 80, or 90, the left side of Eq. (30) can therefore be expressed in two terms:

$$100 \int_{X_S^*}^p F_y^*(X^*) dX^* = 100 \int_{X_S^*}^{X_{F2}^*} F_y^*(X^*) dX^* + 100 \int_{X_{F2}^*}^{X_p^*} F_y^*(X^*) dX^*, \quad (31)$$

where the 1st term on the right side of this equation is a constant, given in Table 2 for the two ranges of boundary-layer stabilities. If this constant is denoted by $P_{X_{F2}^*}$, the range over which to integrate can be made smaller, beginning at X_{F2}^* , 360 instead of X_S^* , and extending to X_p^* :

$$P_{X_{F2}^*} + 100 \int_{X_{F2}^*}^{X_p^*} F_y^*(X^*) dX^* = p. \quad (32)$$

In the integration term of this equation, 25 may be used as an upper limit for X^* because 90% of fluxes will always be below 25, which is also why Fig. 2 of Kljun et al. (2015) only extends to 25. Thus, an increment in X^* can be evaluated by

$$\Delta X^* = \frac{25 - X_{F2}^*}{1000} \quad (33)$$

365 To find X_p^* from Eq. (32), X_p^* needs to be expressed explicitly.



Although an integer m rarely exists that satisfies $X_{F2}^* + m\Delta X^* = X_p^*$, it can easily hold that $X_{F2}^* + m\Delta X^* \approx X_p^*$, from which an explicit equation for X_p^* can be derived. In this case, the inequality $X_{F2}^* + m\Delta X^* > X_p^*$ can lead to:

$$p_{X_{F2}^*} + 100 \int_{X_{F2}^*}^{X_{F2}^* + m\Delta X^*} F_y^*(X^*) dX^* = p_+ \geq p, \quad (34)$$

where m can be between 1 and 1000 as long as $X^* < 25$ (Eq. 33). Since Eq. (34) integrates to an upper limit that is slightly greater than X_p^* and the result is slightly greater than p , we should also find the upper limit and result that is barely less than 370 X_p^* and p , respectively. This limit must be $X_{F2}^* + (m-1)\Delta X^* < X_p^*$, which yields:

$$p_{X_{F2}^*} + 100 \int_{X_{F2}^*}^{X_{F2}^* + (m-1)\Delta X^*} F_y^*(X^*) dX^* = p_- \leq p. \quad (35)$$

Now X_p^* is a value bounded by $X_{F2}^* + (m-1)\Delta X^*$ and $X_{F2}^* + m\Delta X^*$. In the process of numerical integration, the values of p_+ , p_- , and m can be easily identified (Appendix C). The following section shows how Eqs. (32) to (35) may be used to find a 375 solution to X_p^* .

3.3.1 Solution to X_p^*

Equation (34) minus (32) leads to:

$$100 \int_{X_p^*}^{X_{F2}^* + m\Delta X^*} F_y^*(X^*) dX^* = p_+ - p. \quad (36)$$

The Intermediate Value Theorem reforms this equation as

$$380 \quad 100(X_{F2}^* + m\Delta X^* - X_p^*) F_y^*(X_\xi^*) = p_+ - p, \quad (37)$$

where X_ξ^* is an intermediate value in the range from X_p^* to $X_{F2}^* + m\Delta X^*$ and makes $F_y^*(X_\xi^*)$ equal to the average of $F_y^*(X^*)$ over the range. Similarly, Eq. (32) minus (35) leads to:

$$100[X_p^* - X_{F2}^* - (m-1)\Delta X^*] F_y^*(X_\zeta^*) = p - p_-, \quad (38)$$

where X_ζ^* is an intermediate value in the range from $X_{F2}^* + (m-1)\Delta X^*$ to X_p^* and makes $F_y^*(X_\zeta^*)$ equal to the average of 385 $F_y^*(X^*)$ over this range. Because both X_ξ^* and X_ζ^* are very close, in fact within a range as small as the size of ΔX^* , and whereas $F_y^*(X^*)$ is a continuous function and both $F_y^*(X_\xi^*)$ and $F_y^*(X_\zeta^*)$ can be considered almost equal, their ratio tends to be 1. As a result, the ratio of Eq. (37) to (38) leads to:



$$\frac{X_{F2}^* + m\Delta X^* - X_p^*}{X_p^* - X_{F2}^* - (m-1)\Delta X^*} \approx \frac{p_+ - p}{p - p_-} . \quad (39)$$

If this equation is solved for X_p^* , the result is an interpolation equation:

$$390 \quad X_p^* \approx X_{F2}^* + \left(m - \frac{p_+ - p}{p_+ - p_-} \right) \Delta X^* \quad (40)$$

Now X_p^* may be calculated, and its result used in Eq. (29) to calculate *FETCH_p*.

3.3.2 Example

In order to acquire *FETCH_70* for the same conditions as described in section 3.2.3, we use numerical integration as shown in Eqs. (34) and (35) (see Appendix C for application of integration) to find the inputs needed for the interpolation 395 in Eq. (40), which results in X_{70}^* (i.e., X_p^* at subscript $p = 70$). Given the value of X_{F2}^* from Table 1, ΔX^* can be computed from Eq. (33) to be 2.3662560×10^{-2} . At $m = 102$, $p_+ = 70.114313$ from Eq. (34), and $p_- = 69.868805$ from Eq. (35). Next, X_{70}^* can be computed from Eq. (40) as

$$X_{70}^* \approx X_{F2}^* + \left(m - \frac{p_+ - 70}{p_+ - p_-} \right) \Delta X^* = 3.7400033 \quad (41)$$

Using this value, Eq. (29) generates the following result:

$$400 \quad \text{FETCH_70} = kX_{70}^* z \frac{\bar{u}(z)}{u_*} \left(1 - \frac{z}{h} \right)^{-1} = 102.65 \text{ m} \quad (42)$$

This example illustrates that instead of extensive numerical integrations in the field, Eqs. (34) and (35) may be solved beforehand for X_{70}^* , X_{80}^* , and X_{90}^* (Table 3) due to X_p^* being independent of field measurements. Then, these values, along with field measurements, may be used in Eq. (29) to find their final field scale values.

Table 3 Nondimensional upwind fetch X_p^* , where subscript p indicates 70, 80, or 90. At a nondimensional scale, a $p\%$ portion of the 405 measured flux is contributed by its footprint area within X_p^* , assuming the sources/sinks of passive gas are uniform over a flat field. (z is aerodynamic height for measurements and L is Monin-Obukhov length.)

Atmospheric stability	X_{70}^*	X_{80}^*	X_{90}^*
Convective $z/L < 0$	3.7400033 ^a	5.5734341	10.371083
Neutral/stable $z/L \geq 0$	4.3702906	6.9142010	14.612024

^a For each number, at least eight digits are kept for significance of computations in single precision.



