

Quadratic Magnetic Gradients from 7-~~SC~~ and 9-~~SS~~spacecraft Constellations

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10 **Abstract.** To reveal the dynamics of magnetised plasma, it is essential to know the geometrical structure of the magnetic field, which is closely related to its linear and quadratic [spatial](#) gradients. Estimation of the linear magnetic gradient requires at least four magnetic measurements [simultaneously](#), whereas calculation of the quadratic gradients of the magnetic field generally requires at least ten [simultaneous measurements](#). This study is ~~therefore~~ aimed at yielding linear and quadratic [spatial](#) gradients of the magnetic field based on magnetic measurements from [the](#) nine-spacecraft HelioSwarm or seven-spacecraft Plasma
15 Observatory constellations. Time-series magnetic measurements and transfer relationships between different reference frames were used to yield the apparent velocity of the magnetic structure as well as the components of the quadratic magnetic gradient along the direction of motion, while ~~simultaneously~~ elucidating the linear gradient and remaining components of the quadratic magnetic gradient using the least-squares method. Calculation via several iterations was applied to achieve satisfactory accuracy. ~~The tests~~ [Tests](#) for the situations of magnetic flux ropes and dipole magnetic field have ~~verified~~ the validity and
20 accuracy of this approach. The results suggest that using ~~time-series~~ [simultaneous temporal](#) magnetic measurements from constellations comprising at least seven spacecraft and nonplanar configurations can yield linear and quadratic [spatial](#) gradients of the magnetic field.

Key Points:

1. ~~An iterative~~ [iteration](#) algorithm for the quadratic magnetic gradient based on measurement with constellations
25 comprising at least seven spacecraft is presented.
2. [Magnetic flux ropes and dipole magnetic field testing](#) verifies the validity and accuracy of the approach.
3. [Constellations containing at least seven spacecraft with nonplanar configurations](#) are required for the approach.

30 **Key Words:** Multiple Spacecraft Measurements, Space Plasmas, Magnetic field, Quadratic Magnetic Gradient, Least Squares Method

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1 Introduction

Multi-spacecraft constellations offer unique opportunities to observe plasma processes at various spatiotemporal scales simultaneously. Magnetic measurements *in situ* with multi-spacecraft constellations, in particular, allow the deduction of magnetic gradients such that fine magnetic structures, current densities, and magnetic geometries can be investigated. In general, magnetic measurements from constellations comprising at least four spacecraft forming a nonplanar configuration are required to deduce the three-dimensional (3-D) linear spatial gradient of a magnetic field (Harvey, 1998; Chanteur, 1998; Chanteur and Harvey, 1998; Shen et al., 2003; De Keyser, et al., 2007; De Keyser, 2008; Hamrin et al., 2008; Shen et al., 2012). In addition, linear spatial gradients of any scalar (e.g., plasma moments) or vector (e.g., an incompressible velocity field) fields can be obtained in a similar manner. Such a linear gradient can be obtained by considering a Taylor's expansion around the origin (e.g., the four-spacecraft mesocentre) up to the first order; the linear gradient that yields a (unique) solution, which matches the magnetic measurements, can then be obtained using the least-squares method (Harvey, 1998; Chanteur, 1998; Chanteur and Harvey, 1998; Shen et al., 2003; Broeren and Klein et al., 2013).

Cluster (Escoubet et al., 1997, 2001) and Magnetospheric MultiScale (MMS; Burch et al., 2015) are four-spacecraft constellations that form a tetrahedral configuration. Using the simultaneous magnetic measurements of such missions allows the linear spatial gradient of the magnetic field, e.g., the current density distribution, to be estimated and the topology of the magnetic field to be further derived (Dunlop et al., 2002b; Shen et al., 2003, 2008, 2012, 2014; Shi et al., 2005; Runov et al., 2006; Shi et al., 2010; Zhang et al., 2011; Rong et al., 2011; Burch and Phan, 2016; Dong et al., 2018; Pitout and Bogdanova, 2021; Haaland et al., 2021). Furthermore, four-point magnetic field measurements can also be applied to determine the orientation and motion of planar discontinuities (Russell et al., 1983; Dunlop et al., 2002a; Sonnerup et al., 2004), as well as the geometry of curved boundary layers (Shen et al., 2020; Kieokaew et al., 2018; Kieokaew and Foullon, 2019; Shen et al., 2020). For a planar constellation or a constellation comprising three spacecraft, only a two-dimensional linear magnetic gradient in the constellation plane can generally be derived (Vogt et al., 2009, 2013; Shen et al., 2012). Nevertheless, for certain structures such as one-dimensional and force-free structures, magnetic measurements from planar constellations or even Double Star constellations can also be reduced to a three-dimensional linear magnetic gradient (Vogt et al., 2009, 2013; Shen et al., 2012).

To estimate second spatial derivatives of obtain high-order gradients in the magnetic field (or Hessian matrix over each component of the magnetic field), simultaneous magnetic measurements from a constellation with more spacecraft are required. Considering a Taylor's expansion of the magnetic field around the origin up to the second order, there are 10 unknown

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65 parameters: 1 magnetic measurement at the origin, 3 components of linear magnetic gradient, and 6 components of second-
order magnetic gradient (i.e., the quadratic gradient tensor is symmetric). This problem can be considered as a system of
equations. To obtain a unique solution to the system of equations, we need a number of unknown parameters to be equal to, or
less than, the number of the constraints (i.e., equations). Therefore, 10-point measurements are required to solve the quadratic
gradient. For a scalar field, the spatial quadratic gradient in 3-D consists of 9 components, e.g., a tensor of 2nd order gradient
70 with 3 by 3 dimension. For the spatial quadratic gradient of the magnetic field in 3-D, e.g., $\mathbf{B} = (B_x, B_y, B_z)$, where \mathbf{B} is the
magnetic field, we have 18 components in total. To completely solve such a quadratic gradient, we thus need at least 18
constraints.

which contains 18 components in total, a constellation with at least ten spacecraft is required to solve the system of
equations (Chanteur, 1998; Shen et al., 2021b), with the limitation given that not all spacecraft are simultaneously within the
75 same quadratic surface (Zhou and Shen, 2024). Nevertheless, the quadratic gradient of a magnetic field can still be estimated
from four-spacecraft constellations if additional current density measurements deduced from electron and ion measurements
and certain physical constraints such as Ampère's law and Magnetic Gauss's law, are utilised (Liu et al., 2019; Torbert et al.,
2020; Shen et al., 2021a). Utilising the linear and quadratic gradients of the magnetic field means that the complete geometry
of a magnetic field, which concerns linear, e.g., current sheets, and nonlinear spatial structures, e.g., magnetic flux ropes, can
80 be determined (Shen et al., 2021a). Furthermore, the calculation of quadratic spatial gradients of physical electromagnetic and
plasma quantities in general allows us to study nonlinear plasma dynamics involving second-order spatial derivatives such as
in plasma turbulence (e.g., Politano and Pouquet, 1998a, b; Yang, 2019; Pecora et al. 2023) and nonlinear wave dynamics (e.g.
Chian et al. 1998; 2022), among others.

