





37 **1 Introduction**

38

39 Concern and uncertainty about the magnitude of sea level rise due to melting of the  
40 Greenland and Antarctic ice sheets have led to increased interest in improved ice sheet  
41 and glacier modeling. The gold standard is a full-Stokes model (i.e., a model that solves  
42 the nonlinear, non-Newtonian Stokes system of equations for incompressible ice sheet  
43 dynamics) because it is applicable to all geometries and flow regimes. However, the  
44 Stokes model is computationally demanding and expensive to solve. It is a nonlinear,  
45 three-dimensional model involving four variables, namely, the three velocity components  
46 and pressure. In addition, pressure is a Lagrange multiplier enforcing incompressibility  
47 and this creates a more difficult indefinite “saddle point” problem. As a result, full-  
48 Stokes models exist but are not commonly used in practice (examples are FELIX-S, Leng  
49 et al., 2012; Elmer/Ice, Gagliardini et al., 2013).

50

51 Because of these difficulties with the Stokes model, there is much interest in  
52 simpler and cheaper approximate models. There is a hierarchy of very simple models  
53 such as the shallow ice (SIA) and shallow-shelf (SSA) models, and there are also various  
54 higher-order approximations. These culminate in the Blatter-Pattyn (BP) approximation  
55 (Blatter, 1995; Pattyn, 2003), which is currently used in production code packages such  
56 as ISSM (Larour et al., 2012), MALI (Hoffman et al., 2018; Tezaur et al., 2015) and  
57 CISM (Lipscomb et al., 2019). This approximation is based on the assumption of a small  
58 ice sheet aspect ratio, i.e.,  $\varepsilon = H/L \ll 1$ , where  $H, L$  are the vertical and horizontal  
59 length scales, and consequently it eliminates certain stress terms and implicitly assumes  
60 small basal slopes. Both the Stokes and Blatter-Pattyn models are described in detail in  
61 Dukowicz et al. (2010), hereafter referred to as DPL (2010). Although the Blatter-Pattyn  
62 model is reasonably accurate for large-scale motions, accuracy deteriorates for small  
63 horizontal scales, less than about five ice thicknesses in the ISMIP–HOM model  
64 intercomparison (Pattyn et al., 2008; Perego et al., 2012), or below a 1 km resolution as  
65 found in a detailed comparison with full Stokes calculations (Rückamp et al., 2022). This  
66 can become particularly important for calculations involving details near the grounding  
67 line where the full accuracy of the Stokes model is needed (Nowicki and Wingham,  
68 2008). Attempts to address the problem while avoiding the use of full Stokes solvers  
69 include variable grid resolution coupled with a Blatter-Pattyn solver (Hoffman et al.,  
70 2018) and variable model complexity, where a Stokes solver is embedded locally in a



71 lower order model (Seroussi et al., 2012). Better approximations, more accurate than  
72 Blatter-Pattyn but cheaper than Stokes, are not currently available.

73

74 The present paper introduces two innovations that may begin to address some of  
75 these issues. The first is a novel transformation of the Stokes model, described in §3,  
76 which puts it into a form closely resembling the Blatter-Pattyn model and differing only  
77 by the presence of a few extra terms. This allows a code to be switched over from Stokes  
78 to Blatter-Pattyn, and vice-versa, globally or locally, by the use of a single parameter that  
79 turns off these extra terms. As a result, variable model complexity can be very simply  
80 implemented, as described in §6.1. The second innovation is the introduction of new  
81 finite element grids that decouple the discrete continuity equation and allow it to be  
82 solved for the vertical velocity in terms of the horizontal velocity components. Several  
83 elements that may be used to construct such grids are described in Appendix C in both  
84 2D and 3D, primarily the first order P1-E0 and second order P2-E1 elements (these two  
85 elements are so-named because they employ edge-based pressures). These grids facilitate  
86 new approximations that improve on the Blatter-Pattyn approximation within the  
87 framework of the transformed Stokes model. We describe two such approximations in  
88 §6.2. There is another very significant benefit. A conventional ice sheet Stokes model  
89 discretized on such a grid is numerically equivalent to an inherently stable positive-  
90 definite minimization (i.e., optimization) problem, as demonstrated in Appendix D. This  
91 is in contrast to the ubiquitous Stokes finite element practice of needing to use elements  
92 that satisfy the “inf-sup” or “LBB” condition for stability (see Elman et al., 2014, and the  
93 brief discussion in §4.3.1).

94

## 95 **2 The Standard Formulation of the Stokes Ice Sheet Model**

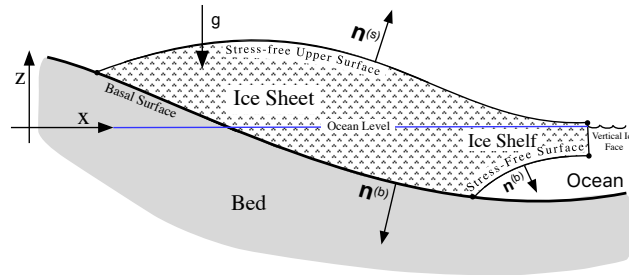
### 96 **2.1 The Assumed Ice Sheet Configuration**

97

98 An ice sheet may be divided into two parts, a part in contact with the bed and a floating  
99 ice shelf located beyond the grounding line. The Stokes ice sheet model is capable of  
100 describing the flow of an arbitrarily shaped ice sheet, including a floating ice shelf as  
101 illustrated in Fig. 1, given appropriate boundary conditions (e.g., Cheng et al., 2020).  
102 One limitation of the methods proposed here, in common with the Blatter-Pattyn model,  
103 will be that upper and basal surfaces must be able to be connected by a vertical line of sight,

104 as is the case in Fig. 1. Here, for simplicity, we will only consider a fully grounded ice  
 105 sheet with periodic lateral boundary conditions, i.e., no ice shelf.

106



107

108

109 **Figure 1** A simplified illustration of the admissible ice sheet configuration.

110

111 Referring to Fig. 1, the entire surface of the ice sheet is denoted by  $S$ . An upper

112 surface, labeled  $S_s$  and specified by  $\zeta_s(x, y, z) = z - z_s(x, y) = 0$ , is exposed to the

113 atmosphere and thus experiences stress-free boundary conditions. The bottom or basal

114 surface, denoted by  $S_B$  and specified by  $\zeta_b(x, y, z) = z - z_b(x, y) = 0$ , is in contact with

115 the bed. The basal surface may be subdivided into two sections,  $S_B = S_{B1} + S_{B2}$ , where

116  $S_{B1}$ , specified by  $z = z_{b1}(x, y)$ , is the part where ice is frozen to the bed (a no-slip

117 boundary condition), and  $S_{B2}$ , specified by  $z = z_{b2}(x, y)$ , is where frictional sliding

118 occurs. We assume Cartesian coordinates such that  $x_i = (x, y, z)$  are position coordinates

119 with  $z = 0$  at the ocean surface, and the index  $i \in \{x, y, z\}$  represents the three Cartesian

120 indices. Later we shall have occasion to introduce the restricted index  $(i) \in \{x, y\}$  to

121 represent just the two horizontal indices. The associated unit normal vectors are  $n_i^{(s)}$ ,

122  $n_i^{(b1)}$ ,  $n_i^{(b2)}$  at the stress-free and basal surfaces, respectively. For the particular geometry

123 illustrated in Fig. 1 we see that  $n_z^{(s)} > 0$  and  $n_z^{(b1)}, n_z^{(b2)} < 0$ . Unit normal vectors

124 appropriate for the ice sheet configuration of Fig. 1 are given by



$$n_i^{(s)} = \left( n_x^{(s)}, n_y^{(s)}, n_z^{(s)} \right) = \frac{\partial \zeta_s(x, y, z) / \partial x_i}{\left| \partial \zeta_s(x, y, z) / \partial x_i \right|} = \frac{(-\partial z_s / \partial x, -\partial z_s / \partial y, 1)}{\sqrt{1 + (\partial z_s / \partial x)^2 + (\partial z_s / \partial y)^2}},$$

$$n_i^{(b)} = \left( n_x^{(b)}, n_y^{(b)}, n_z^{(b)} \right) = -\frac{\partial \zeta_b(x, y, z) / \partial x_i}{\left| \partial \zeta_b(x, y, z) / \partial x_i \right|} = \frac{(\partial z_b / \partial x, \partial z_b / \partial y, -1)}{\sqrt{1 + (\partial z_b / \partial x)^2 + (\partial z_b / \partial y)^2}}.$$

126

## 127 2.2 The Stokes Equations

128

129 The Stokes model is given by a system of nonlinear partial differential equations and  
 130 associated boundary conditions (Greve and Blatter, 2009; DPL, 2010). In a Cartesian  
 131 coordinate system the Stokes equations, the three momentum equations and the  
 132 continuity equation, for the three velocity components  $u_i = (u, v, w)$  and the pressure  $P$   
 133 are given by

$$134 \quad \frac{\partial \tau_{ij}}{\partial x_j} - \frac{\partial P}{\partial x_i} + \rho g_i = 0, \quad (2)$$

$$135 \quad \frac{\partial u_i}{\partial x_i} = 0, \quad (3)$$

136 where  $\rho$  is the density, and  $g_i$  is the acceleration due to gravity vector, arbitrarily  
 137 oriented in general but here taken to be oriented in the negative z-direction,  
 138  $g_i = (0, 0, -g)$ . Repeated indices imply summation (the Einstein notation). The  
 139 deviatoric stress tensor  $\tau_{ij}$  is given by

$$140 \quad \tau_{ij} = 2\mu_n \dot{\epsilon}_{ij}, \quad (4)$$

141 where  $\mu_n$  is a nonlinear ice viscosity defined by

$$142 \quad \mu_n = \eta_0 \left( \dot{\epsilon}^2 \right)^{(1-n)/2n}, \quad (5)$$

143 and  $\dot{\epsilon}^2 = \dot{\epsilon}_{ij} \dot{\epsilon}_{ij} / 2$  is the second invariant of the strain rate tensor  $\dot{\epsilon}_{ij}$ . The strain rate  
 144 tensor is given by

$$145 \quad \dot{\epsilon}_{ij} = \frac{1}{2} \left( \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right), \quad (6)$$

146 and therefore the second invariant may be written out as



$$147 \quad \dot{\epsilon}^2 = \frac{1}{2} \left[ \left( \frac{\partial u}{\partial x} \right)^2 + \left( \frac{\partial v}{\partial y} \right)^2 + \left( \frac{\partial w}{\partial z} \right)^2 \right] + \frac{1}{4} \left[ \left( \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right)^2 + \left( \frac{\partial u}{\partial z} + \frac{\partial w}{\partial x} \right)^2 + \left( \frac{\partial v}{\partial z} + \frac{\partial w}{\partial y} \right)^2 \right]. \quad (7)$$

148 Note that the second invariant is positive-definite, i.e.,  $\dot{\epsilon}^2 \geq 0$ . As usual, ice is assumed  
 149 to obey Glen's flow law, where  $n$  is the Glen's law exponent ( $n = 1$  for a linear  
 150 Newtonian fluid, and typically  $n = 3$  in ice sheet modeling, resulting in a nonlinear non-  
 151 Newtonian fluid). The coefficient  $\eta_0$  is defined by  $\eta_0 = A^{-1/n} / 2$ , where  $A$  is an ice flow  
 152 factor, here taken to be a constant but in general depending on temperature and other  
 153 variables (see Schoof and Hewitt, 2013). The three-dimensional Stokes system (2), (3)  
 154 requires a set of boundary conditions at every bounding surface, each set being composed  
 155 of three components. Aside from the periodic lateral boundary conditions used in our test  
 156 problems, the relevant boundary conditions are as follows

157 (1) Stress-free boundary conditions on surfaces  $S_s$  not in contact with the bed, such  
 158 as the upper surface  $S_s$ :

$$159 \quad \tau_{ij} n_j^{(s)} - P n_i^{(s)} = 0. \quad (8)$$

160 The basal boundary conditions are given by

161 (2) No-slip or frozen to the bed conditions on surface segment  $S_{B1}$ :

$$162 \quad u_i = 0 \quad (9)$$

163 (3) Frictional tangential sliding conditions on surface segment  $S_{B2}$ :

164 Frictional conditions are more complicated and are discussed in detail in Appendix A. In  
 165 summary, these conditions are composed of two parts,

166 (3a) A single condition enforcing tangential flow at the basal surface:

$$167 \quad u_i n_i^{(b2)} = 0. \quad (10)$$

168 (3b) Two conditions specifying the horizontal components of the tangential  
 169 frictional stress force vector. From Appendix A, the simplest representation of these two  
 170 conditions is

$$171 \quad n_z^{(b2)} \left( \tau_{(i)j} n_j^{(b2)} + f_{(i)} \right) - n_{(i)}^{(b2)} \left( \tau_{zj} n_j^{(b2)} + f_z \right) = 0, \quad (11)$$

172 where  $(i) \in \{x, y\}$  is the notation previously introduced for restricted (horizontal) indices,

173 and  $f_i$  is a specified frictional sliding force vector, tangential to the bed ( $n_i^{(b2)} f_i = 0$ ).



174 This is potentially a complicated function of position and velocity (e.g., Schoof, 2010),  
 175 however, here we assume only simple linear frictional sliding,

$$176 \quad f_i = \beta(x) u_i, \quad (12)$$

177 where  $\beta(x) > 0$  is a position-dependent drag law coefficient. For simplicity we assume  
 178 there is no melting or refreezing at the bed resulting in vertical inflows or outflows. If  
 179 needed, these can be easily added (Dukowicz et al., 2010; Heinlein et al., 2022).

180

### 181 **2.3 The Stokes Variational Principle**

182

183 A variational principle, if available, is usually the most compact way of representing a  
 184 particular problem. The Stokes model possesses a variational principle that is  
 185 particularly useful for discretization purposes and for the specification of boundary  
 186 conditions (see DPL, 2010, for a fuller description of the variational principle applied to  
 187 ice sheet modeling). There are a number of significant advantages. For example, all  
 188 boundary conditions are conveniently incorporated in the variational formulation, all  
 189 terms in the variational functional, including boundary condition terms, contain lower  
 190 order derivatives than in the momentum equations, and the solution of the discrete  
 191 problem automatically involves a symmetric matrix. In discretizing the momentum  
 192 equations, stress terms at boundaries involve derivatives that require information from  
 193 across boundaries. This problem does not arise in the variational formulation since all  
 194 terms are evaluated in the interior. Finally, stress-free boundary conditions, as at the  
 195 upper surface for example, need not be specified at all since they are automatically  
 196 incorporated in the functional as natural boundary conditions. In discrete applications,  
 197 the variational method presented here is closely related to the Galerkin finite element  
 198 method, a subset of the weak formulation method in which the test and trial functions are  
 199 the same (see Schoof, 2010, in connection with the Blatter-Pattyn model).

200

201 The variational functional for the standard Stokes model may be written in two  
 202 alternative forms:

203 (1) Basal boundary conditions imposed using Lagrange multipliers:

$$204 \quad \mathcal{A}[u_i, P, \lambda_i, \Lambda] = \int_V dV \left[ \frac{4n}{n+1} \eta_0 (\dot{\epsilon}^2)^{(1+n)/2n} - P \frac{\partial u_i}{\partial x_i} + \rho g w \right] \\
 + \int_{S_{B1}} dS \lambda_i u_i + \int_{S_{B2}} dS \left[ \Lambda u_i n_i^{(b2)} + \frac{1}{2} \beta(x) u_i u_i \right], \quad (13)$$

205 where  $\lambda_i$  and  $\Lambda$  are Lagrange multipliers used to enforce the no-slip condition and  
 206 frictional tangential sliding, respectively. As in DPL (2010), arguments enclosed in  
 207 square brackets, here  $u_i, P, \lambda_i, \Lambda$ , indicate those variables that are used in the variation of  
 208 the functional.

209 (2) Basal boundary conditions imposed by direct substitution:

210 In this case, the two conditions (9), (10) are used directly in the functional to specify all  
 211 three velocity components  $u_i$  in the first case, and the vertical velocity  $w$  in terms of the  
 212 horizontal velocity components in the second case, along the entire basal boundary in  
 213 both the volume and surface integrals in (13). In particular, (10) is used in the following  
 214 form,

$$215 \quad w = -\frac{u_{(i)} n_{(i)}^{(b2)}}{n_z^{(b2)}} = u_{(i)} \frac{\partial z_b}{\partial x_{(i)}}, \quad (14)$$

216 to replace  $w$  in terms of the horizontal velocity components  $u_{(i)}$  on the basal boundary  
 217 segment  $S_{B2}$ . Here we use  $z_b$  as a shorthand notation for  $z_b(x, y)$ . The variational  
 218 functional in this case becomes

$$219 \quad \mathcal{A}[u_i, P] = \int_V dV \left[ \frac{4n}{n+1} \eta_0 (\dot{\epsilon}^2)^{(1+n)/2n} - P \frac{\partial u_i}{\partial x_i} + \rho g w \right] \\ + \frac{1}{2} \int_{S_{B2}} dS \beta(x) \left( u_{(i)} u_{(i)} + \left( u_{(i)} n_{(i)}^{(b2)} / n_z^{(b2)} \right)^2 \right). \quad (15)$$

220 Note that (14) has been explicitly used to replace  $w$  in the basal boundary component of  
 221 the functional (15) but, importantly, it must also be used in the volume integral part of  
 222 (15) to replace all values of  $w$  that lie on the basal boundary segment  $S_{B2}$ .