4 Discussion

This study details the application of Kljun et al.'s. (2015) flux footprint equations (Eq. 19) into eddy-covariance flux systems so that footprint characteristics of measured flux can be computed every interval of flux data output in the field. 410 These computed flux footprint characteristics are those required by datasets documented in AmeriFlux (2018) and adopted by regional, national, and international flux networks (e.g., NYS Mesonet, ChinaFlux, and FluxNet). Previously, these characteristics have been evaluated only through computationally laborious numerical integration (Kormann and Meixner, 2001; Kljun et al., 2002; 2025), not suitable for the limited computation capacity typically found in field computer 415 processors. Therefore, the development in this study focuses on field computation-saving methodologies, now adopted into the EasyFlux series programs (Campbell Scientific Inc, 2022). Indeed, the nondimensional forms of fetch (Eq. 13) and footprint equations (Eq. 19) from Eqs. (6) to (14) in Kljun et al. (2015) make field computation-saving methodologies applicable (Appendix C).

It should be noted that the naming and selected footprint variables in this study were chosen to be in conformity with the 420 2018 AmeriFlux data variable format. Furthermore, data precision was optimized to match the computation precision inside field eddy-covariance flux systems. And lastly, the algorithm for the estimation of planetary boundary layer height h from measured variables in an eddy-covariance flux system was a major consideration for this study, and details concerning it are described in Appendix B. Beyond the immediate applications in this study, the developed equations found herein and in 425 Kljun et al. (2015) have important implications for the optimization of eddy-covariance measurement height in order to maximize the proportion of measured flux from the footprint area of most interest. In the following sections, more discussion is given to the merits of Eq. (19), the expression of variables, the optimization of data precision, the algorithm for h , and more applications of equations.

4.1 Merits of Kljun et al.'s. (2015) flux footprint equations

Computing flux footprint characteristics such as $FETCH_p$, where subscript p is 70, 80, or 90, and FP_FETCH_INTRST , 430 has typically been challenging in the field because approaches like Hsieh et al (2000) or Kormann and Meixner (2001) require computationally laborious numerical integrations. The use of nondimensional flux footprint equations found in Kljun et al. (2015) can reduce or fully avoid numerical integration. For $FETCH_p$, given Table 3, only an analytical equation (Eq. 29) is needed, requiring a simple algebraic calculation. For FP_FETCH_INTRST , given Table 2, Eqs. (22), and (26), the numerical integration is reduced to a fractional zone, as shown in Fig 2, from one turning point to X_{intrst}^* .

435 4.2 Variable expressions

The names of some variables in this study, such as FP_FETCH_INTRST , $FETCH_MAX$, $FETCH_70$, $FETCH_80$, and $FETCH_90$ and FP_FETCH_INTRST , are lengthy but used for the sake of consistency with the data variable format documented in AmeriFlux (2018).

4.3 Data precision



440 Unlike a desktop or laptop computer, a computation module like a CR6 or CR1000X datalogger is smaller in size, lower
in power consumption, and more durable in rugged environment conditions, plus it has multiple functionalities for control,
measurement, communication, computation, and data storage. As such, the performance of a microprocessor inside the
computation module is optimized for all mentioned functionalities through balancing its size, power consumption, and
durability. For optimization, single precision is used for data processing inside the microprocessor. Accordingly, eight
445 significant digits in single precision are kept for the data shown in the three data tables and Eqs. (27), (28), and (41) of this
paper. However, it should be noted that these data were computed from Eq (19) using double precision processing on a
desktop computer, even though the precision of data from Eq. (19) is hardly warranted because it depends on the precision of
equation parameters that were statistically estimated (section 2.5). Nonetheless, considering Eq. (19) as an exact equation,
this study carefully warrants the accuracy of numerical integrations and the precision of data for computations of flux
450 footprint characteristics.

4.4 Algorithm for planetary boundary layer height

455 The planetary boundary-layer height (h) is one of the required variables in the flux footprint equations of Kljun et al.
(2015), (see Eqs. 21, 22, 29, and 42). Unlike other variables, it is not directly measured or commonly computed from
measured data in eddy-covariance systems. And while it can be directly measured using a ceilometer (e.g., SkyVUE Pro,
Campbell Scientific Inc. 2025), it is often cost prohibitive. As shown in Eqs. (B1) and (B3), h is theoretically related to other
commonly measured variables in an eddy-covariance system. Since the main body of this paper focuses on computations of
flux footprint characteristics, this algorithm is developed in Appendix B, although the algorithm is still a key finding from
this study.

4.5 Applications of equations developed in this study

460 Equation (33) is used to calculate a ΔX^* value for use in Eqs. (34) to (40). In reference to Fig. 2 of Kljun et al. (2015), an
assumed top limit of 25 for X^* is used for this calculation. Between X_s^* and 25, the integration of Eq. (19) is equal to 96.50%
and 93.98% for convective and neutral/stable atmospheric stabilities, respectively. Accordingly, in Eqs. (34) to (40), the p
value should be $\leq 96.50\%$ under convective atmospheric stability or $\leq 93.98\%$ under neutral/stable atmospheric stability. In
the case that p is above these ranges, the value from Eq. (33) is still applicable because it has a higher resolution than if 25
465 were replaced with a higher value in Eq. (33). Alternatively, ΔX^* also can be extended or narrowed, depending on the
accuracy required for $FETCH_p$.

Although Eq. (40) was developed for cases of p equal to 70, 80, or 90 to compute $FETCH_{70}$, $FETCH_{80}$, or $FETCH_{90}$, it can be used for any p value. In reference to the cumulative footprint values in Table 2, X_{F2}^* in this equation can be
replaced with X_{F1}^* , X_{\max}^* , or X_s^* , depending on the p value under different atmospheric stabilities. In reference to the
470 integration resolution values also in Table 2, the integration resolution for the corresponding zone can be used as a value of
 ΔX^* .



For example (Table 2), X_{F2}^* in Eq. (40) should be replaced with X_s^* under convective atmospheric stability if $p < 1.1321783$ or under neutral/stable atmospheric stability if $p < 0.87452260$, in which case ΔX^* would be 4.1726699×10^{-4} and 3.1310199×10^{-4} , respectively.

475 4.6 Optimize measurement height

Perhaps the most significant application of flux footprint equations is the optimization of measurement height of eddy-covariance sensors (i.e., z_m , the height of measurement volume center above the ground), such as a sonic anemometer and a gas analyzer (Horst and Weil, 1994). Over a flat field with uniform flux sources/sinks, the higher the measurement volume, the farther the flux footprint can extend away from the flux tower whereas the lower the measurement volume, the closer the flux footprint converges to the flux tower (Fig. 1). Over a flat field, z_m is generally optimized for an expected percentage p of measured flux from a given upwind fetch or for maximization of measured flux contribution from a targeted area covered by an ecosystem of interest.