The HelioSwarm mission (Klein et al. 2023) is a nine-spacecraft constellation consisting of one hub (mothercraft)
85 and eight nodes (daughtercraft) planned to be launched in 2029 by the National Aeronautics and Space Administration (NASA).
The swarm of nine spacecraft will allow for the first time simultaneous cross-scale observations of turbulent solar-wind
plasmas for the first time in the vicinity of Earth. Specifically, each spacecraft of HelioSwarm will be equipped with a Fluxgate
magnetometer and a Search-Coil magnetometer, allowing comprehensive measurements of magnetic fields at 9 points
simultaneously. Plasma Observatory (Retinò et al. 2022) is a new European Space Agency (ESA)'s new mission with a seven-
90 spacecraft constellation in the Solar-Terrestrial environments, currently under Phase-A study. One important topic for these
two new multi-spacecraft constellations is to ascertain how the linear and quadratic gradients of the magnetic field can be
inferred from seven- or nine-point magnetic measurements, allowing the fine, nonlinear spatial structures of the magnetic
field in a space plasma to be identified. In this study, a new algorithm for calculating the linear and quadratic spatial gradients
of the magnetic field from seven- or nine-point simultaneous magnetic measurements was derived using the least-squares
95 method. By considering the transformation of reference frame involving mixed space-time derivatives of the magnetic field,
we demonstrate that 7- or 9-point simultaneous measurements can be used to estimate quadratic spatial gradients. Here, by
exploiting the least-squares method, we propose an iterative approach to achieve an optimal solution.

as well as an iterative approach that considers the reference frame transformation of the magnetic field.

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The remainder of this paper is as follows. The new algorithm for calculating the linear and quadratic magnetic gradients from 7- or 9-point simultaneous magnetic measurements is presented in Section 2; a description of the tests conducted for two typical nonlinear magnetic structures: (a cylindrical force-free flux rope and a dipole magnetic field), which were utilized to check the validity and accuracy of the new algorithm, is given in Section 3; the accuracy error of the algorithm is evaluated in section 4; and finally, the conclusions are presented in Section 5.

2. Methodology

2.1 The scheme

Calculation of the linear and quadratic gradients of a magnetic field generally requires simultaneous magnetic measurements from at least ten spacecraft. There are $3+9+18=30$ parameters in the Taylor expansion up to second order, and $3N$ magnetic field measurements in an array with N spacecraft are needed accordingly. Thus, using the magnetic measurements of nine-spacecraft (9S/C) HelioSwarm or seven-spacecraft (7S/C) Plasma Observatory constellation means that additional constraints are required. The transfer relationships between different reference frames are the proper limitations used for completely determining the spatial linear and quadratic gradients of the magnetic field. In these limits, we assume that the magnetic structures are slowly evolving during their passages through the multi-point constellations such that any differences in the measurements at different spacecraft can be attributed to the spatial variations rather than the temporal changes (i.e., evolution of magnetic structures).

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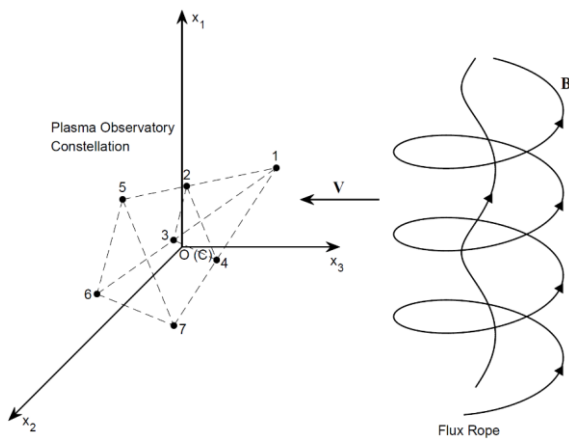


Figure 1. Schematic plot showing observation of a magnetic structure by the Plasma Observatory Constellation, which is composed of seven spacecraft. (The special configuration is similar to the mission term proposal and the actual geometry can deviate from it.) Barycentric coordinates are adopted; thus, the centre C of the constellation overlaps with the origin O of the Cartesian coordinates (x_1, x_2, x_3) the magnetic structure is assumed to be moving at velocity \mathbf{V} relative to the constellations, and the x_3 axis is presumed to be anti-parallel to \mathbf{V} .

The Taylor expansion of the magnetic field within two orders is expressed using:

$$\mathbf{B}(\mathbf{t}, \mathbf{r}) = \mathbf{B}(\mathbf{t}, \mathbf{r}_c) + (\mathbf{r} - \mathbf{r}_c) \cdot \nabla \mathbf{B}(\mathbf{t}, \mathbf{r}_c) + \frac{1}{2} (\mathbf{r} - \mathbf{r}_c) (\mathbf{r} - \mathbf{r}_c) \cdot \nabla \nabla \mathbf{B}(\mathbf{t}, \mathbf{r}_c). \quad (1)$$

While the Taylor expansion of each component of the magnetic field at each spacecraft α can be written as:

$$f_{(\alpha)} = f_c + x_{(\alpha)}^i (\nabla_i f)_c + \frac{1}{2} x_{(\alpha)}^i x_{(\alpha)}^j (\nabla_i \nabla_j f)_c = f_c + x_{(\alpha)}^i g_i + \frac{1}{2} x_{(\alpha)}^i x_{(\alpha)}^j G_{ij}. \quad (2)$$

Where f represents any one of the three components B^1, B^2, B^3 of the magnetic field \mathbf{B} . The first-order gradient is denoted as $g_i \equiv (\nabla_i f)_c$ where $(i=1, 2, 3, \text{i.e. the three cartesian components of } \mathbf{B})$ and the second-order gradient is denoted as $G_{ij} \equiv (\nabla_i \nabla_j f)_c$ where $(i, j=1, 2, 3)$.