223

224 As described in DPL (2010), a variational procedure, i.e., taking the variation  
 225 with respect to the independent functions  $u_i, P, \lambda_i, \Lambda$  in (13), and  $u_i, P$  in (15), yields the  
 226 full set of Euler-Lagrange equations and boundary conditions that specify the standard  
 227 Stokes model, equivalent to (2)-(11). In the case of (13), the system determines all the  
 228 discrete variables specified on the mesh: the velocity components and the pressure,  $u_i, P$ ,  
 229 together with the Lagrange multipliers,  $\lambda_i, \Lambda$ . In the case of (15), the system only  
 230 determines the unspecified velocity variables  $u_i$  and the pressure  $P$ . The specified

231 values of velocity are then obtainable a posteriori from (9) or (14). As a result, system  
232 (15) is smaller and simpler and is therefore the one predominantly used in this paper.

233

### 234 **3. A Transformation of the Stokes Model**

#### 235 **3.1 Origin of the Transformation**

236

237 The transformation is motivated by the Blatter-Pattyn approximation. Consider the  
238 vertical component of the momentum equation and the corresponding stress-free upper  
239 surface boundary condition in the Blatter-Pattyn approximation (from DPL, 2010, for  
240 example), which are given by

$$\begin{aligned} 241 \quad & \frac{\partial}{\partial z} \left( 2\mu_n \frac{\partial w}{\partial z} \right) - \frac{\partial P}{\partial z} - \rho g = 0, \\ & \left( 2\mu_n \frac{\partial w}{\partial z} - P \right) n_z^{(s)} = 0 \quad \text{at } z = z_s(x, y). \end{aligned} \quad (16)$$

242 These equations may be rewritten in the form

$$\begin{aligned} 243 \quad & \frac{\partial}{\partial z} \left( P - 2\mu_n \frac{\partial w}{\partial z} + \rho g (z - z_s(x, y)) \right) = 0, \\ & \left( P - 2\mu_n \frac{\partial w}{\partial z} + \rho g (z - z_s(x, y)) \right) n_z^{(s)} = 0 \quad \text{at } z = z_s(x, y). \end{aligned} \quad (17)$$

244 This suggests the introduction of a new variable  $\tilde{P}$ , to be called the transformed pressure,

$$245 \quad \tilde{P} = P - 2\mu_n \frac{\partial w}{\partial z} + \rho g (z - z_s(x, y)), \quad (18)$$

246 which simplifies system (17) as follows

$$\begin{aligned} 247 \quad & \frac{\partial \tilde{P}}{\partial z} = 0, \\ & \tilde{P} n_z^{(s)} = 0 \quad \text{at } z = z_s(x, y). \end{aligned} \quad (19)$$

248 This is a complete one-dimensional partial differential system, that, when integrated from  
249 the top surface down yields

$$250 \quad \tilde{P} = 0. \quad (20)$$

251 Thus, the transformed pressure vanishes in the Blatter-Pattyn case. The definition (18)

252 forms the basis of the present transformation but we also use the continuity equation to

253 eliminate  $\partial w / \partial z$  as is done in the Blatter-Pattyn approximation (see DPL, 2010).

254 Therefore, the transformation consists of eliminating  $P$  and  $\partial w/\partial z$  in the Stokes system  
 255 (2), (4)-(11) (i.e., everywhere except in the continuity equation (3) itself) by means of

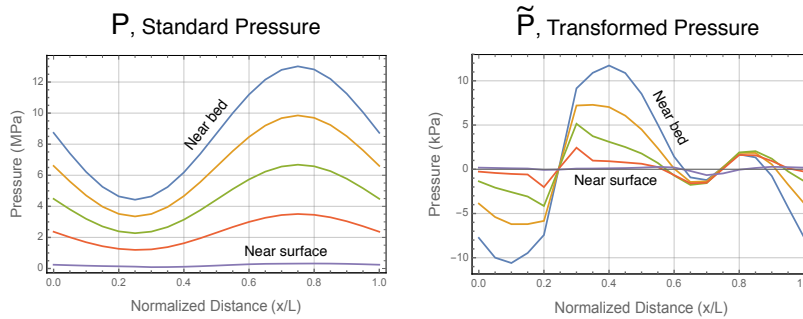
256 
$$P = \tilde{P} - 2\mu_n \left( \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} \right) + \rho g(z_s - z), \quad (21)$$

257 
$$\frac{\partial w}{\partial z} = - \left( \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} \right), \quad (22)$$

258 where  $z_s$  is a shorthand notation for  $z_s(x, y)$ .

259

260 In the standard Stokes system the pressure  $P$  is primarily a Lagrange multiplier  
 261 enforcing incompressibility but with a very large hydrostatic component. The  
 262 transformation eliminates the hydrostatic pressure from  $\tilde{P}$ , as illustrated in Fig. 2 where  
 263 the two pressures, plotted along grid lines, from Exp. B in the ISMIP–HOM model  
 264 intercomparison (Pattyn et al., 2008) at  $L = 10$  km are compared. The standard Stokes  
 265 pressure  $P$  is some three orders of magnitude larger than the transformed pressure  $\tilde{P}$ .



266

267 **Figure 2.** Standard pressure  $P$  compared to the transformed pressure  $\tilde{P}$  in Exp. B from  
 268 the ISMIP–HOM model intercomparison. Note that  $P$  is in MPa while  $\tilde{P}$  is in kPa.  
 269

270 The transformed pressure  $\tilde{P}$  is again a Lagrange multiplier enforcing  
 271 incompressibility, i.e., it may be viewed as the effective pressure in the transformed  
 272 system. Alternatively, since  $\tilde{P} = 0$  in the Blatter-Pattyn approximation, the definition of  
 273  $\tilde{P}$  from (18) may be written as  $\tilde{P} = P - P_{BP}$ , where

274 
$$P_{BP} = -2\mu_n \left( \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} \right) + \rho g(z_s - z)$$

275 is the effective Blatter-Pattyn pressure (Tezaur et al., 2015). As a result, we have  
 276  $P = P_{BP} + \tilde{P}$ , and therefore  $\tilde{P}$  is actually the “Stokes” correction to the Blatter-Pattyn  
 277 pressure.

278

### 279 3.2 The Transformed Stokes Equations

280

281 Introducing (21), (22) into the Stokes system of equations (2)-(11) results in the  
 282 following transformed Stokes system:

$$283 \quad \frac{\partial \tilde{\tau}_{ij}}{\partial x_j} - \hat{\xi} \frac{\partial \tilde{P}}{\partial x_i} - \rho g \frac{\partial z_s}{\partial x_{(i)}} = 0, \quad (23)$$

$$284 \quad \hat{\xi} \frac{\partial u_i}{\partial x_i} = 0, \quad (24)$$

285 where quantities that are modified in the transformation are indicated by a tilde, e.g.,  $\tilde{P}$ .

286 Corresponding to (4), the modified Stokes deviatoric stress tensor  $\tilde{\tau}_{ij}$  is given by

$$287 \quad \tilde{\tau}_{ij} = 2\tilde{\mu}_n \left( \tilde{\epsilon}_{ij} + \frac{\partial u_{(i)}}{\partial x_{(i)}} \delta_{ij} \right), \quad (25)$$

288 where  $\delta_{ij}$  is the Kronecker delta, the modified strain rate tensor  $\tilde{\epsilon}_{ij}$ , corresponding to (6),

289 is given by

$$290 \quad \tilde{\epsilon}_{ij} = \begin{bmatrix} \frac{\partial u}{\partial x} & \frac{1}{2} \left( \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) & \frac{1}{2} \left( \frac{\partial u}{\partial z} + \xi \frac{\partial w}{\partial x} \right) \\ \frac{1}{2} \left( \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) & \frac{\partial v}{\partial y} & \frac{1}{2} \left( \frac{\partial v}{\partial z} + \xi \frac{\partial w}{\partial y} \right) \\ \frac{1}{2} \left( \frac{\partial u}{\partial z} + \xi \frac{\partial w}{\partial x} \right) & \frac{1}{2} \left( \frac{\partial v}{\partial z} + \xi \frac{\partial w}{\partial y} \right) & - \left( \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} \right) \end{bmatrix} \quad (26)$$

291 and, corresponding to (5), the modified viscosity,

$$292 \quad \tilde{\mu}_n = \eta_0 \left( \tilde{\epsilon}^2 \right)^{(1-n)/2n}, \quad (27)$$

293 is given in terms of the second invariant  $\tilde{\epsilon}^2 = \tilde{\epsilon}_{ij} \tilde{\epsilon}_{ij} / 2$ , which, in expanded form becomes

$$294 \quad \tilde{\epsilon}^2 = \left( \frac{\partial u}{\partial x} \right)^2 + \frac{\partial u}{\partial x} \frac{\partial v}{\partial y} + \left( \frac{\partial v}{\partial y} \right)^2 + \frac{1}{4} \left[ \left( \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right)^2 + \left( \frac{\partial u}{\partial z} + \xi \frac{\partial w}{\partial x} \right)^2 + \left( \frac{\partial v}{\partial z} + \xi \frac{\partial w}{\partial y} \right)^2 \right]. \quad (28)$$

295 The dummy variables  $\xi = 1, \hat{\xi} = 1$  identify terms that are dropped in the Blatter-Pattyn  
 296 approximation, as explained below. Since (28) differs from (7) only by the use of  
 297 substitution (22), the transformation leaves the second invariant  $\tilde{\epsilon}^2$  and viscosity  $\tilde{\mu}_n$   
 298 unchanged provided the continuity equation (24) is satisfied, i.e.,  $\tilde{\epsilon}^2 = \epsilon^2$  and  $\tilde{\mu}_n = \mu_n$ ,  
 299 and in particular, the transformed second invariant remains positive-definite, i.e.,  $\tilde{\epsilon}^2 \geq 0$ .  
 300

301 The boundary conditions for the transformed equations, corresponding to (8)-(11),  
 302 are given by

303 BCs on  $S_S$  : 
$$\tilde{\tau}_{ij} n_j^{(s)} - \tilde{\xi} \tilde{P} n_i^{(s)} = 0 , \quad (29)$$

304 BCs on  $S_{B1}$  : 
$$u_i = 0 , \quad (30)$$

305 BCs on  $S_{B2}$  : 
$$u_i n_i^{(b2)} = 0 , \quad (31)$$

306 
$$n_z^{(b2)} \left( \tilde{\tau}_{(i)j} n_j^{(b2)} + \beta(x) u_{(i)} \right) - n_{(i)}^{(b2)} \left( \tilde{\tau}_{zj} n_j^{(b2)} + \beta(x) u_{(j)} n_{(j)}^{(b2)} / n_z^{(b2)} \right) = 0 . \quad (32)$$

307 Equations (31), (32) constitute the three required boundary conditions for frictional  
 308 sliding (see Appendix A). Note that (32) differs from (11) because (14) has been used to  
 309 eliminate the vertical velocity component  $w$  in favor of the horizontal velocity  
 310 components  $u_{(i)}$ .

311

312 The dummy variables  $\xi, \hat{\xi}$  in (23)-(25) and (26)-(29) have been introduced to  
 313 identify the terms that are neglected in the two types of the Blatter-Pattyn approximation  
 314 that we consider in §3.4. Specifically, these two types are (a) the standard Blatter-Pattyn  
 315 approximation,  $\xi = 0, \hat{\xi} = 0$ , as originally derived (Blatter, 1995; Pattyn, 2003; DPL,  
 316 2010), which solves for just the horizontal velocity components  $u, v$ , and (b) the extended  
 317 Blatter-Pattyn approximation,  $\xi = 0, \hat{\xi} = 1$ , described more fully later, which contains the  
 318 standard approximation and also provides the additional equations for determination of  
 319 the consistent vertical velocity component  $w$  and pressure  $\tilde{P}$ . Keeping all terms, i.e.,  
 320  $\xi = 1, \hat{\xi} = 1$ , specifies the full transformed Stokes model.

321

322 The transformed system (25)-(32) and the standard Stokes system (2)-(11) yield  
 323 exactly the same solution. However, in common with the Blatter-Pattyn approximation,





324 transformation (21) implies the use of a gravity-oriented coordinate system because of the  
 325 particular form of the gravitational forcing term, while the standard Stokes model does  
 326 not have this restriction. This is only a minor limitation. A somewhat more restrictive  
 327 limitation is the appearance of  $z_s(x, y)$ , an implicitly single valued function, to describe  
 328 the vertical position of the upper surface of the ice sheet. This means that care must be  
 329 taken in case of reentrant upper surfaces (i.e., S-shaped in 2D) and sloping cliffs at the ice  
 330 edge, a restriction not present in the standard Stokes model. As noted earlier, we assume  
 331 that the upper and basal surfaces are connected by a vertical line of sight. With a  
 332 reentrant ice surface, such a vertical line must be broken up into individual segments with  
 333 each segment having its own “upper” surface location  $z_s(x, y)$ . Fortunately, such  
 334 situations do not normally arise in practice. Exactly these same limitations exist in the  
 335 Blatter-Patten model, which does not hinder its use in practice.

336

### 337 **3.3 The Transformed Stokes Variational Principle**

338

339 It is easy to verify that the transformed Stokes system (23)-(32) results from the variation  
 340 with respect to  $u_i, \tilde{P}$  of the following functional:

$$\begin{aligned}
 341 \quad \tilde{\mathcal{A}}[u_i, \tilde{P}] = & \int_V dV \left[ \frac{4n}{n+1} \eta_0 (\tilde{\epsilon}^2)^{(1+n)/2n} - \hat{\xi} \tilde{P} \frac{\partial u_i}{\partial x_i} + \rho g u_{(i)} \frac{\partial z_s}{\partial x_{(i)}} \right] \\
 & + \frac{1}{2} \int_{S_{B2}} dS \beta(x) \left( u_{(i)} u_{(i)} + \left( u_{(i)} n_{(i)}^{(b2)} / n_z^{(b2)} \right)^2 \right), \quad (33)
 \end{aligned}$$

342 where  $\tilde{\epsilon}^2$  is the transformed second invariant from (28). Basal boundary conditions in  
 343 (33) are imposed by direct substitution, as in (15). Alternatively, one could impose  
 344 boundary conditions using Lagrange multipliers, as in (13), but direct substitution is  
 345 preferred because it is simpler and involves fewer variables. The remarks made in §2.3  
 346 about replacing all values of  $w$  that lie on the basal boundary segment  $S_{B2}$  by (14) apply  
 347 here also.

348

### 349 **3.4 Two Blatter-Pattyn Approximations**

#### 350 **3.4.1 The Standard Blatter-Pattyn Approximation**

351

352 The standard (or traditional) Blatter-Pattyn approximation (originally introduced by  
 353 Blatter, 1995; Pattyn, 2003; later by DPL, 2010; Schoof and Hewitt, 2013) is obtained by

354 setting  $\xi = 0, \hat{\xi} = 0$ . This yields the following Blatter-Pattyn variational functional in  
 355 terms of horizontal velocity components only,

$$356 \quad \mathcal{A}_{BP}[u_{(i)}] = \int_V dV \left[ \frac{4n}{n+1} \eta_0 (\hat{\epsilon}_{BP}^2)^{(1+n)/2n} + \rho g u_{(i)} \frac{\partial z_s}{\partial x_{(i)}} \right] \quad (34)$$

$$+ \frac{1}{2} \int_{S_{B2}} dS \beta(x) \left( u_{(i)} u_{(i)} + \zeta \left( u_{(i)} n_{(i)}^{(b2)} / n_z^{(b2)} \right)^2 \right),$$

357 where

$$358 \quad \hat{\epsilon}_{BP}^2 = \left( \frac{\partial u}{\partial x} \right)^2 + \frac{\partial u}{\partial x} \frac{\partial v}{\partial y} + \left( \frac{\partial v}{\partial y} \right)^2 + \frac{1}{4} \left[ \left( \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right)^2 + \frac{\partial u^2}{\partial z} + \frac{\partial v^2}{\partial z} \right], \quad (35)$$

359 and the corresponding Euler-Lagrange equations and boundary conditions are given by

$$360 \quad \frac{\partial \tau_{(ij)}^{BP}}{\partial x_j} - \rho g \frac{\partial z_s}{\partial x_{(i)}} = 0; \quad \begin{cases} \tau_{(ij)}^{BP} n_j^{(b2)} + \beta(x) \left( u_{(i)} + \zeta \left( u_{(i)} n_{(i)}^{(b2)} / n_z^{(b2)} \right) n_{(i)}^{(b2)} / n_z^{(b2)} \right) = 0 \\ \text{on } S_{B2}, \quad \tau_{(ij)}^{BP} n_j^{(s)} = 0 \text{ on } S_S, \quad u_{(i)} = 0 \text{ on } S_{B1}, \end{cases} \quad (36)$$

361 where the Blatter-Pattyn stress tensor  $\tau_{(ij)}^{BP}$  is

$$362 \quad \tau_{(ij)}^{BP} = \eta_0 (\hat{\epsilon}_{BP}^2)^{(1-n)/2n} \begin{bmatrix} 2 \left( 2 \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} \right) & \left( \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) & \frac{\partial u}{\partial z} \\ \left( \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) & 2 \left( \frac{\partial u}{\partial x} + 2 \frac{\partial v}{\partial y} \right) & \frac{\partial v}{\partial z} \end{bmatrix}. \quad (37)$$

363 There is a new dummy variable  $\zeta$  in (34) introduced to identify the basal boundary term  
 364 that is normally dropped ( $\zeta = 0$ ) in the standard Blatter-Pattyn approximation but which  
 365 was restored ( $\zeta = 1$ ) in Dukowicz et al. (2011) to better deal with arbitrary basal  
 366 topography.

367

368 The Blatter-Pattyn model is a well-behaved nonlinear approximate system for the  
 369 horizontal velocity components  $u, v$  because in this case the variational formulation is  
 370 actually a convex optimization problem whose solution minimizes the functional. As  
 371 noted in the Introduction, the Blatter-Pattyn approximation is widely used in practice as  
 372 an economical and relatively accurate ice sheet model. If desired, the vertical velocity  
 373 component  $w$  is computed a posteriori by means of the continuity equation.