4.6.1 Optimization of z_m for an expected percentage of measured flux within a given upwind fetch

Given an upwind fetch, $FETCH_p$, from which a $p\%$ measured flux is expected, Eq. (29) can be rearranged and solved 485 for the optimized height, z_{mp} :

$$z_{mp} = \frac{u_* h FETCH_p}{k h \bar{u}(z) X_p^* + u_* FETCH_p} + d, \quad (43)$$

where z_{mp} is chosen for the prevailing atmospheric stability at a site (e.g., the case in section 3.2.3) since the height of system sensors is typically inconvenient to adjust after installation. As an example, given $FETCH_{90}$ to be 500 m, atmospheric stability as described in section 3.2.3, d to be 0.25 m, and X_{90}^* from Table 3 for $z/L < 0$, Eq. (43) generates z_{mp} to be 9.00 m.

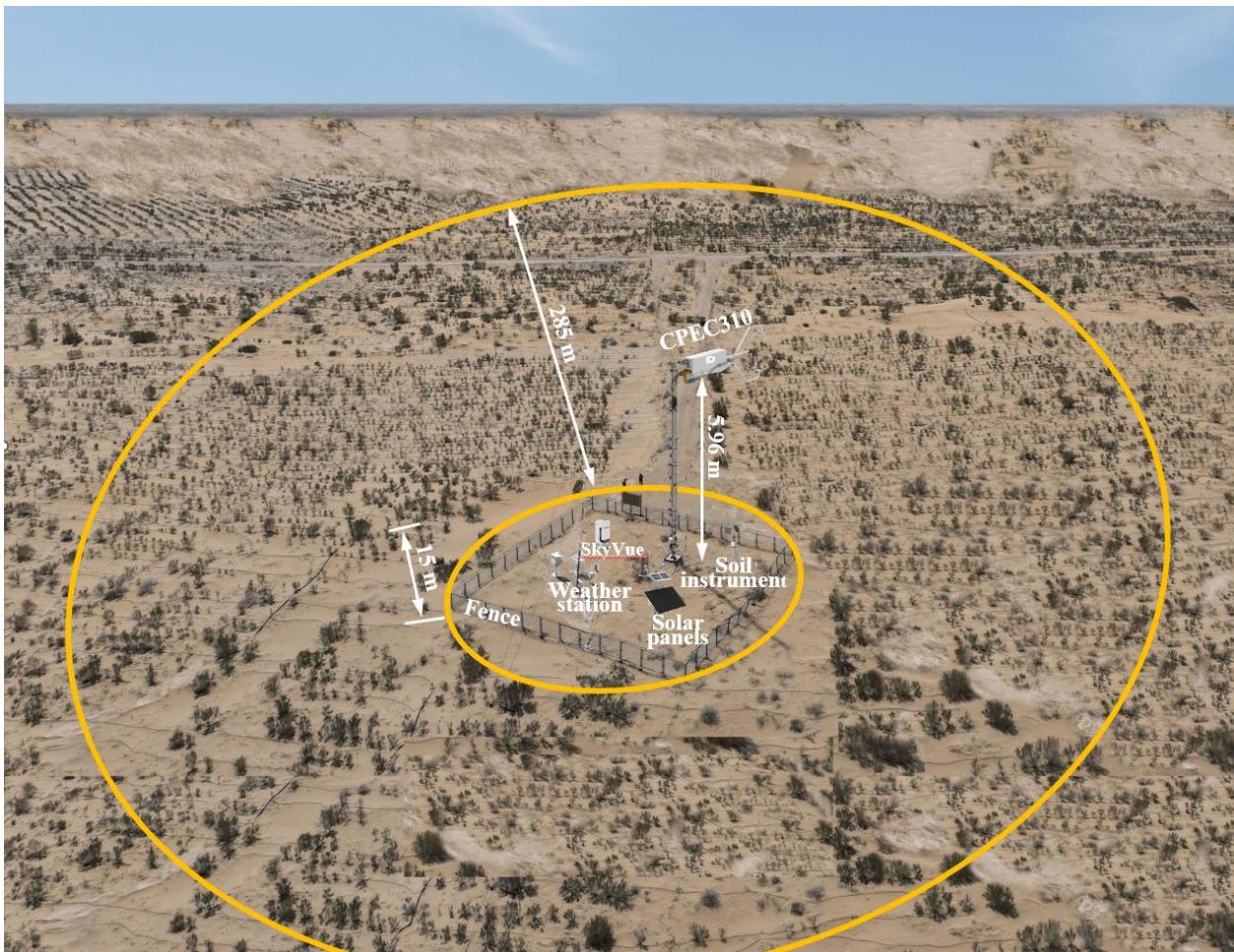
490 Equation (43) describes z_{mp} essentially as a function of $FETCH_p$ because the other aerodynamic variables in the equation are given for a site's prevailing atmospheric stability. Using X_{70}^* , X_{80}^* , and X_{90}^* values from Table 3, z_{m70} , z_{m80} , and z_{m90} corresponding to $FETCH_{70}$, $FETCH_{80}$, and $FETCH_{90}$ under the prevailing atmospheric stability can be generated from Eq. (43). For any percentage of measured gas flux from a given upwind fetch $FETCH_p$, X_p^* value needed by Eq. (43) can be numerically computed from Eq. (40).

495 4.6.2 Optimization of z_m to maximize measured flux from the targeted area of an ecosystem of interest

A common practice in eddy-covariance system installation is to optimize z_m . The optimization aims to maximize the measured fluxes from the targeted area covered by an ecosystem of interest while minimizing the influence of fluxes from the area covered by undesirable ecosystems outside the target area and from the fenced area disturbed by station facilities (e.g., supporting structure), instruments for other micrometeorological variables (e.g., radiation, soil moisture, and rain), and solar panels for power supply to the system (Fig. 3). The degree of influence depends on many factors such as the type and area of undesirable ecosystems, the size of fenced areas, the volume of facilities, the surface of solar panels. Although the fluxes from the undesirable ecosystems and the disturbed area will unavoidably contaminate the measurement



volume, it can be minimized through the optimization of z_m (Kormann and Meixner, 2001). Depending on surface roughness mostly accounted by d along with $\bar{u}(z)$ gradient and atmospheric boundary-layer stability accounted by h (Rebmann et al., 505 2018), for the optimization of z_m , the fraction of measured flux from the targeted area can be evaluated from flux footprint equations.