Conventionally, 10 ten-point simultaneous measurements are necessary to infer draw both the first- and second-order spatial gradients of a physical scalar quantity-field (Chanteur, 1998; Shen et al., 2021b); thus, additional physical constraints are required. To obtain such the measurements spatial gradients with the 9S/C HelioSwarm and 7S/C Plasma Observatory, we consider adding additional physical constraints to the system of equations. Figure 1 shows a schematic diagram about observation of a magnetic structure by the Plasma Observatory Constellation. The shape of the constellation is ideal but this does not change the generality and applicability of our method.

The following transformation relationship involving the mixed space-time derivatives is used for the magnetic measurements:

$$\partial_t \mathbf{B} = -\mathbf{V} \cdot \nabla \mathbf{B} \text{ and } \partial_t \nabla \mathbf{B} = -\mathbf{V} \cdot \nabla \nabla \mathbf{B}. \quad (3)$$

By computing the temporal derivative $\partial_t \mathbf{B}$ and then the temporal derivative of spatial gradient $\partial_t \nabla \mathbf{B}$ (This relationship allows both the apparent velocity \mathbf{V} of the magnetic structure and the nine components of the quadratic magnetic gradient tensor along the direction of motion, $\hat{\mathbf{V}} \cdot \nabla \nabla \mathbf{B}$, to be obtained (Shen et al., 2021a). The constraints to the Eq. (3) are that the plasmas are highly conductive and have a very low velocity ($V/c \ll 1$, where V is the apparent speed of the magnetic structure and c is the speed of light in vacuum), and the physical processes are slowly evolving at low frequencies. The truncation errors in formula (3) are on the order V/c , where V is the apparent speed of the magnetic structure and c is the speed of light in vacuum.

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2.1.1 The zeroth iteration

150 First, the temporal variation rate $\partial_t \mathbf{B}$ and first-order magnetic gradient $(\nabla \mathbf{B})^{(0)}$, where the uppercase label (0) denotes the zeroth iteration order, can readily be obtained from seven- or nine-point simultaneous magnetic measurements. Here, the temporal change rate of the magnetic field can be obtained from the temporal (time-series) measurements at each spacecraft. The linear spatial gradient can be obtained using four-spacecraft techniques (Chanteur, 1998; Harvey, 1998; Shen et al. 2003), and Using equation (3), we can thus obtain the apparent velocity \mathbf{V} of the magnetic structure (Shen et al., 2021a). and Next, the longitudinal components of the second-order magnetic gradient, $\hat{\mathbf{V}} \cdot \nabla \nabla \mathbf{B} = \frac{1}{V} \partial_3 \nabla \mathbf{B}$, can be deduced from the transformation relationship (3). These steps will be described in detail in Section 2.2.1. Finally, the remaining nine components of the second-order magnetic gradient (i.e., the transverse components $G_{rs}^{(1)} = (\nabla_r \nabla_s f)^{(1)}$ where $(r, s = 1, 2)$) can be determined from the seven- or nine-point simultaneous magnetic measurements using the least-squares method, allowing a first-order quadratic magnetic gradient $(\nabla \nabla \mathbf{B})^{(1)}$ to be obtained, as described next.

2.1.2 The first order iteration

160 Provided with the zeroth+first-order quadratic magnetic gradient $(\nabla \nabla \mathbf{B})^{(1)}$, the corrected first-order linear magnetic gradient $(\nabla \mathbf{B})^{(1)}$ can be found using the least-squares method. Furthermore, the corrected apparent velocity $\mathbf{V}^{(1)}$ of the magnetic structure and the longitudinal components of the second-order quadratic magnetic gradient $(\hat{\mathbf{V}} \cdot \nabla \nabla \mathbf{B})^{(2)}$ can be obtained from the transformation relationship (3). Again, the corrected transverse components of the quadratic magnetic gradient $(G_{rs}^{(2)}(r, s = 1, 2))$ are obtained using the least-squares method, allowing a firstsecond-order quadratic magnetic gradient $(\nabla \nabla \mathbf{B})^{(2)}$ to be obtained.

170 The iterations are performed repeatedly until results are converge, which means satisfactory results are achieved. For the 7S/C Plasma Observatory, the seven-point magnetic measurements in 3-D yield $7 \times 3 = 21$ independent parameters, while the reference frame transformation provides nine constraints, resulting in $21 + 9 = 30$ input parameters in total. The objective is to determine the magnetic field (three parameters), first-order gradient (nine parameters), and quadratic magnetic gradient (18 parameters) at the mesocentre of the constellation, a total of $3 + 9 + 18 = 30$ parameters. Therefore, this scheme is reasonable such that the solution to the system of equation can be uniquely determined.

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Clearly, the 9S/C magnetic measurements of HelioSwarm are sufficient to draw first-order and quadratic magnetic gradients using this method. These results indicate that the developed method is suitable for constellations comprising at least seven spacecraft.

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2.2 Specific (operational/practical) steps of the algorithm

Details of the steps used are given below.