374

375 **Remark #1:** The original formulation (e.g., Pattyn, 2003) also approximates the normal  
 376 unit vectors  $n_i^{(b2)}$  on the frictional part of the basal boundary  $S_{B2}$  by making the small  
 377 slope approximation (Dukowicz et al., 2011; Perego et al., 2012). However, this  
 378 additional approximation is unnecessary since any computational savings are negligible.

### 380 3.4.2 The Extended Blatter-Pattyn Approximation

381  
 382 A second form of the Blatter-Pattyn approximation is obtained from the transformed  
 383 variational principle (33) by making the assumption,

$$384 \quad \left\| \frac{\partial w}{\partial x} \right\| \ll \left\| \frac{\partial u}{\partial z} \right\|, \quad \left\| \frac{\partial w}{\partial y} \right\| \ll \left\| \frac{\partial v}{\partial z} \right\|, \quad (38)$$

385 and therefore neglecting  $\partial w/\partial x, \partial w/\partial y$  in the transformed second invariant  $\tilde{\epsilon}^2$ , or  
 386 equivalently, in the strain rate tensor  $\tilde{\epsilon}_{ij}$  from (26), consistent with the original small  
 387 aspect ratio approximation (Blatter, 1995; Pattyn, 2003; DPL, 2010; Schoof and  
 388 Hindmarsh, 2008). This corresponds to setting  $\xi = 0, \hat{\xi} = 1$  in the transformed Stokes  
 389 model. That is, we neglect vertical velocity gradients but keep the pressure Lagrange  
 390 multiplier term. This will be called the extended Blatter-Pattyn approximation (EBP)  
 391 because, in contrast to the standard Blatter-Pattyn approximation, all the variables, i.e.,  
 392  $u, v, w, \tilde{P}$ , are retained. Notably, assumption (38) is equivalent to just setting  $w = 0$  in  
 393 the second invariant  $\tilde{\epsilon}^2$  in the full transformed Stokes model (i.e., with  $\xi = 1, \hat{\xi} = 1$ ). In  
 394 other words, the extended BP approximation is obtained by neglecting vertical velocities  
 395 everywhere in (33) except where they occurs in the velocity divergence term. This aspect  
 396 of the transformed Stokes model will be exploited later to obtain approximations that  
 397 improve on Blatter-Pattyn. Thus, the extended Blatter-Pattyn functional is given by

$$398 \quad \mathcal{A}_{EBP}[u_i, \tilde{P}] = \int_V dV \left[ \frac{4n}{n+1} \eta_0 \left( \dot{\epsilon}_{BP}^2 \right)^{(1+n)/2n} - \tilde{P} \frac{\partial u_i}{\partial x_i} + \rho g u_{(i)} \frac{\partial z_s}{\partial x_{(i)}} \right] \\
 + \frac{1}{2} \int_{S_{B2}} dS \beta(x) \left( u_{(i)} u_{(i)} + \varsigma \left( u_{(i)} n_{(i)}^{(b2)} / n_z^{(b2)} \right)^2 \right), \quad (39)$$

399 where the Blatter-Pattyn second invariant  $\dot{\epsilon}_{BP}^2$  is given by (35). Taking the variation of  
 400 the functional, the resulting system of extended Blatter-Pattyn Euler-Lagrange equations  
 401 and their boundary conditions is given by

402 (1) Variation with respect to  $u_{(i)}$  yields the horizontal momentum equation:

$$403 \quad \frac{\partial \tau_{(i)j}^{BP}}{\partial x_j} - \frac{\partial \tilde{P}}{\partial x_{(i)}} - \rho g \frac{\partial z_s}{\partial x_{(i)}} = 0; \quad \left\{ \begin{array}{l} \tau_{(i)j}^{BP} n_j^{(s)} - \tilde{P} n_{(i)}^{(s)} = 0 \text{ on } S_S, \quad u_{(i)} = 0 \text{ on } S_{B1}, \\ \tau_{(i)j}^{BP} n_j^{(b2)} + \beta(x) \left( u_{(i)} + \zeta \left( u_{(k)} n_{(k)}^{(b2)} / n_z^{(b2)} \right) n_{(i)}^{(b2)} / n_z^{(b2)} \right) = 0 \\ \text{on } S_{B2}, \end{array} \right. \quad (40)$$

404 where  $\tau_{(i)j}^{BP}$  is given by (37).

405 (2) Variation with respect to  $w$  yields the vertical momentum equation:

$$406 \quad -\frac{\partial \tilde{P}}{\partial z} = 0; \quad \tilde{P} n_z^{(s)} = 0 \text{ on } S_S, \quad (41)$$

407 (3) Variation with respect to  $\tilde{P}$  yields the continuity equation:

$$408 \quad \frac{\partial w}{\partial z} + \frac{\partial u_{(i)}}{\partial x_{(i)}} = 0; \quad w = 0 \text{ on } S_{B1}, \text{ or } w = -u_{(i)} n_{(i)}^{(b2)} / n_z^{(b2)} \text{ on } S_{B2}. \quad (42)$$

409 This appears to be a coupled system for the complete set of variables,  $u, v, w, \tilde{P}$ , just as in  
 410 the transformed Stokes model. However, it is apparent that the vertical momentum  
 411 equation system (41) is decoupled and results in  $\tilde{P} = 0$ , as was shown in §3.1. This  
 412 eliminates pressure from the horizontal momentum equation (40), making it identical to  
 413 the standard Blatter-Pattyn system (36). Finally, having obtained the horizontal  
 414 velocities from the solution of (40), the continuity equation (42) may be solved for the  
 415 vertical velocity component  $w$  (but see the comments regarding the discrete case that  
 416 follow (43)).

417

418 In summary, the extended Blatter-Pattyn model, (40)-(42), is equivalent to the  
 419 standard Blatter-Pattyn model, (36), for the horizontal velocities,  $u, v$ , except that it also  
 420 includes two additional equations that determine the pressure  $\tilde{P}$  and the vertical velocity  
 421  $w$ , which are usually ignored in the standard Blatter-Pattyn approximation when only the  
 422 horizontal velocity is of interest. Because of this, we distinguish between the *Blatter-*  
 423 *Pattyn model* that solves for just the two horizontal velocities (i.e., the standard Blatter-  
 424 Pattyn approximation, (36)), and the *Blatter-Pattyn system* that solves for all the variables  
 425 (i.e., the extended Blatter-Pattyn approximation, (40)-(42)). It may not be obvious why  
 426 we wish to deal with the extended Blatter-Pattyn system since we already know that it is  
 427 equivalent to the simpler Blatter-Pattyn model. As it turns out, the Blatter-Pattyn system  
 428 is needed for future applications, to be described in §6, because it allows for a dual-model

429 code and for easy switching between the Blatter-Pattyn and Stokes models, which may be  
430 a useful feature in a general ice sheet code (e.g., ISSM, Larour et al., 2012), and because  
431 it also enables an adaptive hybrid scheme where the cheaper Blatter-Pattyn  
432 approximation is used locally within a Stokes model.

433

434 To complete the solution of the Blatter-Pattyn system once pressure  $\tilde{P}$  and the  
435 horizontal velocities  $u, v$  are available, the continuity equation (42) needs to be solved for  
436 the vertical velocity  $w$ . The use of the continuity equation to solve for the vertical  
437 velocity  $w$  is a novel feature of the Blatter-Pattyn approximation since the continuity  
438 equation is not normally used for this purpose. Using Leibniz's theorem, the continuity  
439 equation may be integrated starting from the bottom to obtain the vertical velocity in  
440 terms of horizontal velocity components, as follows

441 
$$w(u, v) = w_{z=z_b} - \int_{z_b}^z \frac{\partial u_{(i)}}{\partial x_{(i)}} dz' = u_{(i)} \frac{\partial z_b}{\partial x_{(i)}} - \int_{z_b}^z \frac{\partial u_{(i)}}{\partial x_{(i)}} dz' = - \frac{\partial}{\partial x_{(i)}} \int_{z_b}^z u_{(i)} dz'. \quad (43)$$

442 Note that we have replaced  $w_{z=z_b}$  by  $u_{(i)} \partial z_b / \partial x_{(i)}$ . This is valid for either of the basal  
443 boundary conditions (9) or (10) (here (10) is in the form given by (14)). When solving  
444 the Blatter-Pattyn system, the right-hand-side is known. However, (43) also works  
445 symbolically when the horizontal velocities  $u_{(i)}$  are not yet known, and therefore  $w(u, v)$   
446 is a functional of the unknown horizontal velocity distribution.

447

448 Thus far, we have only considered continuum results. A discrete finite element  
449 formulation, however, may not be well behaved. The solution of the discretized Stokes  
450 models and the associated Blatter-Pattyn approximations, and the ability to solve for the  
451 vertical velocity as in (43), will depend on the choices made for the grids and for the  
452 finite elements that are to be used. These issues will be discussed next.

453

## 454 **4. Finite Element Discretization**

### 455 **4.1 Standard and Transformed Stokes Discretizations**

456

457 In practice, both traditional Stokes and Blatter-Pattyn models are discretized using finite  
458 element methods (e.g., Gagliardini et al., 2013; Perego et al., 2012). We follow this  
459 practice except that here the discretization originates from a variational principle. This  
460 has a number of advantages (see §2.3 and DPL, 2010). The following is a brief outline of  
461 the finite element discretization. Additional details about the grid and the associated

462 discretization are provided in Appendix C. For simplicity, we confine ourselves to two  
 463 dimensions with coordinates  $(x, z)$  and velocities  $(u, w)$ . Generalization to three  
 464 dimensions should be clear (an example of a three-dimensional grid appropriate for our  
 465 purpose is discussed in Appendix C). Further, we present only the simpler case of direct  
 466 substitution for the basal boundary conditions in the variational functional, i.e., (15) or  
 467 (33). The remarks in this Section apply to both the standard and transformed Stokes  
 468 models; for example, the discrete pressure variable  $p$  may refer to either the standard  
 469 pressure  $P$  or the transformed pressure  $\tilde{P}$ .

470

471 Consider an arbitrary grid with a total of  $N = n_u + n_w + n_p$  unknown discrete  
 472 variables at appropriate nodal locations  $1 \leq i \leq N$ , with  $n_u$  horizontal velocity variables,  
 473  $n_w$  vertical velocity variables, and  $n_p$  pressure variables, such that

474 
$$U = \{U_1, U_2, \dots, U_N\}^T = \left\{ \left\{ u_1, u_2, \dots, u_{n_u} \right\}, \left\{ w_1, w_2, \dots, w_{n_w} \right\}, \left\{ p_1, p_2, \dots, p_{n_p} \right\} \right\}^T = \{u, w, p\}^T \quad (44)$$

475 is the vector containing all the unknown discrete variables. These are the degrees of  
 476 freedom of the model. If using Lagrange multipliers for basal boundary conditions then  
 477 discrete variables corresponding to  $\lambda_2, \Lambda$  must be added. Variables are expanded in

478 terms of shape functions  $N_i^k(\mathbf{x})$  associated with each nodal variable  $i$  in each element

479  $k$ , such that  $U^k(\mathbf{x}) = \sum_i U_i N_i^k(\mathbf{x})$  is the spatial variation of all the variables in element

480  $k$ . The summation is over all variable nodes located in element  $k$ . Shape functions

481 associated with a given node may differ depending on the variable (i.e.,  $u, w$ , or  $p$ ).

482 Substituting into the functional, (15) or (33), integrating and assembling the contributions

483 of all elements, we obtain a discretized variational functional in terms of the nodal

484 variable vectors  $u, w, p$ , as follows

485 
$$\mathcal{A}(u, w, p) = \sum_k \mathcal{A}^k(u, w, p), \quad (45)$$

486 where  $\mathcal{A}^k(u, w, p)$  is the local functional evaluated by integrating over element  $k$ . Since

487 the term in the functional involving the product of pressure and divergence of velocity is

488 linear in pressure and velocity, and the term responsible for gravity forcing is linear in

489 velocity, the functional (45) may be written in matrix form as follows

490 
$$\mathcal{A}(u, w, p) = \mathcal{M}(u, w) + p^T \left( M_{UP}^T u + M_{WP}^T w \right) + u^T F_U + w^T F_W, \quad (46)$$

491 where the shorthand notation from (44) is used, i.e.,  $u = \{u_1, u_2, \dots, u_{n_u}\}^T$ , etc. Parentheses  
 492 indicate a functional dependence on the indicated variables. Comparison with (15) and  
 493 (33) indicates that  $\mathcal{M}(u, w)$  is a nonlinear positive-definite function of the velocity  
 494 components  $u, w$ ,  $M_{UP}, M_{WP}$  are constant  $n_u \times n_p$  and  $n_w \times n_p$  matrices, respectively,  
 495 arising from the incompressibility constraint in the functional, and  $F_U, F_W$  are constant  
 496 gravity forcing vectors, of dimension  $n_u$  and  $n_w$ , respectively. Note that  $F_U = 0, F_W \neq 0$   
 497 in the standard Stokes model and  $F_U \neq 0, F_W = 0$  in the transformed Stokes model. The  
 498 discrete functional  $\mathcal{M}(u, w)$  differs in the two models but it remains positive-definite in  
 499 both, which has important consequences, as will be seen in Appendix D.

500

501 Discrete variation of the functional corresponds to partial differentiation with  
 502 respect to each of the discrete variables in  $U$ . Thus, the discrete Euler-Lagrange  
 503 equations that correspond to the u-momentum, w-momentum, and continuity equations,  
 504 respectively, are given by

$$505 \quad R(u, w, p) = \begin{bmatrix} R_U(u, w, p) \\ R_W(u, w, p) \\ R_p(u, w) \end{bmatrix} = \begin{bmatrix} \mathcal{M}_U(u, w) + M_{UP}p + F_U \\ \mathcal{M}_W(u, w) + M_{WP}p + F_W \\ M_{UP}^T u + M_{WP}^T w \end{bmatrix} = 0, \quad (47)$$

506 where  $R(u, w, p)$  is the residual vector (actually, it is the negative of the usual definition  
 507 of the residual) with components  $R_U(u, w, p) = \partial \mathcal{A} / \partial u$ ,  $R_W(u, w, p) = \partial \mathcal{A} / \partial w$ , and  
 508  $R_p(u, w) = \partial \mathcal{A} / \partial p$ . The functionals  $\mathcal{M}_U(u, w) = \partial \mathcal{M} / \partial u$ ,  $\mathcal{M}_W(u, w) = \partial \mathcal{M} / \partial w$  are  
 509 nonlinear vectors of dimension  $n_u$  and  $n_w$ , respectively. Altogether, (47) is a set of  $N$   
 510 equations for the  $N$  unknown discrete variables  $U_i$ . As explained previously, all  
 511 boundary conditions are already included in functional (46), and therefore are also  
 512 included in the discrete Euler-Lagrange equations (47).

513

514 Since the overall system (47) is nonlinear, it is typically solved using Newton-  
 515 Raphson iteration, expressed in matrix notation as follows

$$516 \quad M(u^K, w^K) \Delta U^{K+1} + R(u^K, w^K, p^K) = 0, \quad (48)$$

517 where  $K$  is the iteration index,  $M(u, w) = \partial^2 \mathcal{A}(U) / \partial U_i \partial U_j$  is a symmetric  $N \times N$   
 518 Hessian matrix, and  $\Delta^{K+1}$  is the column vector given by

519 
$$\Delta U^{K+1} = [u^{K+1} - u^K, w^{K+1} - w^K, p^{K+1} - p^K]^T.$$

520 Given  $U_i^K$  from the previous iteration, (48) is a linear matrix equation that is solved for  
 521 the  $N$  new variables  $U_i^{K+1}$  at each iteration. In view of (46) and (47), the Hessian matrix  
 522  $M(u, w)$  may be decomposed into several submatrices, as follows

523 
$$M(u, w) = \begin{bmatrix} M_{UU}(u, w) & M_{UW}(u, w) & M_{UP} \\ M_{UW}^T(u, w) & M_{WW}(u, w) & M_{WP} \\ M_{UP}^T & M_{WP}^T & 0 \end{bmatrix}. \quad (49)$$

524 Submatrices  $M_{UW}(u, w) = \partial^2 \mathcal{M} / \partial u \partial w$ , etc., depend nonlinearly on  $u, w$ . Thus,  
 525  $M_{UU}(u, w)$ ,  $M_{WW}(u, w)$  are square  $n_u \times n_u$ ,  $n_w \times n_w$  matrices, respectively, while  
 526  $M_{UW}(u, w)$  is a rectangular  $n_u \times n_w$  matrix since  $n_u, n_w$  may not be equal. As noted  
 527 earlier,  $M_{WP}$  is a  $n_w \times n_p$  matrix and therefore not square unless  $n_p = n_w$ . Additionally,  
 528  $M_{UU}(u, w)$  and  $M_{WW}(u, w)$  are symmetric by definition.

529

## 530 4.2 Blatter-Pattyn Discretizations

531

532 For completeness, we express the Blatter-Pattyn approximations from §3.4 in matrix  
 533 form, as follows

534 (1) The standard Blatter-Pattyn model from §3.4.1 takes the simple form

535 
$$R^{BP}(u) = \mathcal{M}_U(u, 0) + F_U = 0, \quad (50)$$

536 with the corresponding Newton-Raphson iteration given by

537 
$$M^{BP}(u^K) \Delta u^{K+1} + R^{BP}(u^K) = 0, \quad (51)$$

538 where the Blatter-Pattyn Hessian matrix is  $M^{BP}(u) = M_{UU}(u, 0)$ .