510 **Figure 3:** A drone view of field situation in a case that a closed-path eddy-covariance system (i.e., CPEC310, Campbell Scientific, Inc., UT, USA) was used to measure the CO₂ and H₂O fluxes over *Haloxylon ammodendron* plantation near bare sand land (farther top area) in Minqin, China. As view, the installation height of CPEC310 sensors should be optimized to maximize the measured fluxes from the area inside the external and outside the inner circles while minimizing the measured fluxes from both the bare sand land area outside the external circle and the fenced area covered by flux tower, weather station, solar panel, ceilometer (SkyVue), and instrument for soil moisture and soil temperature. This view is not scaled.

515

The targeted area is generally in the shape of an annulus centered at the flux tower (Fig. 3) with its external radius R , outside which the area is covered by undesirable ecosystems, and with its inner radius r , inside of which a fenced portion is



the disturbed area. The optimization of z_m is to find a height at which the portion of measured flux from the annulus footprint area is maximized. This portion denoted by F_{pa} , where subscript a indicates annulus, is given from Eq. (19) as

520
$$F_{pa} = \int_{X_r^*}^{X_R^*} F_y^*(X^*) dX^* \quad (44)$$

where X_r^* is the nondimensional fetch corresponding to the inner annulus radius r at field scale and is given from Eq. (13) as

$$X_r^* = \frac{r}{z} \left(1 - \frac{z}{h}\right) \left(k \frac{u(z)}{u_*}\right)^{-1} \quad (45)$$

and X_R^* is the nondimensional fetch corresponding to the outer annulus radius R at field scale, given also from Eq. (13) as

$$X_R^* = \frac{R}{z} \left(1 - \frac{z}{h}\right) \left(k \frac{u(z)}{u_*}\right)^{-1}. \quad (46)$$

525 Given r and R under a specified boundary-layer condition, both X_r^* and X_R^* change with z . For a prevailing boundary layer condition with given h , u_* , and $u(z)$, F_{pa} is a function of z [i.e., $F_{pa}(z)$], with the integration limits of Eq. (44) varying with z . The z value at which $F_{pa}(z)$ reaches its maximum is the optimum aerodynamic height, denoted by z_{\max} . This height is the solution to

$$\frac{dF_{pa}(z)}{dz} \bigg|_{z=z_{\max}} = 0 \quad (47)$$

530 At z_{\max} , the measurement volume of an eddy-covariance system will receive the largest possible portion of fluxes from the annulus area of interest. For the solution of z_{\max} , we find the derivative of $F_p(z)$ with respect to z :

$$\frac{dF_{pa}(z)}{dz} = -\frac{u_*}{ku(z)z^2} \left[RF_y^*(X_R^*) - rF_y^*(X_r^*) \right] \quad (48)$$

Given r and R values, X_r^* and X_R^* can be computed from Eqs. (45) and (46), respectively. In reference to Eq. (19), an analytical solution to z_{\max} for Eq. (47) from Eq. (48) is unavailable, but it can be found graphically, as shown in Fig. 4 for a 535 case of $r = 15$ m and $R = 300$ under the same boundary-layer conditions as in section 3.2.3. The result is accurate to within a centimeter, plenty for sensor installation. In Fig. 4, the $dF_{pa}(z)/dz$ curve crosses the z -axis at z_{\max} , which is 5.71 m. At z_{\max} , $F_{pa}(z)$ exactly reaches its maximum of 84.42% (see dashed line in Fig. 4). Given that d is 0.25 m, z_m can be optimized as 5.96 m (i.e., $z_m = z_{\max} + d$). This optimization methodology was developed by the authors while specifying installation of eddy-covariance sensors at Moorefield, Wellfleet, and Benkelman in Nebraska, USA.

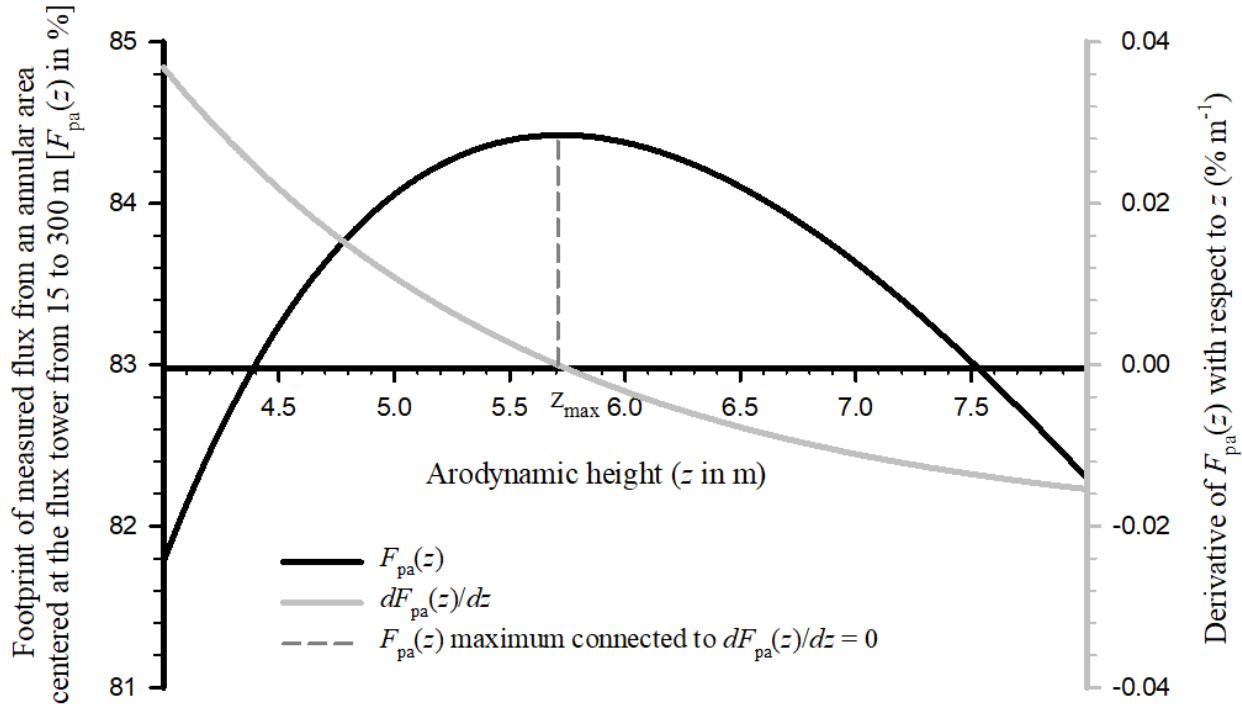


Figure 4: Graphical optimization of the aerodynamic height of the eddy-covariance measurement volume z_{\max} to maximize the portion of measured flux from the annular area centered at the flux tower (e.g., from its inner radius 15 to its external radius 300 m). $F_{\text{pa}}(z)$ values are 545 computed from Eq. (19) for $z/L < 0$, where L is Monin-Obukhov length, and from Eqs. (44) to (46). Values for $dF_{\text{pa}}(z)/dz$ are computed from Eq (48). The optimized aerodynamic height z_{\max} is 5.71 m when $F_{\text{pa}}(z)$ reaches its maximum of 84.42% and where its derivative with respect to z is zero (see dashed line). Given the zero-plane displacement height d to be 0.25 m, the measurement height z_m can be optimized as 5.96 m (i.e., $z_m = z_{\max} + d$). Wind speed is 4.00 m s⁻¹, friction velocity is 0.30 ms⁻¹, and planetary boundary layer height is 1200 m.