2.2.1 The zeroth iteration:

We first assume a linear approximation in space and let $G_{ij}^{(0)} = 0$. The magnetic field $\mathbf{B}_c^{(0)}$ and its linear gradient $(\nabla \mathbf{B})^{(0)}$ at the mesocentre of the constellation can then be obtained using the following formulas (Harvey, 1998; Shen et al., 2003):

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$$B_{ci}^{(0)} = \frac{1}{N} \sum_{\alpha=1}^N B_{\alpha i}, \quad (4)$$

$$(\partial_i B_j)_c^{(0)} = \frac{1}{N} \sum_{\alpha=1}^N B_{\alpha i} r_{\alpha k} R_{kj}^{-1}. \quad (5)$$

where the volume tensor is $R_{kj} = \frac{1}{N} \sum_{\alpha=1}^N r_{\alpha k} r_{\alpha j}$ or $\mathbf{R} \equiv \frac{1}{N} \sum_{\alpha=1}^N \mathbf{r}_{\alpha} \mathbf{r}_{\alpha}^T$, where N is the number of spacecraft within the constellation, and R_{kj}^{-1} is the inverse of the volume tensor R_{kj} . The determinant of the volume tensor is required to be nonzero, i.e., $\mathbf{R} = \det(\mathbf{R}_{kj}) \neq 0$. This is equivalent to that the constellation is non-planar (i.e., not all spacecraft are in the same plane).

The temporal variation rate $(\partial_t \mathbf{B})_c^{(0)}$ is readily obtained from central difference using time-series magnetic observation.

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Now the frame transformation relationships (3) are reduced to the apparent velocity $\mathbf{V}^{(0)}$ of the magnetic structure and the longitudinal components of the quadratic magnetic gradient $(\partial_3 \nabla \mathbf{B})^{(1)}$.

First, the zeroth approximation of the apparent velocity of the magnetic structure $\mathbf{V}^{(0)}$ can be found using the frame transformation relationship:

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$$(\partial_t \mathbf{B})^{(0)} = -\mathbf{V}^{(0)} \cdot (\nabla \mathbf{B})^{(0)}, \quad (6)$$

Then, using the relationship:

$$\partial_t (\nabla \mathbf{B})^{(0)} = -\mathbf{V}^{(0)} \cdot (\nabla \nabla \mathbf{B})^{(1)}, \quad (7)$$

200 the longitude components of the quadratic magnetic gradient at first order can be drawn as:

$$(\partial_3 \nabla \mathbf{B})_c^{(1)} = \frac{1}{V^{(0)}} \partial_t (\nabla \mathbf{B}(t, \mathbf{r}_c))^{(0)}, \quad (8)$$

which are just $(G_{31}^{(1)}, G_{32}^{(1)}, G_{33}^{(1)})$.

The remaining components of the quadratic magnetic gradients can be deduced using the least-squares method.

Assuming that:

$$205 \quad S = \frac{1}{N} \sum_{\alpha=1}^N \left[f_c^{(0)} + x_{(\alpha)}^i g_i^{(0)} + \frac{1}{2} x_{(\alpha)}^i x_{(\alpha)}^j G_{ij}^{(1)} - f_{(\alpha)} \right]^2, \quad (9)$$

which can also be written as:

$$S = \frac{1}{N} \sum_{\alpha=1}^N \left[f_c^{(0)} + x_{(\alpha)}^i g_i^{(0)} - f_{(\alpha)} + \left(1 - \frac{1}{2} \delta_{i3} \right) x_{(\alpha)}^i x_{(\alpha)}^3 G_{i3}^{(1)} + \frac{1}{2} x_{(\alpha)}^p x_{(\alpha)}^q G_{pq}^{(1)} \right]^2. \quad (10)$$

Herewhere, $p, q = 1, 2$.

If $\delta S = 0$, then

$$210 \quad \frac{\partial S}{\partial G_{pq}^{(1)}} = \frac{1}{N} \sum_{\alpha=1}^N 2 \left[f_c^{(0)} + x_{(\alpha)}^i g_i^{(0)} - f_{(\alpha)} + \left(1 - \frac{1}{2} \delta_{i3} \right) x_{(\alpha)}^i x_{(\alpha)}^3 G_{i3}^{(1)} + \frac{1}{2} x_{(\alpha)}^r x_{(\alpha)}^s G_{rs}^{(1)} \right] \cdot x_{(\alpha)}^p x_{(\alpha)}^q = 0. \quad (11)$$

which reduces to:

$$\begin{aligned} f_c^{(0)} \sum_{\alpha=1}^N x_{(\alpha)}^p x_{(\alpha)}^q + \sum_{\alpha=1}^N x_{(\alpha)}^i x_{(\alpha)}^p x_{(\alpha)}^q g_i^{(0)} - \sum_{\alpha=1}^N f_{(\alpha)} x_{(\alpha)}^p x_{(\alpha)}^q + \sum_{\alpha=1}^N \left(1 - \frac{1}{2} \delta_{i3} \right) x_{(\alpha)}^i x_{(\alpha)}^3 x_{(\alpha)}^p x_{(\alpha)}^q G_{i3}^{(1)} \\ + \frac{1}{2} \sum_{\alpha=1}^N x_{(\alpha)}^r x_{(\alpha)}^s x_{(\alpha)}^p x_{(\alpha)}^q G_{rs}^{(1)} = 0 \end{aligned} \quad (12)$$

Resulting in $G_{rs}^{(1)} (r, s = 1, 2)$, i.e., $(G_{21}^{(1)}, G_{22}^{(1)}, G_{11}^{(1)})$.

215 The constellation must be nonplanar to achieve this result. This is verified as the following.

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The equation (12) is that for the parameters $G_{rs}^{(l)} (r, s = 1, 2)$, i.e., $(G_{11}^{(l)}, G_{12}^{(l)}, G_{21}^{(l)}, G_{22}^{(l)})$. With the analysis similar to that in the subsection 4.1 of Following Zhou & Shen (2024), in order the solution exists, it is expected that, in order the solution exists, the position of all the spacecraft in the constellation must not obey the following formula

$$a_{11} \left(x_{(\alpha)}^1 \right)^2 + a_{12} x_{(\alpha)}^1 x_{(\alpha)}^2 + a_{12} x_{(\alpha)}^2 x_{(\alpha)}^1 + a_{22} \left(x_{(\alpha)}^2 \right)^2 = 0. \quad (13)$$

Wherever, $a_{rs} (r, s = 1, 2)$ are fixed coefficients. The above equations can be rewritten as

$$a_{11} \left(x_{(\alpha)}^1 / x_{(\alpha)}^2 \right)^2 + 2a_{12} \left(x_{(\alpha)}^1 / x_{(\alpha)}^2 \right) + a_{22} = 0. \quad (14)$$

Which reduce to $x_{(\alpha)}^1 / x_{(\alpha)}^2 = \text{constant}$. It means that all the spacecraft are in the plane parallel to the x_3 axis or the motion direction. Therefore, it is necessary that the constellation should not be planar in order to deduce the quadratic magnetic gradients as well as the linear magnetic gradient. The next iterations would also require this condition.