539 (2) The extended Blatter-Pattyn approximation from §3.4.2 becomes

$$540 \quad R^{EBP}(u, w, p) = \begin{bmatrix} \mathcal{M}_U(u, 0) + M_{UP}p + F_U \\ M_{WP}p \\ M_{UP}^T u + M_{WP}^T w \end{bmatrix} = 0, \quad (52)$$

541 and the Newton-Raphson iteration is given by

$$542 \quad M^{EBP}(u^K) \Delta U^{K+1} + R^{EBP}(u^K, w^K, p^K) = 0, \quad (53)$$

543 where the associated Hessian matrix is

$$544 \quad M^{EBP}(u) = \begin{bmatrix} M_{UU}(u, 0) & 0 & M_{UP} \\ 0 & 0 & M_{WP} \\ M_{UP}^T & M_{WP}^T & 0 \end{bmatrix}. \quad (54)$$

545

### 546 4.3 Solvability Issues

547

548 We now consider the solution of the three linear matrix problems (48), (51), (53). While  
 549 there is no issue in the continuous case, there may be problems in the discrete case  
 550 depending on the choice of the grid and the finite elements, as noted earlier.

551

#### 552 4.3.1 Solvability of the Standard and Transformed Stokes Models

553

554 The Hessian matrix in the standard and transformed Stokes cases, (49), has the form

$$555 \quad M(u, w) = \begin{bmatrix} A & B \\ B^T & 0 \end{bmatrix}, \quad (55)$$

556 where

$$557 \quad A = A^T = \begin{bmatrix} M_{UU}(u, w) & M_{UW}(u, w) \\ M_{UW}^T(u, w) & M_{WW}(u, w) \end{bmatrix}, \quad B = \begin{bmatrix} M_{UP} \\ M_{WP} \end{bmatrix}, \quad B^T = \begin{bmatrix} M_{UP}^T & M_{WP}^T \end{bmatrix}.$$

558 The general form (55) is characteristic of Stokes-type problems, or more generally, of  
 559 constrained minimization problems using Lagrange multipliers. In finite element  
 560 terminology these are “mixed” problems, meaning that velocity components and the  
 561 pressure occupy different finite element spaces, or else they are “saddle point” problems  
 562 since the Hessian matrix  $M(u, w)$  is symmetric but indefinite, with both positive and

563 negative eigenvalues. This can give rise to solution instabilities. To avoid this, elements  
564 that are to be used must satisfy the so-called inf-sup or LBB condition constraining the  
565 matrix  $B$  in (55). There is a very large literature on the subject, e.g., Elman et al. (2014).  
566 Testing for stability is not trivial. Both the standard and transformed Stokes models are  
567 subject to these issues and in general must use inf-sup-stable finite elements. An  
568 example of an inf-sup stable element is the popular second-order Taylor-Hood P2-P1  
569 element with piecewise quadratic velocity and linear pressure (Hood and Taylor, 1973).  
570 Both the standard and transformed Stokes models are stable using the Taylor-Hood  
571 element. Some results involving the Taylor-Hood element are shown in Fig. 13 for Test  
572 B, one of the test problems described in Appendix B that corresponds to Exp. B from the  
573 ISMIP–HOM model intercomparison (Pattyn et al., 2008).

574

#### 575 **4.3.2 Solvability of the Standard Blatter-Pattyn Model**

576

577 The standard Blatter-Pattyn approximation is not subject to these stability issues since  
578 pressure, the Lagrangian multiplier, is absent in (51). As a result, the standard Blatter-  
579 Pattyn variational formulation (34) is actually a well-behaved and stable positive-definite  
580 minimization or optimization problem.

581

#### 582 **4.3.3 Solvability of the Extended Blatter-Pattyn Model**

583

584 We noted earlier that the transformed Stokes model works well using the Taylor-Hood  
585 element in Test B. Since the extended Blatter-Pattyn model has the same structure as the  
586 transformed full-Stokes model and yields the same solution for horizontal velocity as the  
587 standard Blatter-Pattyn model, one might expect its discrete implementation to behave  
588 well. However, the extended Blatter-Pattyn model fails badly in this problem, with  
589 nonsensical results for the vertical velocity. This may be because there is an additional  
590 requirement for the stability of a Stokes-type problem that is not met in this case, namely,  
591 the matrix  $A$  in (55) must be elliptic on the whole  $u, w$  space (Auricchio et al., 2017).  
592 However, there is a much simpler explanation. Consider the vertical momentum  
593 equation, the second of the extended Blatter-Pattyn model equations from (52). As is  
594 seen in §3.4.2 or from the second of the equations in (52) in the extended Blatter-Pattyn  
595 approximation, this equation is a decoupled linear system for the pressure. Since the  
596 equation involves the  $M_{wp}$  matrix, we have a decoupled set of  $n_w$  equations that needs to

597 be solved for the  $n_p$  pressure variables. This is not possible unless the matrix  $M_{wp}$  is  
598 square. For the same reason, the third of the equations in (52) cannot be solved for  $w$   
599 unless matrix  $M_{wp}^T$  is invertible. In other words, the extended Blatter-Pattyn model (52)  
600 only works when  $n_w = n_p$ , which is not the case in a Taylor-Hood discretization. This is  
601 because in finite element discretizations of Stokes problems, the pressure approximation  
602 is typically one degree lower than the velocity approximation, which leads to fewer  
603 pressure variables than velocity variables. In the case of the Taylor-Hood element, the  
604 difference is very large and we have  $n_w \gg n_p$  (see §7 for more details). This means that  
605 in the extended Blatter-Pattyn model vertical velocity is greatly underdetermined, which  
606 accounts the problem in the Taylor-Hood calculation. This problem also manifests itself  
607 in Taylor-Hood discretizations of Stokes models but to a much lesser extent. For  
608 example, mass is poorly conserved in the Taylor-Hood discretization of the standard  
609 Stokes model (Boffi et al., 2012). In the transformed Stokes case there tend to be  
610 velocity oscillations that tend to go away when using a grid in which  $n_p = n_w$  (see Fig. 13,  
611 Panels E and F).

612

#### 613 4.3.4 The Solvability Condition

614

615 Summarizing, the extended Blatter-Pattyn approximation is problematic unless we have

$$616 \quad n_p = n_w. \quad (56)$$

617 In addition, the resulting square matrix  $M_{wp}$  must be non-singular, which we assume to be  
618 the case for a reasonable finite element discretization. This makes it possible to solve for  
619 the pressure in the extended Blatter-Pattyn system (52) because  $M_{wp}$  is square and  
620 invertible. We henceforth refer to (56), together with non-singularity, as the solvability  
621 condition for the pressure. This is a characteristic or a property associated with the  
622 discrete grid and the boundary conditions. In Appendix C, we consider several grids that  
623 exhibit this property. The specific solvability condition given by (56) applies when direct  
624 substitution is used for basal boundary conditions. The number of unknown pressures  $n_p$   
625 must be augmented if Lagrange multipliers are used and (56) becomes  $n_p + \lambda_z + \Lambda = n_w$   
626 (See Appendix C, §C2).

627

628 The solvability condition has an additional implication. If matrix  $M_{WP}$  is square  
629 and invertible due to (56), then its transpose  $M_{WP}^T$  is also square and invertible. This  
630 implies that the continuity equation in (47) and (52), that is,

$$631 \quad M_{UP}^T \mathbf{u} + M_{WP}^T w = 0, \quad (57)$$

632 is solvable for the vertical velocity  $w$  in terms of the horizontal velocities, as follows

$$633 \quad w(\mathbf{u}) = -M_{WP}^{-T} M_{UP}^T \mathbf{u}, \quad (58)$$

634 where the matrix  $M_{WP}^{-T}$  is defined by

$$635 \quad M_{WP}^{-T} = \left( M_{WP}^T \right)^{-1} = \left( M_{WP}^{-1} \right)^T. \quad (59)$$

636 Note that (58) is the discrete form of equation (43). Thus, since the invertibility of  $M_{WP}$   
637 implies the invertibility of  $M_{WP}^T$ , the solvability condition (56) implies the solvability of  
638 the continuity equation (58), and vice-versa. As we shall see, this property is not just a  
639 useful property but it is necessary for the new Stokes approximations that improve on the  
640 Blatter-Pattyn approximation, as discussed in §6.2.

641  
642 Perhaps the main reason for the importance of the solvability condition is  
643 demonstrated in Appendix D. Appendix D shows that a variational principle that  
644 complies with the solvability condition is equivalent to an optimization or minimization  
645 problem, which is sufficient for the stability of the corresponding Stokes model. Thus,  
646 for example, the extended Blatter-Pattyn model fails with a Taylor-Hood P2-P1 grid,  
647 which does not satisfy the solvability condition, but works well with a variant, the P2-E1  
648 grid, shown in Fig. 13A, that does satisfy the solvability condition. Several finite  
649 elements that satisfy the condition are presented in Appendix C. One particular element,  
650 the P1-E0 element, is particularly useful for use with the transformed Stokes model  
651 because the solvability condition is satisfied locally, i.e., along individual vertical grid  
652 lines, as shown in Appendix C. This element is used in most of the 2D test problems  
653 featured here.

654

## 655 **5. Comparison of the Standard and Transformed Stokes Models**

656

657 To compare the standard and transformed Stokes models we use two 2D test problems,  
658 namely, Exp. B from the ISMIP-HOM benchmark (Pattyn et al, 2008), and Exp. D\*, a  
659 modified version of Exp. D from the ISMIP-HOM suite. A description of these tests is



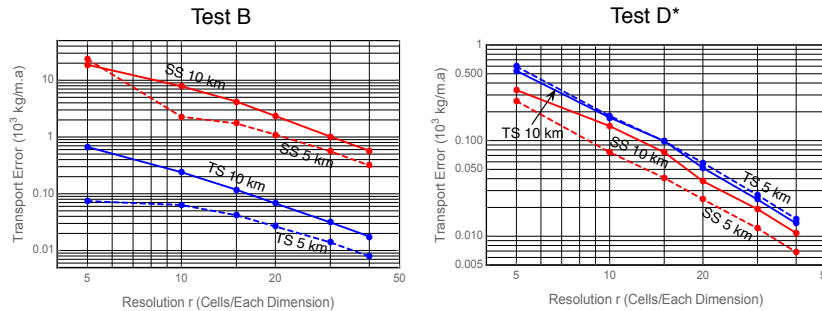
660 provided in Appendix B, where they are referred to as Test B and Test D\*. Test B  
661 involves no-slip boundary conditions on a sinusoidal bed, and Test D\* evaluates sliding  
662 of the ice sheet along a flat bed in the presence of sinusoidal friction. The tests are  
663 discretized using P1-E0 elements on a regular grid composed of  $n$  quadrilaterals in the  
664  $x$ -direction and  $m$  quadrilaterals in the  $z$ -direction, with each quadrilateral divided into  
665 two triangles as illustrated in Figs. C3 and described in Appendix D. The results  
666 presented in this Section are for a relatively coarse 40x40 grid, i.e.,  $m = n = 40$ , except  
667 when we consider the convergence of the models with grid refinement.

### 669 **5.1 Convergence of Solutions with Grid Refinement**

670  
671 We first look at the convergence of the transformed and standard Stokes models as the  
672 grid is refined in Fig. 3. In particular, we look at the convergence of ice transport through  
673 a vertical cross section of the ice sheet at  $x = L$ . The ice transport  $T$  is defined by

$$674 \quad T = \int_{z_b}^{z_s} u(L, z) dz, \quad (60)$$

675 where the vertical profile  $u(L, z)$  is plotted in Fig. 4 for several cases at the 40x40  
676 resolution. Fig. 3 plots the absolute value of the transport error  $E = ||T - T_R||$  as a function  
677 of the resolution  $r$ , where  $r$  is the number of quadrilaterals in either direction (since  
678  $r = m = n$ ) and  $T_R$  is the converged value of the transport obtained by Richardson  
679 extrapolation using the two highest resolution values. The transport is evaluated at  
680 various resolutions  $r = 5, 10, 15, 20, 30, 40$ , and plotted at two domain lengths,  $L = 5$  and  
681 10 km. Trying to estimate the rate of convergence in this way is highly uncertain, as  
682 discussed in §7, but estimating the error is a more reasonable thing to do. Both models  
683 are consistent with second order convergence, as expected from the use of linear  
684 elements, but they behave quite differently in the two test problems. The transformed  
685 Stokes model (TS) is some two orders of magnitude more accurate at all resolutions than  
686 the standard Stokes model (SS) in Test B calculations although they start from the same  
687 initial conditions. However, the accuracy of the two models is quite similar in Test D\*  
688 calculations, with the SS error actually somewhat smaller than the TS error. This is  
689 confirmed when we compare the details of the  $u$ -velocity solutions in Figs. 4 and 5 at the  
690 40x40 resolution. The TS and SS profiles differ noticeably from each other but are quite  
691 similar in the Test D\* case. However, the standard and transformed Stokes models do  
692 eventually converge to the same solution.



693

694

**Figure 3.** Convergence of ice transport in Tests B and D\* with grid refinement.

695

Transformed Stokes plots are in blue and standard Stokes plots are in red.

696

697

### 5.2 The Vertical Profile of Solutions

698

699

Fig. 4 shows the vertical profiles of the horizontal velocity  $u$  at  $x = L$  for the 40x40 resolution in the transformed and standard Stokes models. There is a noticeable difference in the two profiles in Test B, as is to be expected from Fig. 3 results where we see that the SS calculation is not yet as well converged as the TS case at this resolution. Also shown in Fig. 4 are profiles from the two frictional sliding problems, Tests D and D\*. The Test D profile, i.e., Exp. D from the ISMIP-HOM benchmark, is almost vertically constant, indicating that the originally chosen value for basal friction is too small, i.e., more appropriate for a shallow-shelf approximation. This motivated the modification of Test D to Test D\*, as described in Appendix B. In contrast to the Test B case, the standard and transformed frictional Test D and D\* plots cannot be visually distinguished from each other, as might be expected from the similar error convergence for the Test D\* results in Fig. 3.

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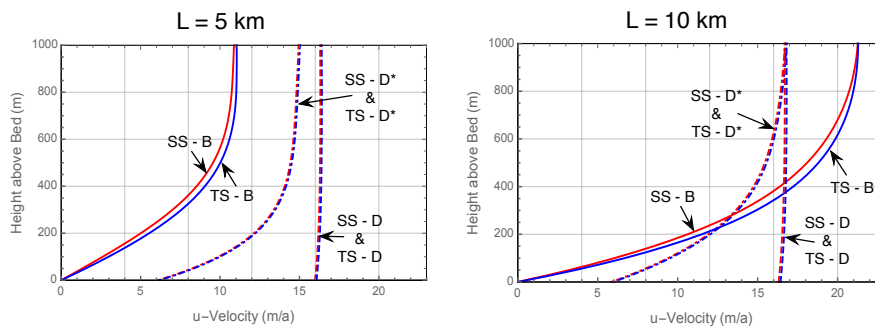
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**Figure 4.** The  $u$ -velocity profile at location  $x = L$  as a function of height from the bed.

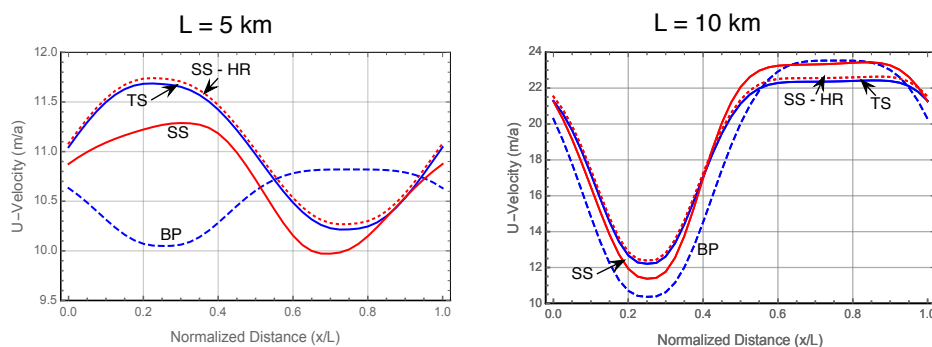
713



714 **5.3 The Upper Surface Horizontal Velocity**

715

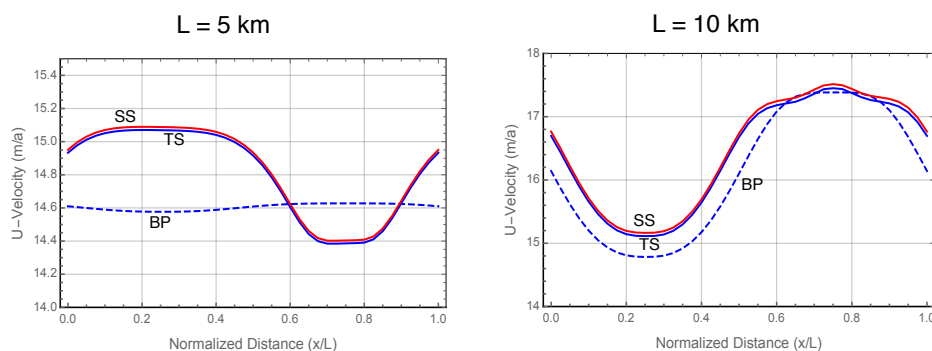
716 Figs. 5 and 6 show the  $u$ -velocity at the upper surface at the  $40 \times 40$  resolution for Tests B  
 717 and  $D^*$ , respectively. This is the basic benchmark used in ISMIP-HOM to compare the  
 718 different ice sheet models. Here we compare four cases: the standard Stokes model (SS),  
 719 the transformed Stokes model (TS), the Blatter-Pattyn (BP) model, and for reference, the  
 720 very high resolution full-Stokes calculation “oga1” presented in the ISMIP-HOM paper  
 721 (SS-HR). The SS-HR calculation is also available independently in Gagliardini and  
 722 Zwinger (2008). Results are presented for two domain lengths,  $L = 5$  km and 10 km, to  
 723 observe the behavior of the SS and TS models in the aspect ratio range where the Blatter-  
 724 Pattyn model begins to fail.