550 5 Summary remarks

Flux datasets are increasingly requiring the inclusion of flux footprint fetch characteristics, specifically the upwind maximum footprint location and the upwind percentage fetches (AmeriFlux, 2018). An additional flux footprint fetch characteristic included in many ChinaFlux datasets is the percentage of measured fluxes attributable to an area of interest. In a field eddy-covariance flux system, these characteristics are ideally evaluated simultaneously with the computations of the 555 flux data every interval of these data output (e.g., 30 min). In order for such evaluations to be time-efficient inside the microprocessor of a field datalogger, time-saving algorithms that retain accuracy through every step are developed from the well-accepted flux footprint equations of Kljun et al. (2015) (i.e., Eq. 19).

As a merit of Kljun et al. (2015), the upwind maximum footprint location, inflection locations, and upwind percentage fetches from their flux footprint equations are in the nondimensional domain, are invariant (Fig. 2, Tables 1 and 3), and can 560 be precisely computed beforehand in a laboratory. Similarly, by using analytical Eqs. (21) and (29), the maximum footprint location and upwind percentage fetches can be converted from their non-dimensional data in Table 3 to field scale units and



then stored inside the microprocessor for immediate use (Appendix C), thus avoiding the use of numerical integration in the field. And finally, because of this merit, the data in Table 3 also reduces the computation load for the interest footprint to a limited amount less than an inflection zone (Eq. 27).

565 The accuracy of the computed footprint characteristics is considered through the division of the footprint equation curve into four inflection zones for integration (Fig. 2, Tables 1 and 2). According to the comments in Appendix A of Kljun et al. (2015), better accuracy in the flux footprint characteristics leads to us adopting the equations with parameters from their Table A1 for convection and neutral/stable atmospheric boundary layer stabilities (Eq. 19, Fig. 2), instead of using their universal flux footprint Eqs. (14) and (17) of Kljun et al. (2015). Where possible, eight significant digits of data (Tables 1, 2, 570 and 3; Eqs. 27, 28, and 41) are kept for all computations at single precision, which is an additional consideration to warrant the accuracy in the flux footprint characteristics.

As shown in all application equations in this study (e.g., Eq. 21), the planetary boundary-layer height is needed as a scaling variable for flux footprint equations of Kljun et al. (2015), but it is not commonly directly measured with field eddy-covariance systems. For this variable to be acquired every data output interval from other variables measured by eddy- 575 covariance flux systems in the field, the applicable algorithm is developed in Appendix B.

As shown in Model (15), nondimensional upwind fetch (X^*) is the independent variable of flux footprint equations. An explicit expression for this fetch or for nondimensional upwind percentage fetch is not available. Thus, a numerical equation for nondimensional upwind percentage fetch is theoretically derived (Eqs. 29 to 40) and a conversion into field scale is shown.

580 Our discussions go beyond the focus of this study for the most practical and significant application of flux footprint equations in eddy-covariance flux measurements, that is to optimize the installation height of eddy-covariance sensors. Optimization means 1) to ensure an expected percentage of measured flux from a targeted upwind fetch and 2) to maximize the contribution of measured fluxes from the footprint area of interest. The methodology for this optimization is additionally 585 discussed (Figs. 3 and 4, Eqs. 43 to 48). With this addition, this study more fully documents the common applications of Kljun et al. (2015) to field eddy-covariance flux systems. This document is intended to be a reference source for flux footprint equation applications, especially for users and developers of EasyFlux series programs found in many Campbell Scientific eddy-covariance flux systems globally.

Code and data availability:

590 The program code related to the methods and algorithms that were developed in this manuscript is available from <https://doi.org/10.5281/zenodo.18143076> under (CC-BY-4.0) license, as are input data to produce the plots for all the simulations presented in this paper (Zhou and Chen, 2025).

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Appendix A: Maximum and inflection locations on the crosswind-integrated footprint curve

At the maximum of the nondimensional crosswind-integrated footprint $F_y^*(X^*)$, its 1st order derivative with respect to nondimensional upwind fetch X^* should satisfy

$$\left. \frac{dF_y^*(X^*)}{dX^*} \right|_{X^* = X_{\max}^*} = 0, \quad (\text{A1})$$

600 where X_{\max}^* is its maximum location. From Model (15), this 1st order derivative is given by

$$\frac{dF_y^*(X^*)}{dX^*} = a \left[b(X^* - d_0)^{b-1} + c(X^* - d_0)^{b-2} \right] \exp\left(-\frac{c}{X^* - d_0}\right). \quad (\text{A2})$$

Following Eq. (A1), setting this equal to zero leads to:

$$b(X_{\max}^* - d_0)^{b-1} + c(X_{\max}^* - d_0)^{b-2} = 0, \quad (\text{A3})$$

and solving for X_{\max}^* leads to Eq. (20).

605 At an inflection point of $F_y^*(X^*)$, its 2nd order derivative with respect to X^* must satisfy

$$\left. \frac{d}{dX^*} \left(\frac{dF_y^*(X^*)}{dX^*} \right) \right|_{X^* = X_I^*} = 0, \quad (\text{A4})$$

where X_I^* is the nondimensional upwind inflection location on the curve of $F_y^*(X^*)$ and its subscript I indicates inflection.

This subscript can be F1 or F2 for the 1st or 2nd inflection locations (Fig. 2). Equation (A4) is a further derivative of Eq. (A2), given by:

$$610 \quad \left. \frac{d}{dX^*} \left(\frac{dF_y^*(X^*)}{dX^*} \right) \right|_{X^* = X_I^*} = a \exp\left(-\frac{c}{X^* - d_0}\right) \left[b(b-1)(X^* - d_0)^{b-2} + 2c(b-1)(X^* - d_0)^{b-3} + c^2(X^* - d_0)^{b-4} \right]. \quad (\text{A5})$$

To satisfy Eq. (A4),

$$b(X_I^* - d_0)^2 + 2c(X_I^* - d_0) + \frac{c^2}{b-1} = 0. \quad (\text{A6})$$

The solutions to X_I^* from this equation are the two inflection locations in terms of nondimensional upwind fetch X_{F1}^* and X_{F2}^* given in Eqs. (24) and (25), respectively, and shown in Fig. 2.