2.2.2 First order iteration

Assuming that:

$$S = \frac{1}{N} \sum_{\alpha=1}^N \left[f_c^{(l)} + x_{(\alpha)}^i g_i^{(l)} + \frac{1}{2} x_{(\alpha)}^i x_{(\alpha)}^j G_{ij}^{(l)} - f_{(\alpha)} \right]^2. \quad (15)$$

If $\delta S = 0$, then:

$$\frac{\partial S}{\partial f_c^{(l)}} = 0, \quad \frac{\partial S}{\partial g_i^{(l)}} = 0. \quad (16)$$

From $\frac{\partial S}{\partial f_c^{(l)}} = 0$, it can be assumed that:

$$\frac{1}{N} \sum_{\alpha=1}^N \left[f_c^{(1)} + x_{(\alpha)}^i g_i^{(1)} + \frac{1}{2} x_{(\alpha)}^i x_{(\alpha)}^j G_{ij}^{(1)} - f_{(\alpha)} \right] = 0. \quad (4517)$$

Meaning that:

$$f_c^{(1)} = \frac{1}{N} \sum_{\alpha=1}^N f_{(\alpha)} - \frac{1}{2N} \sum_{\alpha=1}^N x_{(\alpha)}^i x_{(\alpha)}^j G_{ij}^{(1)} = \frac{1}{N} \sum_{\alpha=1}^N f_{(\alpha)} - \frac{1}{2} R^{ij} G_{ij}^{(1)}. \quad (4618)$$

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If $\frac{\partial \mathcal{S}}{\partial g_i^{(1)}} = 0$, this can be reduced to:

$$\frac{1}{N} \sum_{\alpha=1}^N \left[f_c^{(1)} + x_{(\alpha)}^i g_i^{(1)} + \frac{1}{2} x_{(\alpha)}^i x_{(\alpha)}^j G_{ij}^{(0)} - f_{(\alpha)} \right] x_{(\alpha)}^k = 0, \quad (4719)$$

245 i.e.,

$$\frac{1}{N} \sum_{\alpha=1}^N x_{(\alpha)}^k x_{(\alpha)}^i g_i^{(1)} + \frac{1}{2} \frac{1}{N} \sum_{\alpha=1}^N x_{(\alpha)}^k x_{(\alpha)}^i x_{(\alpha)}^j G_{ij}^{(0)} - \frac{1}{N} \sum_{\alpha=1}^N f_{(\alpha)} x_{(\alpha)}^k = 0. \quad (4820)$$

The tensor $R^{kij} = \frac{1}{N} \sum_{\alpha=1}^N x_{(\alpha)}^k x_{(\alpha)}^i x_{(\alpha)}^j$ is then defined, resulting in:

$$R^{ki} g_i^{(1)} + \frac{1}{2} R^{kij} G_{ij}^{(0)} - \frac{1}{N} \sum_{\alpha=1}^N f_{(\alpha)} x_{(\alpha)}^k = 0. \quad (4921)$$

Therefore, the first magnetic gradient is:

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$$g_\ell^{(1)} = -\frac{1}{2} (R^{-1})^{k\ell} R^{kij} G_{ij}^{(0)} + (R^{-1})^{k\ell} \cdot \frac{1}{N} \sum_{\alpha=1}^N f_{(\alpha)} x_{(\alpha)}^k. \quad (2022)$$

Using equation (3), it is now possible to obtain the corrected apparent velocity $\mathbf{V}^{(1)}$ of the magnetic structure and the longitudinal components of the corrected quadratic magnetic gradient $(\partial_3 \nabla \mathbf{B})^{(2)} \left((\partial_3 \partial_i \mathbf{B})^{(2)} \right)$, as in the zeroth iteration.

The least-squares method is then used to obtain the remaining nine components of the corrected quadratic magnetic gradient.

If:

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$$\begin{aligned}
S &= \frac{1}{N} \sum_{\alpha=1}^N \left[f_{(c)}^{(1)} + x_{(\alpha)}^i g_i^{(1)} + \frac{1}{2} x_{(\alpha)}^i x_{(\alpha)}^j G_{ij}^{(2)} - f_{(\alpha)} \right]^2 \\
&= \frac{1}{N} \sum_{\alpha=1}^N \left[f_c^{(1)} + x_{(\alpha)}^i g_i^{(1)} - f_{(\alpha)} + \left(1 - \frac{1}{2} \delta_{i3} \right) x_{(\alpha)}^i x_{(\alpha)}^3 G_{i3}^{(2)} + \frac{1}{2} x_{(\alpha)}^p x_{(\alpha)}^q G_{pq}^{(2)} \right]^2
\end{aligned} \tag{2423}$$

Then $G_{pq}^{(2)}$ ($p, q = 1, 2$) can be obtained using the same procedure as that used for the zeroth iteration. So that all the components of the corrected quadratic magnetic gradient $(\nabla \nabla \mathbf{B})^{(2)}$ are obtained.

Similarly, two or more iterations can be performed until stable linear and second-order magnetic gradients are obtained.

This algorithm requires that the constellation be composed of at least seven spacecraft and that its configuration is non-planar. Because both the 9S/C HelioSwarm and 7S/C Plasma Observatory satisfy these requirements, the linear and quadratic magnetic gradients can be readily obtained.

The Curlometer technique (Dunlop et al., 2002b) is used to calculate the current density based on multiple spacecraft magnetic measurements, with the relative error estimated by the ratio between the divergence and curl of the magnetic field,

i.e., $\left| \frac{\nabla \cdot \mathbf{B}}{\nabla \times \mathbf{B}} \right|$. If the length and the magnetic field are normalized by the characteristic distance and magnetic strength (D, B_0) ,

the equation becomes $\left| \frac{\bar{\nabla} \cdot \bar{\mathbf{B}}}{\bar{\nabla} \times \bar{\mathbf{B}}} \right| \approx \left| \frac{\bar{\nabla} \cdot \bar{\mathbf{B}}}{1} \right| = |\bar{\nabla} \cdot \bar{\mathbf{B}}|$. Therefore, the dimensionless divergence of the magnetic field calculated

with observation data can be regarded a reasonable measure of the relative error within the linear magnetic gradient. Similarly,

the dimensionless $\left| \bar{\nabla} (\bar{\nabla} \cdot \bar{\mathbf{B}})_c \right|$ can be used as a measure describing the relative error in the quadratic magnetic gradient derived using the method.