725

726 **Figure 5.** Upper surface  $u$ -velocity,  $u(x, z_s)$  - Test B, No-slip boundary conditions.

727



728

729 **Figure 6.** Upper surface  $u$ -velocity,  $u(x, z_s)$  - Test  $D^*$ , Modified frictional sliding case.

730

731 The TS and the SS-HR plots in Fig. 5 lie on top of one another (the SS-HR plot  
 732 (dotted) has been slightly offset upwards for clarity), indicating that the transformed



733 Stokes model is already fully converged, and confirming that the standard and  
734 transformed Stokes models do indeed converge to the correct Stokes solution. We again  
735 observe that the SS results are not yet converged in Test B at this resolution, particularly  
736 at  $L = 5$  km. As also seen in the ISMIP-HOM benchmark paper, the Blatter-Pattyn  
737 calculation (BP) shows large deviations from the Stokes results, especially so at  $L = 5$   
738 km where surface velocity is entirely out of phase with the Stokes results. Test D\*  
739 frictional sliding results follow a similar pattern in Fig. 6. Since convergence of the SS  
740 and TS models is very similar in the frictional case, the SS and TS plots overlie one  
741 another (the SS plot has been offset slightly upwards for visibility), confirming that the  
742 two Stokes models converge to the same solution. As was seen in Test B, the Blatter-  
743 Pattyn error is quite large at  $L = 10$  km, and dramatically so at  $L = 5$  km.

744

## 745 **6. Some Applications of the Transformed Stokes Model**

### 746 **6.1 Adaptive Switching between Stokes and Blatter-Pattyn Models**

747

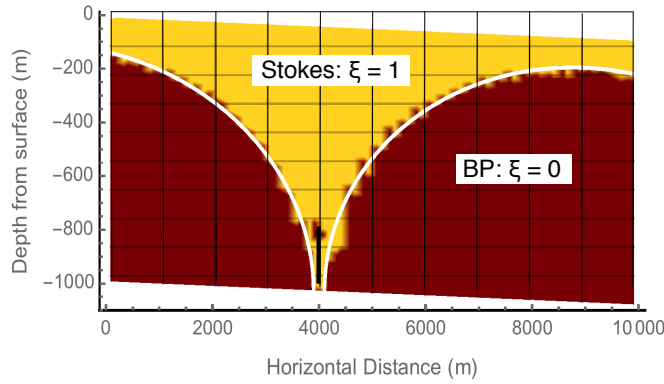
748 One way of reducing the cost of a full Stokes calculation is to use it adaptively with a  
749 cheaper approximate model in a given problem. That is, one may use the cheaper model  
750 in those parts of a problem where it is accurate, and the more expensive full Stokes model  
751 where the approximate model loses accuracy. One example of such an adaptive approach  
752 is the tiling method by Seroussi et al. (2012). However, there are drawbacks to such  
753 methods, such as the difficulty of incorporating two or more presumably quite different  
754 models into a single model, and the additional complexity of a transition zone in order to  
755 couple the disparate models.

756

757 Using the transformed Stokes model in such an adaptive role is attractive because  
758 it may be switched between the Stokes and Blatter-Pattyn cases simply by switching the  
759 parameter  $\xi \in \{0,1\}$  between its two values. To avoid complications and more difficult  
760 programming it is essential that both the Stokes and the Blatter-Pattyn parts of the code  
761 have the same number of discrete variables. This implies that the extended Blatter-Pattyn  
762 approximation ( $\hat{\xi} = 1$ ) must be used, which therefore implies the use of a grid that  
763 satisfies the solvability condition for reasons discussed in §4 and Appendix C. To do  
764 this, we will discretize using the P1-E0 element. To demonstrate the idea of adaptive  
765 switching with a transformed Stokes model, we introduce a new test problem, Test O,  
766 described in Appendix B and illustrated in Fig. B1. This consists of an inclined ice slab  
767 whose movement is obstructed by a thin obstacle protruding 20% of the ice depth up



768 from the bed. No-slip boundary conditions are applied along the bed and on the obstacle  
 769 itself. Because of the localized nature of the obstacle, the conditions for the Blatter-  
 770 Pattyn approximation to be valid, (38), must fail near the obstacle and therefore the full  
 771 Stokes model is needed for good accuracy, at least locally.



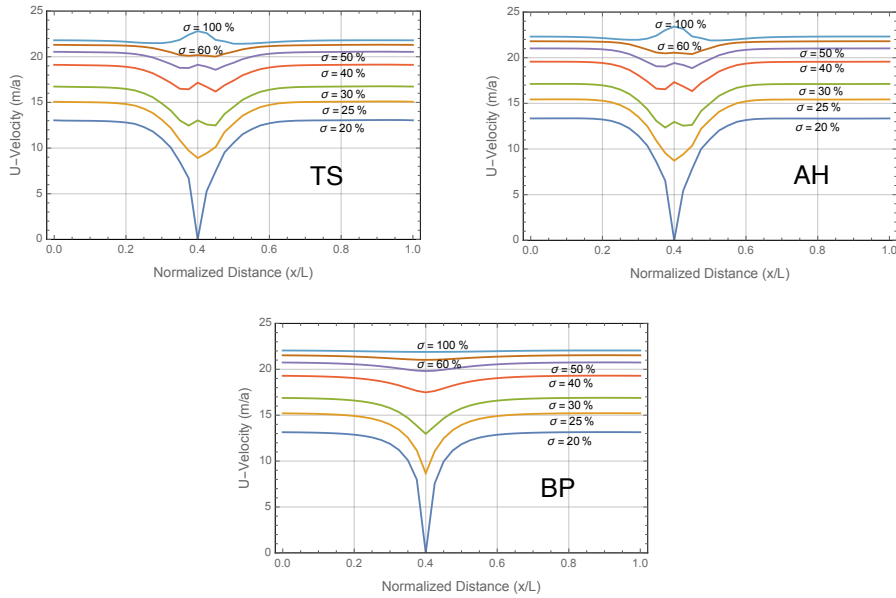
772  
 773 **Figure 7.** Mask function (white curve,  $z = F_M(x)$ ) to indicate where the Stokes and BP  
 774 models are activated in the adaptive hybrid 20% obstacle test problem. The dark brown  
 775 region delineates the region where  $|\partial w/\partial x| \leq 0.1|\partial u/\partial z|$  in a Blatter-Pattyn calculation.

776  
 777 To implement this idea, we first use a Blatter-Pattyn calculation to outline regions  
 778 where  $|\partial w/\partial x| \leq 0.1|\partial u/\partial z|$ , approximately localizing where the Blatter-Pattyn  
 779 approximation is valid. This determines a mask function  $z = F_M(x)$ , illustrated in Fig. 7  
 780 by the two white curves, that specifies where the two models must be used. Defining the  
 781 centroid of a triangular element by  $(x_c, z_c)$ , the code makes the following selection in  
 782 each element,

783

$$\begin{aligned} z_c \leq F_M(x_c) &\Rightarrow \text{Set } \xi = 0, \text{ i.e., the Blatter-Pattyn region,} \\ z_c > F_M(x_c) &\Rightarrow \text{Set } \xi = 1, \text{ i.e., the Stokes region.} \end{aligned}$$

784 Somewhat counterintuitively, the Stokes region occupies the upper part of the domain in  
 785 Fig. 7 and includes the obstacle, while the Blatter-Pattyn region occupies much of the  
 786 bottom part of the domain. It would be possible to introduce a transition zone, e.g.,  
 787  $0 \leq \xi(x, z) \leq 1$ , but this was not deemed necessary and it was not done in the present  
 788 calculation.



789

790 **Figure 8.** Comparing results for the Transformed Stokes (TS, i.e., the exact Stokes),  
 791 the Adaptive-Hybrid (AH), and the Blatter-Pattyn (BP) models for Test O.

792

793 The Adaptive-Hybrid results are shown in Fig. 8, which shows curves of the  
 794 horizontal velocity  $u$  at seven different vertical positions specified as a percentage of the  
 795 distance between top and bottom, such that  $\sigma = 100\%$  is at the top surface. The top right  
 796 panel shows the results for the adaptive-hybrid model. For comparison, the top left panel  
 797 and the bottom panel show results for the full Stokes and the Blatter-Pattyn calculations,  
 798 respectively. All calculations are at the  $40 \times 40$  resolution. The Adaptive-Hybrid results  
 799 are very similar to the full Stokes results, reproducing most features of the velocity  
 800 profiles, including the velocity bump at the top surface, indicating that even the top  
 801 surface feels the presence of the obstacle. The Blatter-Pattyn results are much less  
 802 accurate; they completely miss the details of the flow near the obstacle. We also  
 803 calculate a measure of the error relative to the transformed Stokes results, the overall  
 804 RMS u-Error, defined as follows

805

$$\text{RMS u-Error} = \sqrt{\sum_{k=1}^{n_u} (u_k - u_k^{TS})^2 / n_u}, \quad (61)$$

806

where  $u_k^{TS}$  is the transformed Stokes horizontal velocity discrete variable. The overall  
 807 RMS u-Error in the Blatter-Pattyn case is 0.493 m/a while the Adaptive-Hybrid error is  
 808 0.440 m/a, smaller in the Blatter-Pattyn case, as expected, but the difference is not big



809 and not as striking as the visual differences in Fig. 8. Nevertheless, the adaptive-hybrid  
 810 method can be judged successful by the results presented in Fig. 8 alone. Unfortunately,  
 811 a reasonable estimate of the computational cost savings cannot be made because of the  
 812 small-scale nature of these calculations that were carried out on a personal computer.

813

## 814 **6.2. Two Stokes Approximations Beyond Blatter-Pattyn**

815

816 As shown in §3.4, simply setting  $w=0$  in the second invariant  $\tilde{\epsilon}^2$  in the transformed  
 817 functional  $\tilde{\mathcal{A}}$ , given by (28) and (33), respectively, results in the Blatter-Pattyn system of  
 818 equations. This suggests that approximating the vertical velocity  $w$  in the transformed  
 819 functional would be a good way to create approximations that improve on the Blatter-  
 820 Pattyn approximation since providing no information at all, i.e.,  $w=0$ , already produces  
 821 an excellent approximation. We will look at only two such methods in this Section even  
 822 though many other variations are possible. The first method, to be called the BP+  
 823 approximation, improves the Blatter-Pattyn approximation simply by using a lagged  
 824 value of the vertical velocity in the functional (33). It is implemented using a  
 825 combination of Newton and Picard iterations such that at each Newton iteration the  
 826 variational functional is evaluated using the known vertical velocity  $w^K$  from the  
 827 previous iteration, where  $K$  is the iteration index. The vertical velocity,  $w^K = w(u^K)$ , is  
 828 obtained by using (58) together with a grid that is consistent with an invertible continuity  
 829 equation, such as the P1-E0 grid from Appendix C. The second method, to be called the  
 830 Dual-Grid approximation, approximates the transformed Stokes model by discretizing the  
 831 continuity equation on a coarser grid. Since vertical velocity  $w$  is to be determined by  
 832 inverting the continuity equation, this has the effect of approximating the vertical velocity  
 833 while at the same time reducing the number of pressure and vertical velocity variables.  
 834 The degree of grid coarsening determines the accuracy of the resulting approximation.

835

### 836 **6.2.1 An Improved Blatter-Pattyn or BP+ Approximation**

837

838 To prepare, we introduce a pair of 2D variational quasi-functionals,  $\tilde{\mathcal{A}}_{PS1}[u, w]$  and  
 839  $\tilde{\mathcal{A}}_{PS2}[\tilde{P}]$ . Noting that  $\tilde{P}=0$  in the Blatter-Pattyn approximation, we drop the pressure  
 840 term from the transformed functional (33) and define a new functional,

$$841 \quad \tilde{\mathcal{A}}_{PS1}[u, w] = \int_V dV \left[ \frac{4n}{n+1} \eta_0 (\tilde{\varepsilon}^2)^{(1+n)/2n} + \rho g u \frac{\partial z_s}{\partial x} \right] + \frac{1}{2} \int_{S_{B2}} dS \beta(x) \left( u^2 + \zeta \left( u n_x^{(b2)} / n_z^{(b2)} \right)^2 \right), \quad (62)$$

842 where

$$843 \quad \tilde{\varepsilon}^2 = \left( \frac{\partial u}{\partial x} \right)^2 + \frac{1}{4} \left( \frac{\partial u}{\partial z} + \frac{\partial w}{\partial x} \right)^2. \quad (63)$$

844 Since the continuity equation has been eliminated, we introduce incompressibility  
 845 separately by defining another functional,

$$846 \quad \tilde{\mathcal{A}}_{PS2}[p] = \int_V dV p \left( \frac{\partial u}{\partial x} + \frac{\partial w}{\partial z} \right). \quad (64)$$

847 Since direct substitution is used for boundary conditions, then (9) and (14) are the  
 848 appropriate basal boundary conditions needed to specify  $w$  in (64); no boundary  
 849 condition is required for the pressure. Here we are effectively viewing the pressure  $p$  as  
 850 a “test function” in the finite element sense. This gives us great flexibility to create  
 851 elements that satisfy the solvability condition (56). In a triangulation, for example, some  
 852 pressures may be assigned to every two triangles, as in a P1-E0 grid, while others may be  
 853 assigned to a single triangle to achieve an equal number of pressure and vertical velocity  
 854 variables.

855

856 The discrete variation of  $\tilde{\mathcal{A}}_{PS1}[u, w]$  with respect to  $u$ , results in a set of  $n_u$  Euler-  
 857 Lagrange equations,

$$858 \quad \hat{R}_U(u, w) = \frac{\partial \tilde{\mathcal{A}}_{PS1}(u, w)}{\partial u} = M_U(u, w) + F_U = 0. \quad (65)$$

859 This may be recognized as the standard Blatter-Pattyn model, (50), when  $w = 0$ . The

860 discrete variation of  $\tilde{\mathcal{A}}_{PS2}[p]$  with respect to  $p$ , results in the continuity equation, (57),

$$861 \quad \hat{R}_p(u, w) = \frac{\partial \tilde{\mathcal{A}}_{PS2}(p)}{\partial p} = M_{UP}^T u + M_{WP}^T w = 0. \quad (66)$$

862 These two systems are now combined to form the BP+ approximation, as follows

$$863 \quad \hat{R}(u, w) = \left[ \hat{R}_U(u, w), \hat{R}_p(u, w) \right]^T = 0. \quad (67)$$

864 This is a single system of  $n_u + n_p$  equations to determine the  $n_u + n_w$  discrete velocities

865  $u, w$ , implying that (67) is viable only on grids satisfying the solvability condition,

866  $n_p = n_w$ . Just as in the standard Blatter-Pattyn approximation in §3.4.1, the vertical  
 867 momentum equation is missing, but instead of neglecting  $w$ , the vertical velocity is now  
 868 obtained consistently from the continuity equation.

869

870 There are two ways of solving the BP+ system (67), as follows

871 (1) BP+, Newton/Picard iteration version:

872 If  $w = \hat{w}(x_i)$  is some arbitrary specified function of position, then (65) becomes a  
 873 nonlinear set of  $n_u$  equations that may be solved for the horizontal velocity  $u$  using  
 874 Newton iteration, as follows

875 
$$\hat{M}_{UU}(u^K, \hat{w})\Delta u + \hat{R}_U(u^K, \hat{w}) = 0, \quad (68)$$

876 where  $\hat{M}_{UU}(u, \hat{w}) = \partial \mathcal{M}_U(u, \hat{w}) / \partial u$ ,  $\Delta u = u^{K+1} - u^K$ , and  $K$  is the iteration index. In

877 particular, if we choose  $\hat{w} = w^K$ , where  $w^K$  is the horizontal velocity from the previous

878 iteration (i.e.,  $w^K = w(u^K)$  from (58), where  $u^K$  is the horizontal velocity from the

879 previous iteration), we obtain the following Picard iteration:

Starting from  $\kappa = 0$ , choose an initial guess,  $u^0 \neq 0$ ,

Do:  $w^K = w(u^K) = M_{pw}^{-1} M_{pu} u^K$ ,

880 
$$\text{Solve } \hat{M}_{UU}(u^K, w^K)\Delta u + \hat{R}_U(u^K, w^K) = 0, \quad (69)$$

$u^{K+1} = u^K + \Delta u$ ,

$\kappa = \kappa + 1$ ,

Repeat until convergence.

881 The advantage of this method is that iteration is rapid since each iteration step is  
 882 equivalent to the short Newton step of the standard Blatter-Pattyn model, (36). On the  
 883 other hand, as a Picard iteration, its convergence is expected to be only linear.

884

885 (2) BP+, Quasi-variational, Newton iteration version:

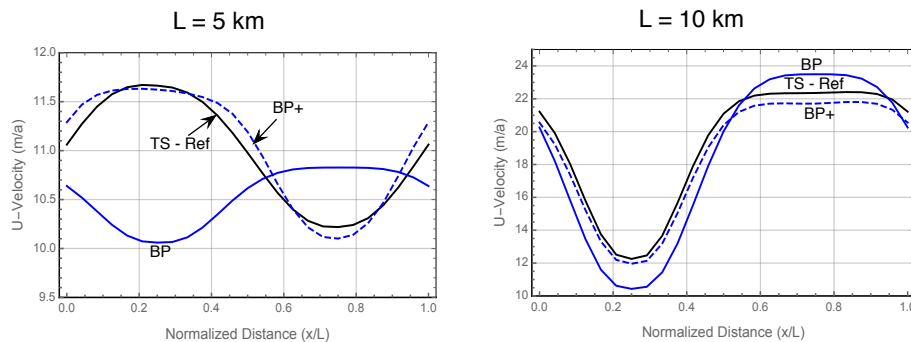
886 Although a variational principle does not exist, it is still possible to make use of  
 887 Newton-Raphson iteration to obtain second order convergence. To do this, we treat (67)  
 888 as a single multidimensional nonlinear system and solve it using Newton-Raphson  
 889 iteration, as follows

890 
$$\begin{bmatrix} \hat{M}_{UU}(u^K, w^K) & \hat{M}_{UW}(u^K, w^K) \\ M_{PU} & M_{PW} \end{bmatrix} \begin{bmatrix} \Delta u \\ \Delta w \end{bmatrix} + \begin{bmatrix} \hat{R}_U(u^K, w^K) \\ \hat{R}_P(u^K, w^K) \end{bmatrix} = 0, \quad (70)$$

891 where  $\hat{M}_{UU}(u, w) = \partial \hat{R}_U(u, w) / \partial u$  and  $\hat{M}_{UW}(u, w) = \partial \hat{R}_U(u, w) / \partial w$ . The convergence is  
 892 quadratic once in the basin of attraction but each iteration is more expensive than in the  
 893 Picard version because the linear system (70) is approximately double the size of the one  
 894 in (69). It remains to be seen which version proves to be preferable in practice.