615 **Appendix B: Estimation of planetary boundary layer height from measured variables in eddy-covariance flux systems**

In order to compute the footprint characteristics using equations of Kljun et al. (2015), the planetary boundary layer height (h) is required by Eqs. (21), (22), (29), (42), (45), and (46). Fortunately, it may be estimated using commonly measured variables in eddy-covariance systems. Appendix B in Kljun et al. (2015) summarizes the equations for h under 620 different atmospheric boundary-layer stratifications and recommends theoretical equations of h for use in eddy-covariance flux systems.

B1 Equations of h for use in eddy-covariance flux systems

For neutral to stable conditions, Kljun et al. (2015) summarized four equations. One is the primary equation, while the other three are the simplified versions for extreme cases of free atmosphere or strongly stable boundary-layer conditions.

625 The primary equation is an interpolation formula proposed by Nieuwstadt (1981):

$$h = \frac{L}{3.8} \left(\sqrt{1 + 2.28 \frac{u_*}{fL}} - 1 \right), \quad (\text{B1})$$

where L is Monin-Obukhov length, u_* is friction velocity, and f is the Coriolis parameter. In eddy-covariance flux systems, mean values of L and u_* (Rebmann et al., 2012) are computed every data output interval (e.g., 30 min), while f can be computed at any time from (Wallace and Hobbs, 2006)

630 $f = 2\Omega \sin \phi , \quad (\text{B2})$

where Ω is the angular velocity of Earth's rotation (7.2924621×10^{-5} rad s $^{-1}$) and ϕ is the latitude of an eddy-covariance flux station. As a station variable, ϕ is entered by a user into an eddy-covariance flux system before or while an EasyFlux series program is running.

635 For convective atmospheric conditions, an equation explicit to h is not available, however its differential equation with respect to time t is defined by Batchvarova and Cryning (1991) as

$$\frac{dh}{dt} \left(\frac{\gamma h^2}{(1+2A)h - 2BkL} + \frac{Cu_*^2 T}{g(1+A)h - gBkL} \right) = \overline{w' \Theta_0} , \quad (\text{B3})$$

where γ is dry adiabatic lapse rate (commonly 9.8×10^{-3} K m $^{-1}$), A , B , and C are parameters, k is the von Karman constant (0.41), g is acceleration due to gravity (9.81 m s $^{-2}$ at sea level), w' is vertical wind fluctuation, Θ' is potential air temperature fluctuation, and $\overline{w' \Theta_0}$ is the covariance of w' with Θ_0 , which drives the sensible heat flux over the interface between 640 ecosystems and the atmosphere. Over this interface, $\overline{w' \Theta_0}$ can be substituted with the covariance of w' with air temperature



(T), denoted by $\overline{wT'}$, where T' is air temperature fluctuation. This covariance is available in eddy-covariance flux systems. At present, an exact solution to h from Eq. (B3) is not available, but a numerical solution may be expressed as a divided difference form.

B2 The divided difference form of h terms in Eq. (B3)

645 In eddy-covariance flux systems, the aerodynamic and thermodynamic variables used for Eq. (B3), such as u_* , L , T , and $\overline{wT'}$, are computed from measured data averaged over a data output interval, denoted by Δt . As such, h can be derived only on a temporal scale of Δt . Given h_b to be a h value at the beginning of Δt , and h_e at the end, the derivative term can be expressed as

$$650 \quad \frac{dh}{dt} = \frac{h_e - h_b}{\Delta t}, \quad (\text{B4})$$

where, under continuous measurements, h_b over current Δt is h_e over a previous one. While the boundary layer is developing, h_b and h_e are rarely equal, and over a short period of Δt the change from h_b to h_e can be reasonably assumed to be linear. Accordingly, a h value can be approximated from

$$h = \frac{h_e + h_b}{2}, \quad (\text{B5})$$

655 Apparently, h value can be acquired if h_e value is estimated at the end of current Δt . In Eq. (B3), substitution of dh/dt and h with their corresponding divided difference forms (i.e., Eqs. B4 and B5) leads to

$$\begin{aligned} & \frac{\gamma(1+A)}{8\Delta t} h_e^4 + \frac{\gamma}{4\Delta t} ((1+A)h_b - BkL)h_e^3 - \left\{ \frac{\gamma BkL}{4\Delta t} h_b - \frac{1+2A}{2} \left[\frac{Cu_*^2 T}{\Delta t g} - (1+A) \frac{\overline{wT'}}{2} \right] \right\} h_e^2 \\ & - \left\{ \frac{\gamma(1+A)}{4\Delta t} h_b^3 - \frac{\gamma BkL}{4\Delta t} h_b^2 + \frac{(1+A)(1+2A)}{2} \overline{wT'_0} h_b - BkL \left[(3+4A) \frac{\overline{wT'}}{2} - \frac{2Cu_*^2 T}{\Delta t g} \right] \right\} h_e \\ & = \frac{\gamma(1+A)}{8\Delta t} h_b^4 - \frac{\gamma BkL}{4\Delta t} h_b^3 + \frac{1+2A}{2} \left[\frac{Cu_*^2 T}{\Delta t g} + (1+A) \frac{\overline{wT'}}{2} \right] h_b^2 - BkL \left[(3+4A) \frac{\overline{wT'}}{2} + \frac{2Cu_*^2 T}{\Delta t g} \right] h_b + 2(BkL)^2 \overline{wT'}. \end{aligned} \quad (\text{B6})$$

After the parameters A , B , and C in this equation are replaced with their corresponding values 0.2, 2.5, and 8.0 from Appendix B in Kljun et al. (2015), the equation becomes



$$\begin{aligned}
 0 = & 0.15 \frac{\gamma}{\Delta t} h_e^4 + \frac{\gamma}{\Delta t} (0.3h_b - 0.625kL) h_e^3 - \left(0.625 \frac{\gamma k L}{\Delta t} h_b - 5.6 \frac{u_*^2 T}{\Delta t g} + 0.42 \overline{w' T'} \right) h_e^2 \\
 & - \left[0.3 \frac{\gamma}{\Delta t} h_b^3 - 0.625 \frac{\gamma k L}{\Delta t} h_b^2 + 0.84 \overline{w' T'} h_b + k L \left(40.0 \frac{u_*^2 T}{\Delta t g} - 4.75 \overline{w' T'} \right) \right] h_e \\
 & - 0.15 \frac{\gamma}{\Delta t} h_b^4 + 0.625 \frac{\gamma k L}{\Delta t} h_b^3 - \left(5.6 \frac{u_*^2 T}{\Delta t g} + 0.42 \overline{w' T'} \right) h_b^2 + k L \left(40.0 \frac{u_*^2 T}{\Delta t g} + 4.75 \overline{w' T'} \right) h_b - 12.5 k^2 L^2 \overline{w' T'}.
 \end{aligned} \tag{B7}$$

Inside this equation, the only unknown variable is h_e , and since the equation is its 4th order polynomial, there are four possible solutions. One of the positive root values $h_b \in [h_b \pm \delta]$, where $\delta > 0$, must be the solution to h_e . Unfortunately, an explicit solution to h_e from this equation is not available, so a numerical method must be used.