3. Comparison of new method with analytical modelling

In this section, the new method is applied to two analytical magnetic field models: a cylindrical force-free flux rope and a dipole magnetic field to evaluate its validity and accuracy. The applicability of this approach was tested on the 7S/C Plasma Observatory (N=7) under the assumption that the seven-spacecraft cluster crosses a magnetic field structure (as illustrated in Figure. 1) by comparing the linear and quadratic gradients of the magnetic field obtained by the new method with those obtained by accurate modelling.

The positions of the seven spacecrafts in the barycentric coordinate system were generated randomly with Cartesian coordinates between -0.02 and $0.02 R_E$, as seen in Table 1. The 7S/C array is illustrated in Figure. 2.

280 **Table 1. Coordinates of the seven spacecraft in the barycentre coordinate system, with α denoting spacecraft number.**

α	$x_{(\alpha)} (R_E)$	$y_{(\alpha)} (R_E)$	$z_{(\alpha)} (R_E)$
1	0.0105	0.0016	0.0100
2	0.0135	0.0153	-0.0119
3	-0.0124	0.0155	-0.0026
4	0.0138	-0.0114	0.0139
5	0.0044	0.0157	0.0097
6	-0.0134	0.0152	0.0153
7	-0.0074	-0.0005	0.0052

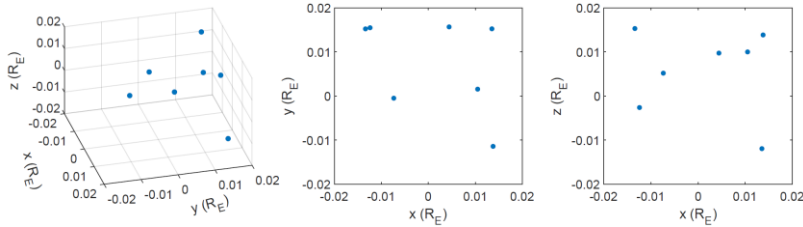


Figure 2. Configuration of the 7-S/C constellation.

The characteristic configuration of the spacecraft is described using several parameters. The three eigenvalues of the volumetric tensor R^j are represented by w_1 , w_2 , and w_3 (where $w_1 \geq w_2 \geq w_3$) (Harvey, 1998), with their square roots representing the characteristic half-widths of the S/C in the three orthogonal directions along the corresponding eigenvectors (Harvey, 1998). The characteristic size of the S/C is twice the square root of the maximum eigenvalue, $L = 2\sqrt{w_1}$ (Robert et al., 1998; Shen et al, 2012). For the 7-S/C constellation tested seven S/Cs in the cluster in this section, the three eigenvalues are $w_1 = 0.1643 R_E^2$, $w_2 = 0.1104 R_E^2$, and $w_3 = 0.0341 R_E^2$. $w_1 = 0.1643 \times 10^{-3} R_E^2$, $w_2 = 0.1104 \times 10^{-3} R_E^2$, and $w_3 = 0.0341 \times 10^{-3} R_E^2$. The characteristic size is $L = 2\sqrt{w_1} = 0.0256 R_E = 163.33 \text{ km}$.

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3.1 Flux Rope

The flux rope was assumed to be force-free and cylindrically symmetrical. The magnetic field of the flux rope can be described using the Helmholtz equation, for which Lundquist (1950) provided analytical solutions in terms of the Bessel functions.

$$B_r = 0, B_\phi(r) = B_0 J_1(\alpha r), B_z(r) = B_0 J_0(\alpha r), \quad (24)$$

where r is the radial distance from the centric axis; α is a constant, with $1/\alpha$ representing the characteristic scale of the flux rope; B_0 is the peak axial field intensity; and J_0 and J_1 are the zero- and first-order Bessel functions of the first kind, respectively. For this test, we set $B_0=60$ nT and $\alpha=1/R_E$.

The 7-S/C array was assumed to cross the flux rope in a straight line at uniform velocity. The array is represented by the barycentre with the red dot in Figure. 3, and moves from $(-2, 0, 0) R_E$ to $(2, 0, 0) R_E$ along the x -axis over a time interval of 100 s. The resolution of the magnetic field measurement was set to 1 s for the time-series observations, and the characteristic size of the 7-S/C array was set to $L=0.0256 R_E$ for the gradients of the magnetic field at the barycentre along the trajectory to be obtained.

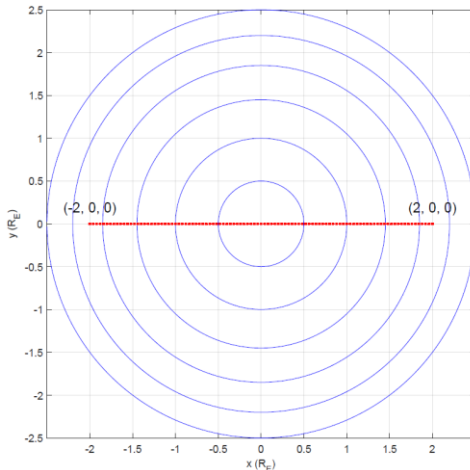


Figure 3. The cylindrical force-free flux rope crossing by the 7-S/C constellation as viewed from the axial direction. Trajectory of the barycentre of the constellation from $(-2, 0, 0) R_E$ to $(2, 0, 0) R_E$ over 100 s is shown by the red dotted line. Blue lines represent magnetic field lines.