895

896 Both BP+ versions converge to the same solution. Fig. 9 compares the upper  
 897 surface u-velocity from the improved Blatter-Pattyn (BP+) approximation to the standard  
 898 Blatter-Pattyn approximation and to a reference exact Stokes calculation. The RMS u-  
 899 Error of the BP+ approximation relative to the exact Stokes case is shown in Fig. 12. The  
 900 BP+ approximation is noticeably more accurate than the BP approximation, especially so  
 901 in the  $L = 5$  km case where the Blatter-Pattyn solution bears no resemblance to the  
 902 correct solution while the BP+ approximation retains very good accuracy. This is  
 903 confirmed by the RMS u-Error results in Fig. 12.



904

905 **Figure 9.** Comparing Approximations. Test B, Upper surface u-velocity.

906 TS-Ref: Transformed Stokes; BP: Blatter-Pattyn; BP+: Improved Blatter-Pattyn.

907

Resolution: 24x24.

908

909 The two versions depend either on solving the continuity equation to obtain

910  $w = w(u)$ , or the use of a grid that incorporates such a solvable continuity equation.

911 Solution of the continuity equation to obtain  $w$  may already be available for the purpose  
 912 of temperature advection in production code packages that either incorporate or are based  
 913 on the Blatter-Pattyn approximation. Thus, these new approximations, and particularly  
 914 the Newton/Picard version, may be especially attractive for use in such codes since they

915 substantially improve the accuracy of the basic Blatter-Pattyn model, as seen in Fig. 9, at  
916 little or no additional cost.

917

### 918 **6.2.2 A “Dual-Grid” Transformed Stokes Approximation**

919

920 In §6.2.1, the BP+ approximation was based on directly approximating or lagging the  
921 vertical velocity  $w$  in the second invariant  $\tilde{\epsilon}^2$  in the transformed functional  $\tilde{\mathcal{A}}$ . Here we  
922 take a different approach and instead approximate the continuity equation in the  
923 transformed Stokes model, which indirectly approximates  $w$ . To do this we discretize  
924 the continuity equation on a grid that is coarser than the one used for the momentum  
925 equations and then interpolate the vertical velocity to the appropriate locations on the  
926 finer grid. This reduces the number of unknown variables in the problem, making it  
927 cheaper to solve but hopefully without much loss of accuracy. As described in Appendix  
928 B, our test problem grids are logically rectangular, divided into  $n$  cells horizontally and  
929  $m$  cells vertically, thus allowing considerable freedom to specify the coarse grid. The  
930 coarse grid is constructed by dividing the fine grid into  $s$  equal segments in each  
931 direction. This presupposes that the integers  $n$  and  $m$  are each divisible by  $s$ , such that  
932 there are  $s^2$  coarse cells in total, with each coarse cell containing  $nm/s^2$  fine cells. The  
933 primary grid (i.e., the fine grid) was chosen to have  $n = m = 24$ , resulting in a reference  
934  $24 \times 24$  fine grid, so as to maximize the number of different coarse grids that may be  
935 used for this test. Coarse grids were constructed using  $s = 2, 3, 4, 6$ , and this resulted in  
936 fine/coarse grid combinations labeled by  $24 \times 12, 24 \times 8, 24 \times 6, 24 \times 4$ , respectively.

937 Similar to a P1-E0 fine grid, coarse grid vertical velocities  $w$  are located at vertices and  
938 pressures at vertical edges. Fig. 10 illustrates the case of a single coarse and four fine  
939 quadrilateral cells for a grid fragment with  $n = m = 2$  and  $s = 1$ . For the Test B problem,  
940 using direct substitution for basal boundary conditions, there will be  $nm$  u-variables and

941  $nm/s^2$  w- and p-variables each, for a total of  $nm(1 + 2/s^2)$  unknown variables,

942 considerably fewer than the  $3nm$  variables in the full resolution (i.e., fine grid) case,  
943 depending on the value of  $s$ . The coarse grid terms in the functional that are affected,

944  $\tilde{P}(\partial u/\partial x + \partial w/\partial z)$  and  $\partial w/\partial x$ , are computed using coarse grid variables and

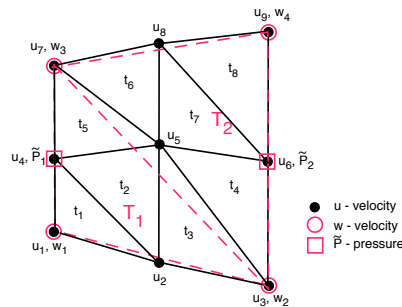
945 interpolated to the fine grid. We will consider two versions of the approximation

946 depending on how the coarse grid terms are calculated and distributed on the fine grid.

947

948 (1) Approximation A, Bilinear interpolation:

949 Referring to Fig. 10, the four velocities at the vertices of the coarse grid  
 950 quadrilateral, i.e.,  $u_1, u_3, u_7, u_9$  and  $w_1, w_2, w_3, w_4$ , are used to obtain  $u, w$  at the remaining  
 951 five vertices of the fine grid by means of bilinear interpolation. Thus, the five velocities  
 952  $u_2, u_4, u_5, u_6, u_8$  are obtained in terms of vertex velocities  $u_1, u_3, u_7, u_9$ , and similarly for the  
 953  $w$  velocities. The resulting complete set of fine grid variables, interpolated from coarse  
 954 grid variables, are used calculate the divergence  $D = (\partial u / \partial x + \partial w / \partial z)$  and the quantity  
 955  $\partial w / \partial x$  in each of the eight triangular elements  $t_1, t_2, \dots, t_8$  of the fine grid. Coarse grid  
 956 pressures  $\tilde{P}_1, \tilde{P}_2$  are associated with the coarse grid triangles  $T_1, T_2$ . The products  $\tilde{P}_1 D$  in  
 957 elements  $t_1, t_2, t_3, t_5$  and  $\tilde{P}_2 D$  in elements  $t_4, t_6, t_7, t_8$  are then accumulated over the entire  
 958 grid to obtain  $\tilde{P}(\partial u / \partial x + \partial w / \partial z)$  for use in the transformed functional  $\tilde{\mathcal{A}}$ . Similarly, the  
 959 quantity  $\partial w / \partial x$  is computed in the fine grid elements from coarse grid variables for use  
 960 in the second invariant  $\tilde{\mathcal{E}}^2$ .



961  
 962

963 **Figure 10.** A Sample of a Coarse/Fine P1-E0 Grid for the Dual-Grid Approximation.  
 964 Resolution:  $n = m = 2, s = 1$ . Coarse grid is in red, fine grid in black.

965

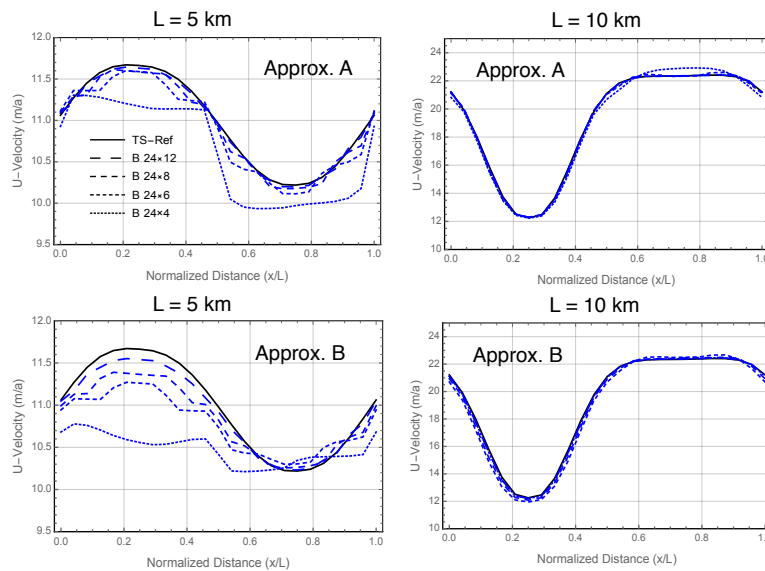
966 (2) Approximation B, Linear interpolation:

967 In this version, the three velocities at the vertices of the two coarse grid triangles  
 968  $T_1$  and  $T_2$ , i.e.,  $u_1, u_3, u_7$  and  $w_1, w_2, w_3$  in  $T_1$ , and  $u_7, u_3, u_9$  and  $w_3, w_2, w_4$  in  $T_2$ ,  
 969 approximate the divergence  $D = (\partial u / \partial x + \partial w / \partial z)$  and the quantity  $\partial w / \partial x$  as constant  
 970 values in the two coarse triangles. The constant quantities  $\tilde{P}_1 D, \tilde{P}_2 D$  are then  
 971 accumulated over the entire grid. The constant quantity  $\partial w / \partial x$  in each coarse triangle is



972 then distributed to each of the eight fine grid elements  $t_1, t_2, \dots, t_8$  depending on whether  
 973 the centroid of the fine triangular element is in  $T_1$  or  $T_2$ . As in the previous case, this is  
 974 then used in the second invariant  $\tilde{\epsilon}^2$  when evaluating the transformed functional  $\tilde{\mathcal{A}}$ .  
 975

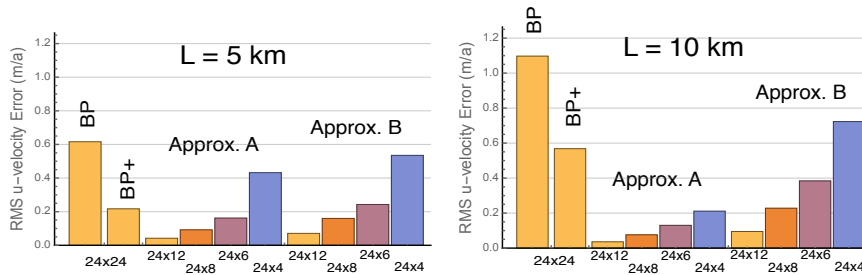
976 While the number and type of unknown variables is the same in the two versions,  
 977 they differ considerably in accuracy, as is seen in Figs. 11 and 12. Fig. 11 compares the  
 978 upper surface u-velocity in both version, Approximations A and B, for the four coarse  
 979 grid combinations and the reference 24x24 fine grid calculation. Fig. 12 compares the  
 980 overall accuracy the same way by means of the RMS u-Error. As might be expected, the  
 981 accuracy of Approx. A is better than the accuracy of Approx. B, particularly in the case  
 982 when  $L = 10$  km. Both versions are more accurate than the Blatter-Pattyn and BP+  
 983 approximations, except at the lowest 24x4 resolution when only the Approx. A version  
 984 retains that distinction.



985  
 986 **Figure 11.** Comparing Approximations A and B. Test B. Upper surface u-velocity.  
 987 TS-Ref: Reference Stokes 24x24; Fine/Coarse resolutions ( $r \times R$ ): 24xR, R=12, 8, 6, 4.  
 988

989 In summary, the dual-grid approximation improves on the Blatter-Pattyn  
 990 approximation in both versions and at all resolutions, as seen in Fig. 12. Compared to the  
 991 BP+ approximations, here the vertical momentum equation is retained, although in  
 992 approximated form. In fact, the solution procedure here is very similar to that of the  
 993 unapproximated Stokes model except that the dimensions of the pressure and the vertical

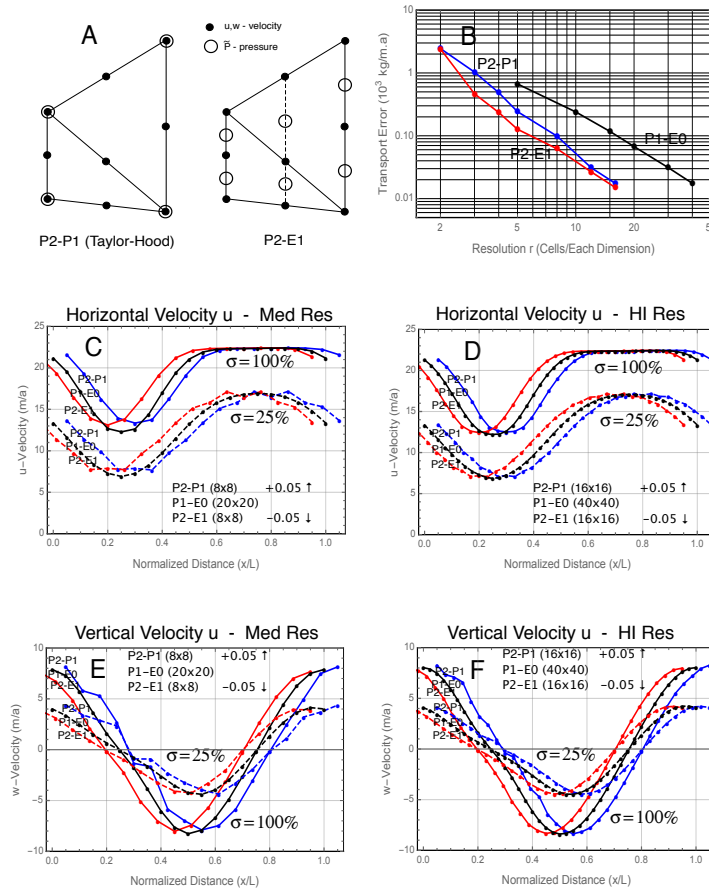
994 velocity variables are reduced. Despite the differences with the unapproximated case, the  
 995 arguments in Appendix D regarding stability extend to the case  $n_u > n_w = n_p$  appropriate  
 996 for the dual-grid approximation. As argued in Appendix D, provided the solvability  
 997 condition  $n_w = n_p$  holds on the coarse grid, the “reduced” continuity equation may be  
 998 solved for the coarse vertical velocity in terms of the fine horizontal velocity  
 999 variables,  $w = w(u)$ , and in turn, the coarse pressure may be obtained in terms of the fine  
 1000 horizontal velocity variables,  $p = p(u)$ , as in (79). As a result, pressure may be  
 1001 eliminated in the dual grid version of the functional, converting the variational  
 1002 formulation into a stable minimization problem. Thus, the solvability condition still  
 1003 applies, but this time it applies to the coarse grid.



1004  
 1005 **Figure 12.** Comparing RMS u-Error in Different Approximations, Test B,  
 1006 Resolutions (r x R): Approx. BP, BP+: 24x24; Approx. A, B: 24xR, R=12, 8, 6, 4.  
 1007

## 1008 7. Second-Order Discretizations

1009  
 1010 So far we have been using first-order elements, primarily P1-E0. However, in current  
 1011 practice Stokes models are often based on the popular second-order Taylor-Hood P2-P1  
 1012 element (Leng et al., 2012; Gagliardini et al., 2013). The two-dimensional P2-P1  
 1013 element, illustrated in Fig. 13A, has velocities on element vertices and edge midpoints  
 1014 and pressures on element vertices, resulting in a quadratic velocity and linear pressure  
 1015 within the element. The element satisfies the conventional inf-sup stability condition  
 1016 (Elman et al., 2014) but not the solvability condition (56). For example, in Test B with  
 1017 direct substitution for basal boundary conditions, the number of vertical velocity  
 1018 variables in the Taylor-Hood element,  $n_w = 4nm$ , is typically much larger than the  
 1019 number of pressure variables,  $n_p = n(m+1)$ , where  $n, m$  have been defined previously.



1020  
 1021 **Figure 13.** Comparing second-order discretizations based on the P2-P1 and P2-E1  
 1022 elements from panel A to first-order discretizations using the P1-E0 element running Test  
 1023 B with  $L=10$  km. For simplicity, only transformed Stokes calculations are compared;  
 1024 standard Stokes results behave similarly. Panel B compares the relative accuracy of the  
 1025 various schemes with increasing resolution, while panels C through F compare the  
 1026 horizontal and vertical velocities at medium and maximum resolutions, i.e.,  $r = 8, 16$  for  
 1027 second-order and  $r = 20, 40$  for first-order cases. Plots labeled  $\sigma = 100\%$  indicate the  
 1028 upper surface while dashed plots labeled  $\sigma = 25\%$  indicate surfaces a quarter of the way  
 1029 up from the bottom.

1030  
 1031 Stokes models work well with a Taylor-Hood grid, as illustrated in Fig. 13, where  
 1032 both P2-P1 and P1-E0 models converge to a common Test B solution, but models that  
 1033 require the solvability condition (56) will not work on a P2-P1 grid, as discussed in  
 1034 connection with the extended Blatter-Pattyn approximation in §4.3.3. For these



1035 applications an alternative will be needed if one wishes to use a second order  
1036 discretization. An alternative second-order element, consistent with an invertible  
1037 continuity equation, can be created by modifying the Taylor-Hood element to produce the  
1038 P2-E1 element illustrated in Fig. 13A. This element is second-order for velocities and  
1039 linear for pressure, just like the P2-P1 element, but the pressure is edge-based, as in the  
1040 P1-E0 element. The pressure is located midway between the velocities on the vertical  
1041 cell edges, including an “imaginary” vertical edge joining the velocities in the middle of  
1042 the vertical column as shown in Fig. 13A. Since pressures are collinear with vertical  
1043 velocities along vertical grid edges as in the P1-E0 element, the analysis in Appendix C,  
1044 §C2, demonstrates that element P2-E1 also satisfies the solvability condition (56).  
1045 Preferably, as explained in Appendix C, §C3, a P2-E1 grid is constructed using vertical  
1046 columns of quadrilaterals. A three-dimensional analog of this element exists and is  
1047 presented in Appendix C.