B3 Numerical solution to h_e

If h_e on the right side of Eq. (B7) is replaced with h_x and represents a value $h_b \in [h_b \pm \delta]$, and 0 on the left side is replaced with $f(h_x)$ but still equals zero in the case of $h_x = h_e$, then $f(h_x)$ is a continuous differentiable function with a non-zero 1st order derivative. Therefore, the Newton-Raphson numerical method is applicable to the solution of $f(h_x)$ at zero for h_e (Burden et al., 2016).

Suppose that $f(h_e) \in C^4 [h \pm \delta]$ and let $h_b \in [h_b \pm \delta]$ be an initial approximation to h_e value such that $f'(h_x)|_{h_x=h_0} \neq 0$

and $|h_e - h_b|$ is sufficiently “small”. Then, the 2nd order Taylor polynomial for $f(h_x)$ about h_b is given by:

$$f(h_x) = f(h_b) + (h_x - h_b) f'(h_b) + \frac{(h_x - h_b)^2}{2} f''(h_\xi), \tag{B8}$$

where h_ξ lies between h_b and h_x . Since $f(h_e) = 0$, this equation becomes

$$f(h_b) + (h_e - h_b) f'(h_b) + \frac{(h_e - h_b)^2}{2} f''(h_\xi) = 0. \tag{B9}$$

Because h_ξ is unknown, h_e cannot be resolved from this equation, but after the 2nd order term with $f''(\xi)$ is dropped, Eq. (B9)

is commonly written as

$$h_e \approx h_b - \frac{f(h_b)}{f'(h_b)}. \tag{B10}$$

The right side of this equation is the h_x -intercept of the tangent line of $f(h_x)$ at $[h_b, f(h_b)]$. The value of this intercept can be denoted by h_{e_1} and is a first approximation for h_e . In Eq. (B10), the approximation sign can become an equal sign if h_{e_1} is used to replace h_e :



680
$$h_{e_1} = h_b - \frac{f(h_b)}{f'(h_b)}, \quad (B11)$$

where

$$f(h_b) = (-1.68h_b^2 + 9.5kLh_b - 12.5k^2L^2)\overline{w'T}, \quad (B12)$$

and

$$\begin{aligned} f'(h_b) = & 1.20 \frac{\gamma}{\Delta t} h_b^3 - 2.50 \frac{\gamma k L}{\Delta t} h_b^2 \\ & + \left(11.2 \frac{u_*^2 T}{\Delta t g} - 1.68 \overline{w'T} \right) h_b - k L \left(40.0 \frac{u_*^2 T}{\Delta t g} - 4.75 \overline{w'T} \right). \end{aligned} \quad (B13)$$

685 If we return to Eqs. (B9) and (B11), we see that the difference between h_{e1} and h_e is small but unknown. Following the Newton-Rapson method, h_{e1} from Eq. (B11) is used to replace h_b , and h_{e_1} on the left side is replaced with a new variable h_{e_2} , with its subscript 2 indicating that it is the 2nd approximation for h_e . In such a way, h_e can be iteratively approached by $h_{e_{n+1}}$, mathematically described as

$$h_{e_{n+1}} = h_{e_n} - \frac{f(h_{e_n})}{f'(h_{e_n})} \quad (B14)$$

690 where subscript n is a positive integer indicating the n^{th} approximation for h_e while $f(h_{e_n})$ and $f'(h_{e_n})$ can be derived in the same way as for Eqs. (B12) and (B13) from $f(h_b)$. Until $|h_{e_{n-1}} - h_{e_{n-1}}| < 1.96\sigma$, and as long as $f(h_{e_n})$ and $f'(h_{e_n})$ are valid, h_e can be acquired by

$$h_e \approx h_{e_{n+1}}, \quad (B15)$$

where σ is the published precision of direct measurements from a ceilometer, and 1.96σ is the accuracy in h_e from the 695 solution procedure above. In this study the precision of the SkyVUETM PRO Ceilometer is used for σ (5 m, Campbell Scientific Inc., 2025). Alternatively, if during the iteration process, $f(h_{e_n})$ and/or $f'(h_{e_n})$ become invalid, h_e can be acquired backwards by

$$h_e \approx h_{e_n} \quad (B16)$$

And with the value of h_e , h value can be calculated from Eq. (B5).

700 While an eddy-covariance flux system is running into a new Δt , the h_e value becomes h_b value of current Δt . However, h_e from a previous Δt does not exist in the first Δt immediately after an eddy-covariance system starts, or in the case data variables such as u_* , L , T , and/or $\overline{w'T}$ are not available due to a system restart, power outage, or heavy precipitation/dust interfering with measurements from the sonic anemometer or gas analyzer. In such a case, for quick



starting or resuming the continuity of data, an alternative approach described in the next section can be used to approximate 705 h for such a “first” output interval.

B4 Approximation to h for the “first” output interval

For any “first” Δt under neutral to stable conditions, h can be computed from Eq. (B1), and under convective conditions, it can be approximated from the 2nd order Lagrange interpolation polynomial:

$$h = \left[\frac{80}{381}(L+30) - \frac{5}{31}(L+15) \right] (L+650) + \frac{12}{3937}(L+15)(L+30), \quad (B16)$$

710 which is developed based on the data from Table 1 in Kljun et al. (2015). Even after a first h value is obtained, if convective conditions persist, Eqs. (B7) and (B11) cannot be used until a trend (e.g., at least two values) of h are known. Once a trend is established, then the current h_e can be estimated, which can then be substituted for h_b in the next Δt over which Eq. (B3) theory can be applied to estimate h under convective conditions.

B5 Estimation of h_b every Δt

715 To establish the trend for h , at least one more value for h must be acquired in the same way as the “first” Δt . Once known, these values provide an estimate of h_e for current Δt :

$$h_e = h + \frac{h - h_p}{2}, \quad (B17)$$

where subscript p indicates a previous interval. The estimate becomes h_b value for next Δt (i.e., the 3rd Δt). This h_b will then be used to compute h_e from Eqs. (B7) to (B16) under convective boundary layer conditions. If conditions become neutral to 720 stable, h is once again directly computed from Eq. (B1) without using h_b .