The linear gradient of the magnetic field ($\nabla_i B_k$) has nine components, while the quadratic gradients ($\nabla_i \nabla_j B_k$) comprise 27 components. According to the analytical flux-rope model and symmetry of the quadratic gradients, only five

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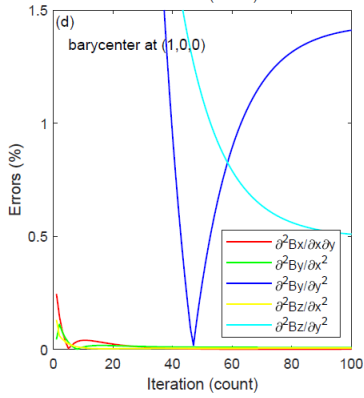
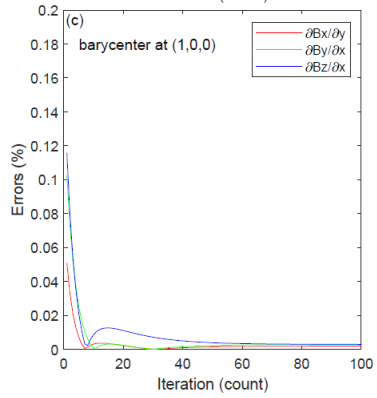
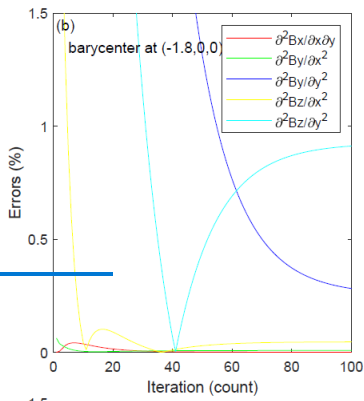
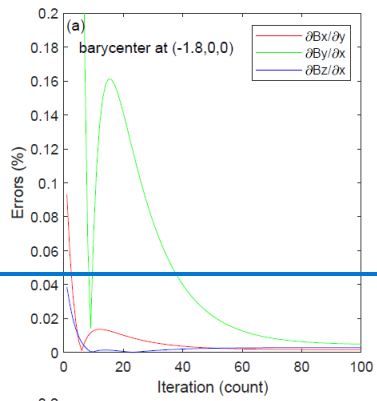
independent components of the quadratic gradients: $\nabla_1\nabla_2B_1$, $\nabla_1\nabla_1B_2$, $\nabla_2\nabla_2B_2$, $\nabla_1\nabla_1B_3$, and $\nabla_2\nabla_2B_3$, and three components of the linear gradients ∇_2B_1 , ∇_1B_2 , and ∇_1B_3 are nonzero points on the x -axis when using Cartesian coordinates, simplifying the comparison between the gradients derived from the proposed method and the analytical model.

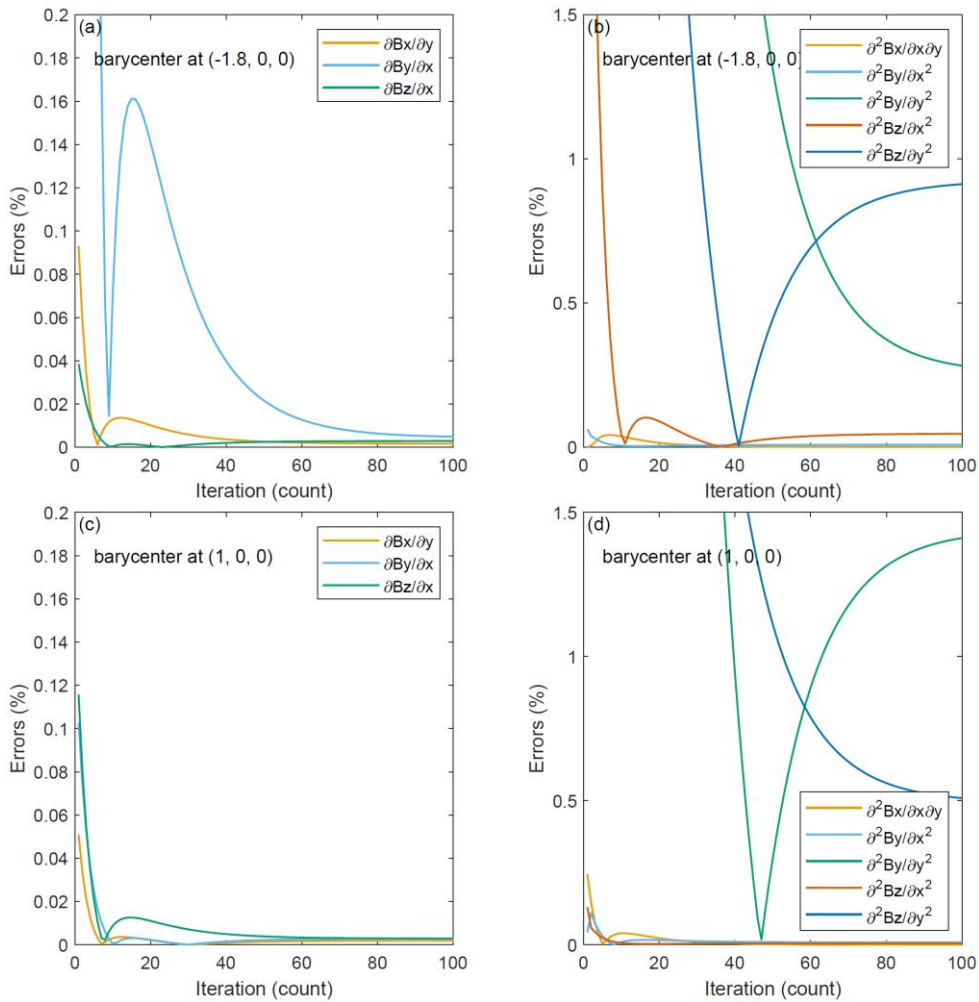
315 The impact of iteration on the results was investigated first, with the results at two different points used to demonstrate the variation in the relative errors under iteration, as illustrated in Figure. 4. The relative error is defined as

$\left| \left(X_{algorithm} - X_{accurate} \right) / X_{accurate} \right| \times 100\%$, where $X_{algorithm}$ and $X_{accurate}$ represent the algorithm gradients derived using the