1048

1049 **Remark #2:** In addition to the P2-E1 element, it is possible to construct other elements  
1050 that feature an invertible continuity equation with second-order accurate velocities. Thus,  
1051 noting that there are  $2nm$  triangular elements in a Test B problem grid, it is sufficient  
1052 that each triangular element contains two pressures, resulting in the same total number of  
1053 vertical velocity and pressure variables, namely,  $n_w = n_p = 4nm$ . The pressure will not be  
1054 linear within the element but this is unimportant since, as noted before, pressure has no  
1055 physical significance.

1056

1057 Fig. 13B shows the approximate error of the ice transport  $T$  from (60) as a  
1058 function of grid refinement for the second-order P2-P1 and P2-E1 grids in transformed  
1059 Stokes Test B calculations, together with similar results for the first-order P1-E0 grid  
1060 from Fig. 3, for comparison. Calculation of the error  $E = \|T - T_R\|$ , as defined in §5.1, is  
1061 difficult because we do not have the converged value of the transport  $T_R$ . To estimate it,  
1062 we use Richardson extrapolation, assuming a rate of convergence proportional to  $r^{-c}$ ,  
1063 where  $r$  is the resolution and  $c$  is the order of convergence, taken to be either  $c = 2$  in a  
1064 first order model and  $c = 3$  in a second order model. This gives a reasonable estimate of  
1065 the magnitude of the error as plotted in Fig. 13B. We note that both second order models  
1066 show approximately the same error at resolution  $r = 16$  as the first order P1-E0 model at  
1067 resolution  $r = 40$ , and similarly for coarser resolutions such as  $r = 8$  and  $r = 20$ ,  
1068 respectively. However, although here the computational costs are not representative, it is



1069 safe to say that these second-order calculations are considerably more expensive than the  
1070 first-order calculations at comparable resolution or accuracy.

1071

1072 Panels C, D in Fig. 13 compare the  $u$ -velocities, and panels E, F compare the  $w$ -  
1073 velocities, respectively, from several Test B calculations using the two second-order  
1074 models in comparison with first-order P1-E0 model results from Fig. 3. Each panel  
1075 shows results from the upper surface ( $\sigma = 100\%$ ) in solid lines and results from a surface  
1076 a quarter of the way up from the bottom ( $\sigma = 25\%$ ) in dashed lines. Panels C, E show  
1077 results from medium resolution calculations ( $r = 8, 20$  in the second-order and first-order  
1078 calculations, respectively) and panels D, F show the corresponding results from the  
1079 higher resolution calculations ( $r = 16, 40$ ). At these resolutions the accuracy of the first-  
1080 and second-order calculations is very similar so for clarity the second-order results are  
1081 displaced horizontally from the first-order results by 0.05 nondimensional units. The P2-  
1082 E1 results in magenta are displaced to the left and the P2-P1 results in blue are displaced  
1083 to the right. In general, models satisfying the solvability condition, namely the P1-E0  
1084 and P2-E1 models, are better behaved than the Taylor-Hood model, particularly in the  
1085 vertical velocity results, panels E and F, where velocity oscillations are present in the P2-  
1086 P1 results. This is presumably related to the well-known “weak” mass conservation of  
1087 the Taylor-Hood element. This problem is greatly improved by “enriching” the pressure  
1088 space with constant pressures in each triangular element (Boffi et al., 2012). In the 2D  
1089 Test B problem this increases the number of pressure variables from  $n_p = n(m+1)$  in the  
1090 basic Taylor-Hood element to  $n(3m+1)$ , much closer to the  $4nm$  needed to satisfy the  
1091 solvability condition. On the other hand, it should be noted that the pressure in the P2-E1  
1092 case is highly oscillatory while in the P2-P1 case it is well behaved. However, this is not  
1093 at all concerning since, as mentioned earlier in Remark #2, the transformed pressure, a  
1094 Lagrange multiplier, has no physical significance.

1095

## 1096 **8. Summary**

1097

1098 This paper introduces two main innovations. Together, the two innovations expand the  
1099 scope of traditional methods used in ice sheet modeling. The first innovation is a  
1100 transformation of the ice sheet Stokes equations into a form that closely resembles the  
1101 Blatter-Pattyn approximate model. This creates the ability to easily convert from one  
1102 model to the other. The variational formulation of the Blatter-Pattyn approximation



1103 differs from the corresponding formulation of the transformed Stokes model only by the  
1104 absence of the vertical velocity  $w$  in the second invariant of the strain rate tensor. This  
1105 makes it possible to create new Stokes approximations by focusing on the smallness of  
1106 vertical velocity compared to other terms in the variational functional. Two such  
1107 approximations are presented, the BP+ approximation and the dual-grid approximation,  
1108 which are cheaper than full-Stokes and more accurate than Blatter-Pattyn. Both  
1109 approximations are based on using an approximate vertical velocity that is obtained  
1110 inexpensively for this purpose, in general by solving the continuity equation for the  
1111 vertical velocity in terms of the horizontal velocity components. In the variational  
1112 formulation, the continuity equation is obtained by variation with respect to the pressure,  
1113 yielding a system of  $n_p$  equations to solve for the  $n_w$  vertical velocity variables. Thus,  
1114 vertical velocity can only be obtained from the solution of the discrete continuity  
1115 equation if the number of unknown vertical velocity variables is equal to the number of  
1116 unknown pressure variables, i.e.,  $n_w = n_p$ . This is called the solvability condition.

1117

1118 The second innovation is the introduction of finite element grids in which the  
1119 solvability condition is satisfied. These grids incorporate a decoupled and invertible  
1120 discrete continuity equation. This has two important consequences. The first is that it  
1121 allows for the numerical solution of the continuity equation for the vertical velocity in  
1122 terms of the horizontal velocity components,  $w = w(u, v)$ , which is a prerequisite in the  
1123 different approximations made possible by the transformed Stokes formulation. A  
1124 second very important consequence is that invertibility of the continuity equation and the  
1125 availability of the vertical velocity in terms of the horizontal velocity components can be  
1126 used to remove the need for pressure as a Lagrange multiplier. Removing the pressure  
1127 from the system of Stokes equations, or from the variational functional, means that a  
1128 Stokes problem discretized with such a grid becomes a well-behaved minimization  
1129 problem rather than a mixed or saddle-point problem. This eliminates the need for the  
1130 inf-sup or LBB condition that is normally required to be satisfied in finite element  
1131 formulations. Some examples of such grids for use in both 2D and 3D are given in  
1132 Appendix C. An important case is the P1-E0 grid that has been used in most of the test  
1133 problems in this paper. To construct such grids we can focus on the term involving  
1134 pressure in the variational functionals (15) and (33) in isolation from the other terms, as is  
1135 done in (64). The pressure may then be considered a finite element “test function”,  
1136 allowing us to construct appropriate test functions that yield  $n_w$  independent equations



1137 corresponding to the linear system of continuity equations (57), which is sufficient to  
1138 solve for the vertical velocity in terms of the horizontal velocity components. This is  
1139 already done in MALI (Hoffman et al., 2018), an ice sheet model based on the Blatter-  
1140 Pattyn approximation, to obtain the vertical velocity  $w$  needed for the advection of ice  
1141 temperature (Mauro Perego, private communication).

1142  
1143 We have also introduced some minor innovations in the implementation of the  
1144 frictional tangential sliding boundary condition that is often challenging to implement  
1145 numerically. Implementation directly into the Stokes equations involves the formation of  
1146 the normal component of the stress force at the boundary. This is extremely complex  
1147 (e.g., see DPL, 2010). Appendix A describes an alternative that avoids this complication.  
1148 The variational formulation makes it possible to also implement this boundary condition  
1149 using Lagrange multipliers, but this may not be desirable because it introduces extra  
1150 variables. A much more attractive alternative is the use of the no-penetration condition in  
1151 the form given by (14) to eliminate the vertical velocity by direct substitution along the  
1152 frictional portion of the basal boundary, as discussed in connection with the functional  
1153 (15). This automatically enforces both the frictional sliding condition and the no-  
1154 penetration condition.

1155  
1156 Finally, we need to point out that no cost comparisons have been presented. This  
1157 is because the present calculations were made on a personal computer using the program  
1158 Mathematica, which is not at all representative of the computer hardware or the methods  
1159 that are used in practical ice sheet modeling. Furthermore, no effort was made to  
1160 optimize the calculations or to take advantage of parallelization. As a result, cost  
1161 comparisons would have been highly misleading.

1162  
1163 **Code Availability**

1164  
1165 All calculations were made using the Wolfram Research, Inc. program Mathematica in a  
1166 development environment. No production code is available.

1167  
1168 **Competing Interests**

1169  
1170 The author has acknowledged that there are no competing interests.

1171



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1173

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1176

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1284

## 1285 **Appendix A: The Frictional Sliding Boundary Condition**

1286

1287 The frictional sliding boundary condition requires the specification of the tangential  
1288 component of the frictional stress force. Dukowicz et al. (2010) obtain this by defining  
1289 the frictional stress force at the basal surface as follows

$$1290 \quad \sigma_{ij} n_j^{(b2)} = (\tau_{ij} - P \delta_{ij}) n_j^{(b2)} = -f_i$$

1291 where  $\sigma_{ij}$  is the stress tensor,  $\delta_{ij}$  is the Kronecker delta, and  $f_i$  is the frictional sliding  
1292 force vector from §2.2, and then subtracting out the normal component. The result is

$$1293 \quad (\tau_{ij} - \tau_n \delta_{ij}) n_j^{(b2)} + f_i = 0 \quad (71)$$

1294 where  $\tau_n = n_i^{(b2)} \tau_{ij} n_j^{(b2)}$  is the normal component of the stress force. However, the three  
1295 components of (71) are not independent because they already satisfy the tangency  
1296 condition at the basal surface. Since we already have one component of the basal  
1297 frictional boundary condition, namely, the tangency condition (10), we therefore need  
1298 only two more conditions and these are typically taken to be the two horizontal  
1299 components of (71). This option is problematic because of the need to form the highly  
1300 complex quantity  $\tau_n$ .

1301

1302 A simpler alternative is obtained by simply using the unneeded vertical  
 1303 component of (71) to eliminate  $\tau_n$  from the two horizontal components. The vertical  
 1304 component of (71) gives

$$1305 \quad \tau_n n_z^{(b2)} = \tau_{zj} n_j^{(b2)} + f_z. \quad (72)$$

1306 Substituting this into (71), we obtain the desired two conditions, as follows

$$1307 \quad n_z^{(b2)} \left( \tau_{(i)j} n_j^{(b2)} + f_{(i)} \right) - n_{(i)}^{(b2)} \left( \tau_{zj} n_j^{(b2)} + f_z \right) = 0. \quad (73)$$

1308 This is boundary condition (11) as used in §2.2.

1309

1310 Alternatively, one could use of a Lagrange multiplier  $\Lambda$  in the variational  
 1311 principle, as is done in (13) and in Dukowicz et al. (2011). This yields the tangency  
 1312 condition (10) together with

$$1313 \quad \tau_{ij} n_j^{(b2)} + (\Lambda - P) n_i^{(b2)} + f_i = 0. \quad (74)$$

1314 Equation (74) provides three conditions, which, together with (10), is one too many.

1315 However, one of these conditions must be used to determine the quantity  $\Lambda - P$ .

1316 Contracting (74) with  $n_i^{(b2)}$ , and using the fact that  $f_i$  is tangential to the basal surface,

1317 gives us  $\Lambda - P = -\tau_n$ , which, when substituted into (74) gives us agreement with (71).

1318 Alternatively, employing the vertical component of (74) to determine  $\Lambda - P$ , yields

$$1319 \quad \Lambda - P = - \left( f_z + \tau_{zj} n_j^{(b2)} \right) / n_z^{(b2)}. \text{ Substituting this into (74) gives the preferred boundary}$$

1320 condition (73).

1321

## 1322 **Appendix B: Test Problems**

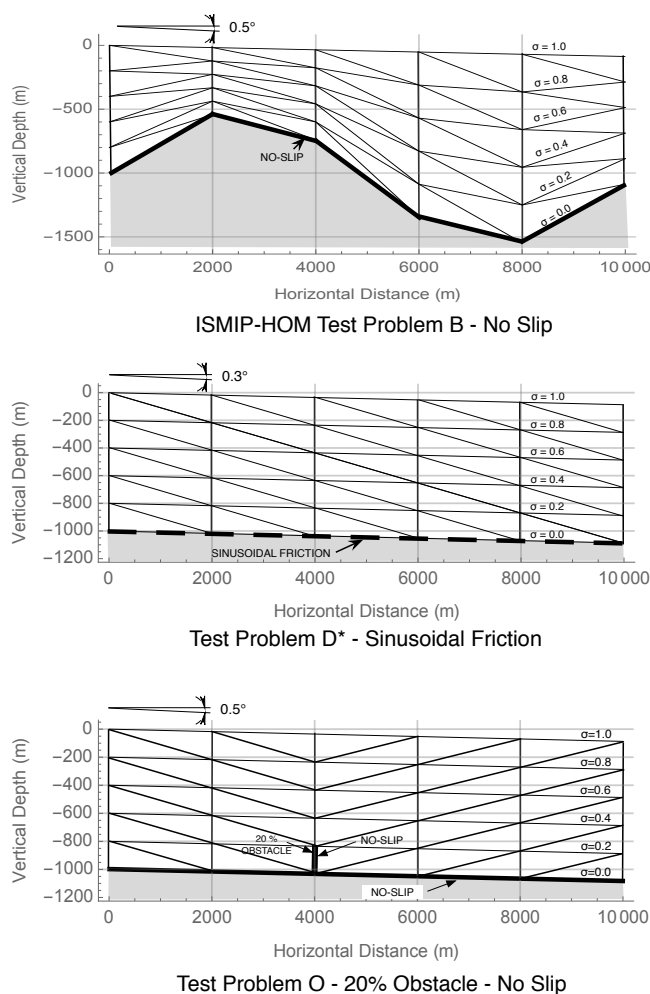
1323

1324 We will use three two-dimensional test problems to demonstrate the new methods. The  
 1325 geometrical configuration of the three test problem grids is illustrated in Fig. B1. The  
 1326 first problem, Test B, is actually Exp. B from the ISMIP-HOM benchmark suite (Pattyn  
 1327 et al., 2008); it features a no-slip condition (infinite friction) on a sinusoidal basal surface.

1328 The second problem, Test D\*, featuring sinusoidal friction along a uniformly sloped  
 1329 plane basal surface, is a replacement with modified parameters for Exp. D from the  
 1330 benchmark suite. This is because the ice flow in Exp. D is very nearly vertically uniform  
 1331 (as seen in Fig. 4), which is more characteristic of a shallow-shelf approximation.

1332 Increasing basal friction in Test D\* rectifies this. These two test problems, Tests B and

1333  $D^*$ , are used to illustrate and compare the performance of the new transformation versus  
1334 the traditional Stokes formulation.



1335

1336 **Figure B1.** Test problem grids. For clarity, a very coarse 5x5 configuration is used.

1337

1338 A third problem, Test O (for “Obstacle”) has been introduced to illustrate  
1339 adaptive switching between the transformed Stokes and the extended Blatter-Pattyn  
1340 model in a problem where the small aspect ratio assumption underlying the Blatter-Pattyn  
1341 approximation fails locally. Test O has a unique feature, namely, a thin no-slip obstacle,  
1342 located at  $x = 4$  km and extending vertically 200 m from the bed (20 % of the ice sheet  
1343 thickness), as illustrated in Fig. B1, which forces the ice flow near the obstacle to adjust  
1344 abruptly. Because of the no-slip boundary conditions along the obstacle surface, a

1345 triangular element in the lee of the obstacle, with one vertical edge and one edge along  
1346 the bed, would be a “null” element since all vertex velocities would be zero. This would  
1347 create zero stress and therefore a local singularity in ice viscosity in the element. To  
1348 avoid this, all elements at the back of the obstacle are “reversed” as compared to the ones  
1349 at the front of the obstacle, as shown in Fig. B1.