B6 Summary

The algorithm developed above was implemented into EasyFlux series (Campbell Scientific Inc. UT, USA) for computing h . The h value is used for the applications of flux footprint equations from Kljun et al. (2015). This value is stored in flux datasets as the variable name *PBLH_F*, following the Ameriflux variable naming convention (AmeriFlux, 2018).

725 Appendix C: Subroutine in EasyFlux for footprint characteristics from Kljun et al. (2015)

C1 Variable notation

Subroutine	main program	
U_star	USTAR	Friction velocity
h_aerodynamic	z	Aerodynamic height
Obukhov	MO_LENGTH	Monin-Obukhov length
h_PBL	PBLH_F	Planetary boundary layer height
u_z	U	Mean wind speed at height of z in the streamwise direction
range_intrst	FETCH_INTRST	Upwind fetch of interest (measurement targeted range)
FP_range_intrst	FP_FETCH_INTRST	Percentage of measured scalar flux from upwind fetch of interest
range(1)	fetch(1) = FETCH_MAX	Upwind location of sources/sinks that contribute most to measured flux
range(2)	fetch(2) = FETCH_70	Upwind fetch within which the sources/sinks contribute 70% to measured flux
range(3)	fetch(3) = FETCH_80	Upwind fetch within which the sources/sinks contribute 80% to measured flux
range(4)	fetch(4) = FETCH_90	Upwind range within which the sources/sinks contribute 90% to measured flux



Dim FP_even 'Summed values of footprint at X^* on the right boundary of sequentially even increment
Dim FP_end 'Footprint at the ending X^* of integration section

C2.2 Computations

795 **a. Variable Preparation**

Select Case Obukhov

Case Is < 0
 stablty_index = 1
 Move (paramtr_symbls(1), 4, paramtr_valus(1, 1), 4)
 800 Move (x_star_infl_max_symbls(1), 3, x_star_infl_max_valus(1, 1), 3)
 Move (cumul_fp_segmnt_symbls(1), 3, cumul_fp_segmnt_valus(1, 1), 3)
 Move (x_star_p_symbls(1), 3, x_star_p_valus(1, 1), 3)

Case Is >= 0
 stablty_index = 2
 805 Move (paramtr_symbls(1), 4, paramtr_valus(2, 1), 4)
 Move (x_star_infl_max_symbls(1), 3, x_star_infl_max_valus(2, 1), 3)
 Move (cumul_fp_segmnt_symbls(1), 3, cumul_fp_segmnt_valus(1, 1), 3)
 Move (x_star_p_symbls(1), 3, x_star_p_valus(2, 1), 3)

EndSelect 'Obukhov

810

b. *FETCH_MAX*

rang(1) = ($k^*x_{max}^*h_{aerodynamic}^*u_z$)/($U_{star}^*(1-h_{aerodynamic}/h_{PBL})$) 'k is van Karman constant, given in main program

c. *FETCH_70, FETCH_80, and FETCH_90*

rang(2) = ($k^*x_{70}^*h_{aerodynamic}^*u_z$)/($U_{star}^*(1-h_{aerodynamic}/h_{PBL})$)

815 rang(3) = ($k^*x_{80}^*h_{aerodynamic}^*u_z$)/($U_{star}^*(1-h_{aerodynamic}/h_{PBL})$)

rang(4) = ($k^*x_{90}^*h_{aerodynamic}^*u_z$)/($U_{star}^*(1-h_{aerodynamic}/h_{PBL})$)

d. *Footprint portion of measured flux within an upwind fetch of interest for measurements in real-scale fields*

Preparation for numerical integration

x_star_intrst = (range_intrst/h_aerodynamic)*(1-h_aerodynamic/h_PBL)*(U_star/(k*u_z))

820

Select Case x_star_intrst

Case Is <= x_f1
 integrtn_incrmnt = (x_star_intrst - d0*(1+1e-7))/1000
 fp_segmnt_ahead = 0
 x_star = d0*(1+1e-7)

825

Case Is > x_f1 AND Is <= x_max
 integrtn_incrmnt = (x_star_intrst - x_f1)/1000
 fp_segmnt_ahead = cumul_x_f1
 x_star = x_f1

830

Case Is > x_max AND Is <= x_f2
 integrtn_incrmnt = (x_star_intrst - x_max)/1000
 fp_segmnt_ahead = cumul_x_max
 x_star = x_max

835

Case Is > x_f2
 integrtn_incrmnt = (x_star_intrst - x_f2)/1000
 fp_segmnt_ahead = cumul_x_f2
 x_star = x_f2

EndSelect 'x_star_intrst

Preliminary values of FP_start, FP_odd, FP_even for use inside an iteration

840 FP_start = (a*(x_star - d0)^b)*EXP(-c/(x_star - d0)) 'Footprint at the starting X^* of integration section
 FP_odd = 0
 FP_even = 0



For i_fp = 1 To 499

```

x_star = x_star + integrtn_incrmnt
FP_odd = FP_odd + (a*(x_star - d0)^b)*EXP(-c/(x_star - d0))

x_star = x_star + integrtn_incrmnt
FP_even = FP_even + (a*(x_star - d0)^b)*EXP(-c/(x_star - d0))

```

Next i_fp
 $FP_end = (a * (x_star - d0)^b) * EXP(-c / (x_star - d0))$
 $FP_even = FP_even - FP_end$

Composite Simpson's Rule for numerical integrations (below, the 2nd term on the right)

$$\text{FP_range_intrst} = \text{fp_segmnt_ahead} + 100 * (\text{integrtm_incrmt} / 3) * (\text{FP_start} + 4 * \text{FP_odd} + 2 * \text{FP_even} + \text{FP_end})$$

EndSub 'Footprnt_Charctrstcs_Kljun_etal2015

C3 The use of subroutine in the main program of EasyFlux series (Campbell Scientific Inc., UT, USA)

The Subroutine to compute the footprint characteristics from Kljun et al. (2015) is used in EasyFlux series through a Call instruction:

Call Footprnt_Charctrstcs_Kljun_etal2015(USTAR, z , MO_LENGTH, PBLH_F, U , FETCH_INTRST, FP_FETCH_INTRST, fetch(1))

For every averaging interval in eddy-covariance systems, *USTAR*, *z*, *MO_LENGTH*, and *U* have their values available from measurements, *PBLH_F* is computed using the algorithm from Appendix B, *FETCH_INTRST* is entered by a user before or while EasyFlux is running, and the values of flux footprint characteristics are output from this subroutine above that is executable as long as a prime is put ahead of each line of text.

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