1350

1351 All tests feature a sloping flat upper surface, given by

1352 
$$z_s(x) = -x \tan(\theta), \quad (75)$$

1353 where  $\theta = 0.5^\circ$  for Tests B and O, and  $\theta = 0.3^\circ$  for Test D\* (note that this differs from the  
1354  $0.1^\circ$  slope in Test D), with a free-stress upper boundary condition in all cases. The  
1355 sinusoidal bottom surface elevation for Test B is specified by

1356 
$$z_b(x) = z_s(x) - H_0 + H_1 \sin(\omega x), \quad (76)$$

1357 where the depth  $H_0 = 1000 \text{ m}$ ,  $H_1 = 500 \text{ m}$ ,  $\omega = 2\pi/L$ , and  $L$  is the perturbation  
1358 wavelength, which is also the domain length. The bottom surface in Tests D\* and O is  
1359 parallel to the upper surface so the bottom surface elevation is

1360 
$$z_b(x) = z_s(x) - H_0. \quad (77)$$

1361 The length  $L$  in the ISMIP-HOM suite ranges from  $5 \text{ km}$  to  $160 \text{ km}$ , but here we  
1362 consider only the two cases at the high end of the aspect ratio  $H_0/L$  range, namely,  
1363  $L = 5 \text{ km}$  and  $L = 10 \text{ km}$ , where the inaccuracy of the Blatter-Pattyn approximation  
1364 becomes noticeable. Lateral boundary conditions in all cases are periodic. The spatially  
1365 varying friction coefficient for Test D\* is given by

1366 
$$\beta(x) = \beta_0 + \beta_1 \sin(\omega x), \quad (78)$$

1367 where the friction coefficients are  $\beta_0 = \beta_1 = 10^4 \text{ Pa a m}^{-1}$  (these are an order of  
1368 magnitude higher than in Test D). Physical parameters used for the test problems are the  
1369 same as in ISMIP-HOM, namely, ice-flow parameter  $A = 10^{-16} \text{ Pa}^{-3} \text{ a}^{-1}$ , ice density  
1370  $\rho = 910 \text{ kg m}^{-3}$ , and gravitational constant  $g = 9.81 \text{ ms}^{-2}$ . In general, units are MKS,  
1371 except where time is given per annum, which is convertible to per second by the factor  
1372  $3.1557 \times 10^7 \text{ s a}^{-1}$ .

1373

1374 **Appendix C: Grids Satisfying the Solvability Condition**

1375 **C1 A Solvable Continuity Equation**

1376

1377 As discussed in §4, the invertibility of the discrete continuity equation, at least in the  
1378 simplest case of direct substitution for basal boundary conditions, requires a special grid  
1379 that satisfies the solvability condition (56), i.e.,  $n_p = n_w$ . Here we discuss several such  
1380 grids and their properties.

1381

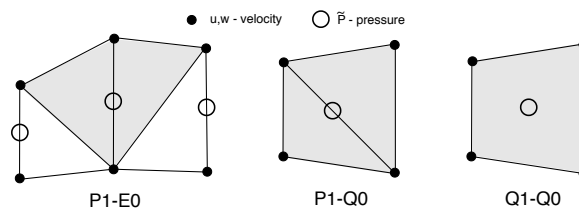
1382 The finite element discretization of our test problems, described in Appendix B  
1383 and illustrated in Fig. B1, is constructed using vertical columns of quadrilaterals that are  
1384 subdivided into triangles. Fig. C1 illustrates three different two-dimensional elements on  
1385 triangles or quadrilaterals that may be used to construct grids that may be used to satisfy  
1386 the solvability condition (56) in certain circumstances. The P1-E0 element is quite  
1387 general and satisfies the solvability condition along each vertical grid edge, as will be  
1388 demonstrated in Appendix C, §C2. As noted before, it has velocities located at triangle  
1389 vertices, resulting in a linear velocity distribution within the triangle (P1), and pressure is  
1390 located on the vertical edge of each triangle, resulting in constant pressure over the two  
1391 triangles that share that edge (E0). A second order version of the P1-E0 element, the P2-  
1392 E1 element, is illustrated in Fig. 13A. The two other elements in Fig. C1, i.e., the P1-Q0  
1393 and Q1-Q0 elements, satisfy the solvability condition when used in the grids for our test  
1394 problems, Tests B and D\*, but may not do so in other problems. The P1-Q0 element also  
1395 has velocities on triangle vertices for a linear velocity distribution within the triangle  
1396 (P1), but pressure is constant within the two triangles that form a quadrilateral (Q0). The  
1397 element Q1-Q0 has velocities located at quadrilateral vertices and pressure centered in  
1398 the quadrilateral, resulting in a bi-quadratic velocity distribution and a constant pressure  
1399 within the quadrilateral (Q0).

1400

1401

1402

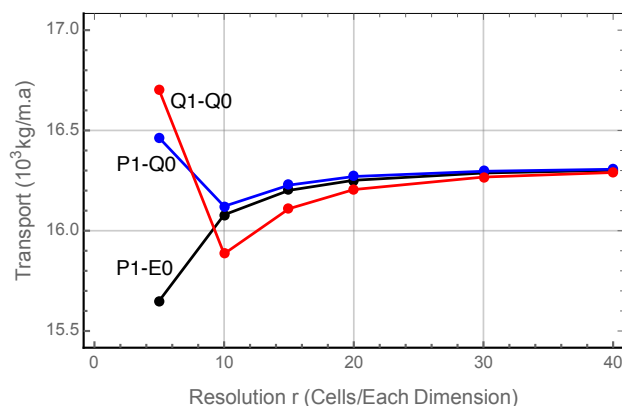
1403



**Figure C1.** Three first-order 2D elements that may be used to satisfy the solvability condition, (56), in Tests B and D\*.



1404 Fig. C2 shows the convergence of ice transport with grid resolution for Test B  
 1405 calculations using these three elements. The solutions are stable and they all converge to  
 1406 the same value for the ice transport. The pressure distribution is smooth in the P1-E0  
 1407 case, but contains very small fluctuations near the surface in the P1-Q0 and Q1-Q0 cases  
 1408 that tend to disappear as the resolution is increased. The Q1-Q0 element is attractive  
 1409 because of its simplicity but it has the potential for a pressure null space, resulting in  
 1410 pressure checkerboarding (Elman et al., 2014, where the element is called Q1-P0). As a  
 1411 result, apparently it is only used in a stabilized form. Here, however, the Q1-Q0 grid  
 1412 satisfies the solvability condition in Test B and behaves well. Overall, these results  
 1413 confirm our expectation of stability for grids when they satisfy the solvability condition  
 1414 as will be discussed in Appendix D. The P1-E0 element is somewhat special because the  
 1415 solvability condition (56) is satisfied individually along each vertical edge in grids that  
 1416 are composed of this element, as opposed to being satisfied over the entire grid as in the  
 1417 other two elements, as we discuss next.



1418  
 1419 **Figure C2.** Convergence of Test B ice transport for grids using the three elements  
 1420 from Fig. C1. All discretizations are stable and converge to the same solution.  
 1421

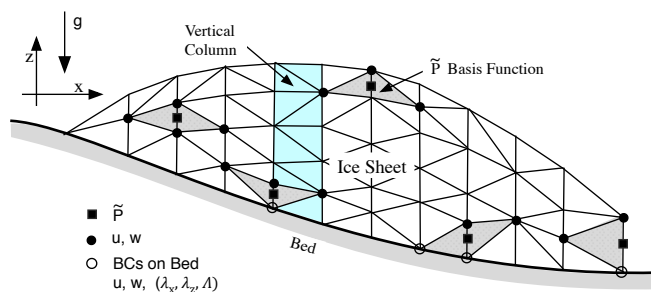
1422 **C2 Proving that the P1-E0 Element Satisfies the Solvability Condition**

1423 The P1-E0 element from Fig. C1 is used in an example grid in Fig. C3. Note that  
 1424 the grid is composed of vertical columns subdivided into triangular elements. To  
 1425 demonstrate that the element meets the solvability condition (56) it is sufficient to  
 1426 consider a single vertical edge extending from the bottom to the top. Assuming there are  
 1427  $m$  edge segments in the vertical direction, there will be  $m + 1$  discrete  $w$  variables and  $m$   
 1428 discrete  $\tilde{P}$  variables, such that each  $\tilde{P}$  variable is located between a pair of  $w$  variables.  
 1429 Since the  $w$  variable at the bed is specified as a boundary condition, either directly as a  
 1430 no-slip condition or in terms of the horizontal velocity component as part of a no-





1431 penetration condition, there will be only  $m$  unknown  $w$  variables, and therefore  $n_w = n_p$   
 1432 along each vertical grid edge, and hence over the entire grid, as desired. In case  
 1433 Lagrange multipliers are used, there will be  $m+1$  unknown discrete  $w$  variables (since  
 1434 now the basal vertical velocity  $w$  is also an unknown). This is matched by  $m$  unknown  
 1435  $\tilde{P}$  variables, supplemented by one  $\lambda_z$  or one  $\Lambda$  unknown Lagrange multiplier variable,  
 1436 depending on the type of boundary condition. Thus, again the number of unknown  
 1437 variables equals the number of equations along every vertical edge, thereby satisfying the  
 1438 solvability condition whether Lagrange multipliers are used or not. Importantly, this  
 1439 means that this element can be used to satisfy the solvability condition irrespective of the  
 1440 boundary conditions on quite arbitrary grids, as illustrated in Fig. C3. These arguments  
 1441 apply for other versions of the P1-E0 element as well, such as the second order version  
 1442 P2-E1 in Fig. 13A or the 3D version in Fig. C4.



1443

1444

1445 **Figure C3.** An illustration of a 2D edge-based P1-E0 grid, composed of vertical columns  
 1446 randomly subdivided into triangles. Pressures are located on the vertical edges.  
 1447 The triangulation and the configuration of the associated pressure basis functions  
 1448 (shown in gray) is quite general, allowing for a flexible triangulation of the domain.

1449

### 1450 C3 Two- and Three-Dimensional Meshes Based on the P1-E0 Element

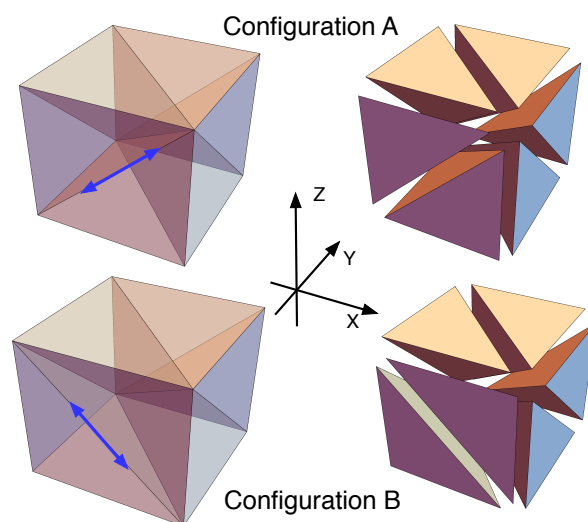
1451

1452 The P1-E0 element has been used on the simple test problem grids in Fig. B1 and  
 1453 performs well. Moreover, the element has great geometric generality so it may be used  
 1454 for quite complicated grids, as in Fig. C3. Generally, there are two triangles associated  
 1455 with a pressure variable, one on each side of a vertical edge, except in situations as in Fig.  
 1456 C3 where the ice sheet ends at a vertical face. Even in this unusual situation there is no  
 1457 problem since the pressure is simply associated with the single triangle on one side of the  
 1458 vertical face.

1458



1459           The two-dimensional P1-E0 element has a relatively simple three-dimensional  
 1460 counterpart, shown in Fig. C4. The mesh again consists of vertical columns, this time  
 1461 composed of hexahedra. Each hexahedron is subdivided into six tetrahedra such that  
 1462 each vertical edge is surrounded by from as few as four to as many as eight tetrahedra.  
 1463 As in the two-dimensional case, velocity components are collocated at vertices, yielding a  
 1464 piecewise-linear velocity distribution in each tetrahedral element, and pressures are  
 1465 located in the middle of each vertical edge so that pressure is constant in the tetrahedra  
 1466 surrounding that edge. Lagrange multipliers, if used, are located at the vertices on the  
 1467 basal surface, yielding a piecewise linear distribution on the basal triangular facet. This  
 1468 arrangement also satisfies the solvability condition (56) since pressures and vertical  
 1469 velocities are again intermingled along a single line of vertical edges from top to bottom,  
 1470 as in the 2D case. Thus, the solvability argument used in the two-dimensional case  
 1471 applies, confirming that the 3D version of the P1-P0 element also satisfies the solvability  
 1472 condition.



1473  
 1474  
 1475  
 1476  
 1477  
 1478

**Figure C4.** Three-dimensional P1-E0 tetrahedral elements that generalize the 2D P1-E0 element of Fig. C1. Configurations A and B differ by having an internal triangular face rotated, as indicated by the blue arrows. Both configurations satisfy the solvability condition.

1479           Fig. C4 shows two of the several possible configurations of a typical hexahedron,  
 1480 including an exploded view of each configuration for clarity. The two configurations  
 1481 differ in having the internal face of the two forward-facing tetrahedra rotated, creating  
 1482 two different forward facing tetrahedra. The remaining six tetrahedra are undisturbed.



1483 Since edges must align when hexahedra (or tetrahedra) are connected, this demonstrates  
1484 that the three-dimensional mesh can be flexibly reconnected and rearranged, just as in the  
1485 two-dimensional case.

1486

1487 **Remark #3:** A closely related and perhaps simpler three-dimensional P1-E0 element is  
1488 one based on the P2-P1 prismatic tetrahedral element used in Leng et al. (2012). A grid  
1489 of these elements is composed of vertical columns of triangular prisms, with triangular  
1490 faces at the top and bottom, which are then each subdivided into three tetrahedra. As in  
1491 Fig. C4, pressures are located on the vertical prism edges.

1492

1493 Meshes composed of P1-E0 elements have another useful property. Since  
1494 pressure and vertical velocity variables alternate along vertical grid lines, the matrix-  
1495 vector products  $M_{wp}P$ ,  $M_{wp}^T w$  in (47), corresponding to  $\partial\tilde{P}/\partial z$  and  $\partial w/\partial z$  in the  
1496 vertical momentum and continuity equations, respectively, consist of simple decoupled  
1497 bi-diagonal one-dimensional difference equations along each vertical grid line for  
1498 determining pressure, as in (79), and the vertical velocity, as in (58). This should be  
1499 particularly advantageous for parallelization.

1500

1501 Just as the two-dimensional second-order P2-E1 element in Fig. 13A is a  
1502 generalization of the P1-E0 element, a three-dimensional second-order P2-E1 element  
1503 may be constructed as a generalization of the P1-E0 element illustrated in Fig. C4.  
1504 Velocities are to be located at the vertices and at midpoints of the tetrahedral edges, and  
1505 pressures are to be located halfway between the velocities on vertical edges, including the  
1506 imaginary vertical edges through the midpoints of the tetrahedral edges, in the same way  
1507 as in the 2D case in Fig. 13A. The P2-E1 element, both 2D and 3D, also satisfies the  
1508 solvability condition since the arguments in Appendix C, §C2, apply here also because  
1509 pressures are again located midway between vertical velocities along all vertical edges.

1510

## 1511 **Appendix D: Proving the Stability of a Stokes Problem with an** 1512 **Invertible Continuity Equation**

1513 Here we show that a discretization of a Stokes problem is stable on a grid that  
1514 satisfies the solvability condition (56), or equivalently, one that is consistent with an  
1515 invertible continuity equation, i.e., (58). This is because such a discretization is  
1516 equivalent to the formulation of an unconstrained problem, i.e., a problem without the use  
1517 of pressure as a Lagrange multiplier. In fact, such a problem is also equivalent to an

1518 optimization problem, or more specifically, to a minimization problem. To demonstrate  
1519 this, consider the full set of discrete Euler-Lagrange equations (47). Recall that the  
1520 solvability condition implies the invertibility of  $M_{wp}$ , and therefore also the invertibility  
1521 of its transpose,  $M_{wp}^T$ , i.e., (59). This means that we can solve for the pressure from the  
1522 vertical momentum equation, the second equation in (47), to obtain

$$1523 \quad p = -M_{wp}^{-1} \left( M_w(u, w(u)) + F_w \right), \quad (79)$$

1524 where we would use  $w(u)$  from (58). Using (79) to eliminate the pressure in the  
1525 horizontal momentum equation, we obtain

$$1526 \quad M_U(u, w(u)) - M_{UP} M_{wp}^{-1} \left( M_w(u, w(u)) + F_w \right) + F_U = 0. \quad (80)$$

1527 This is a nonlinear set of equations for just the horizontal velocity  $u$ , similar in this  
1528 respect to the standard Blatter-Pattyn formulation in that it is no longer a mixed or  
1529 saddle-point problem because pressure is absent. As a result, although still a rather  
1530 complicated nonlinear problem, it should not suffer from the stability issues discussed in  
1531 §4.3.1. Alternatively, using  $w = w(u)$  in the functional (46) eliminates the pressure term  
1532 because continuity is already satisfied, and one obtains a reduced functional,

$$1533 \quad \mathcal{A}(u) = \mathcal{M}(u, w(u)) + u^T F_U + w(u)^T F_w. \quad (81)$$

1534 This implies that  $\mathcal{A}(u)$  is a positive-definite functional involving only the horizontal  
1535 velocity components because  $\mathcal{M}(u, w(u))$  is positive-definite (see §4.1), which means  
1536 that now the Stokes variational formulation represents an optimization, or more  
1537 specifically, a minimization problem. It is therefore a well-defined and stable problem  
1538 for the horizontal velocities (albeit numerically very expensive). We conclude that the  
1539 solution of a Stokes model on a grid satisfying the solvability condition, or equivalently,  
1540 one that allows for an invertible discrete continuity equation is stable and well behaved.  
1541

1542 Note that the arguments here and in §4 apply to arbitrary values of  $n_u, n_w, n_p$ , and  
1543 in particular, they apply in the case  $n_u > n_w = n_p$  that is relevant to the “dual-grid”  
1544 approximation of §6.2.2. As a result, we conclude that the dual-grid approximation is  
1545 also stable provided the solvability condition (56) holds on the coarse grid.  
1546



1547 **Remark #4:** Instead of the standard formulations of the Stokes problem that include the  
1548 pressure, such as (46) or (47), one could consider using the corresponding pressure-free  
1549 formulation, (80) or (81), to solve for  $u$ , followed by (58) and (79) if one is interested in  
1550 the vertical velocity and pressure. This corresponds to a discrete version of the pressure-  
1551 free formulation attempted analytically by Dukowicz (2012). However, this formulation  
1552 couples together large parts of the grid and produces a dense Hessian matrix when using  
1553 Newton-Raphson iteration, thus making the conventional numerical solution extremely  
1554 costly and therefore impractical, particularly for large problems.