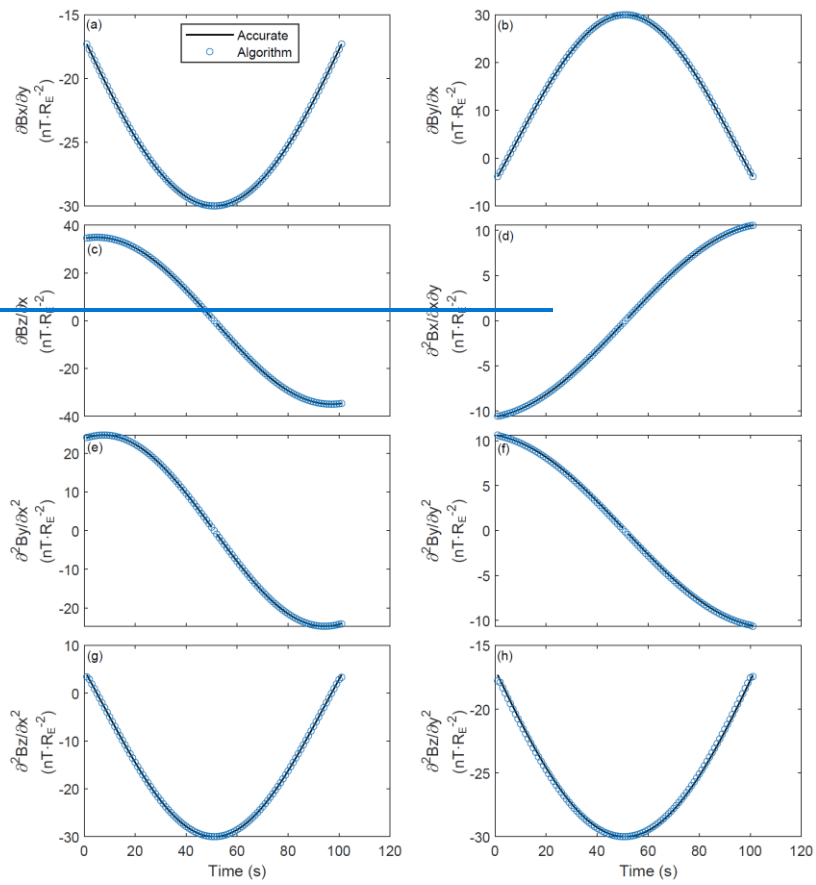
new method and accurate values from the analytical model at the barycentre, respectively. As shown in Figure. 4(a) and 4(c), the linear gradients converged to certain values within 50 iterations, and the final relative errors were less than 0.02%. Figure

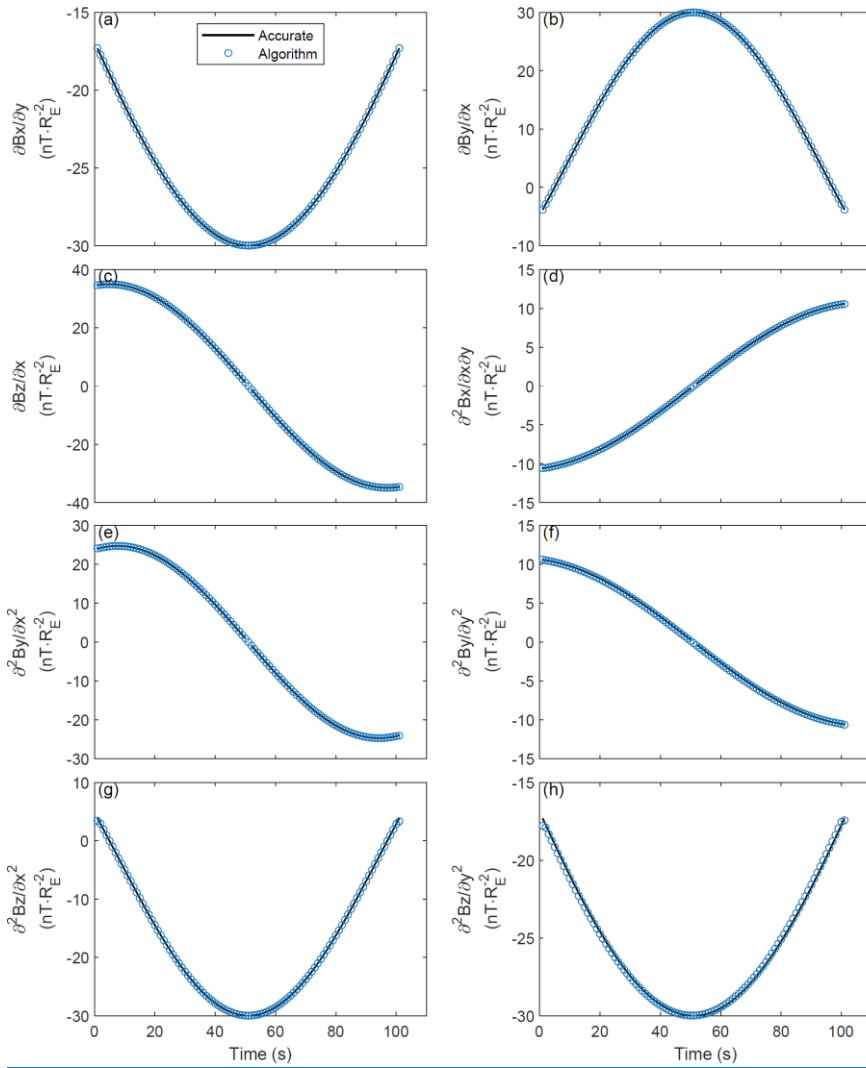
320 4(b) and 4(d) also indicate that the quadratic gradients converge. However, some quadratic gradients converged faster than others with fewer relative errors, and final relative errors of no more than 1.5% were obtained after 100 iterations. The maximum number of iterations was set to 100; thus, the gradients could be derived with good accuracy.





325 **Figure 4.** Relative errors in the nonzero components of the linear and quadratic gradients with different iteration numbers at different barycentre.





330 Figure 5. Time series showing nonzero components of the linear and quadratic gradients. Circles and solid lines represent the results obtained using the algorithm and accurate modelling, respectively.

Figure 5 shows a comparison of the nonzero linear and quadratic gradients at the barycentre derived from our method with those derived from the analytical model. The algorithm gradients are consistent with the accurate gradients, indicating that the proposed method is effective and precise when used with flux ropes.

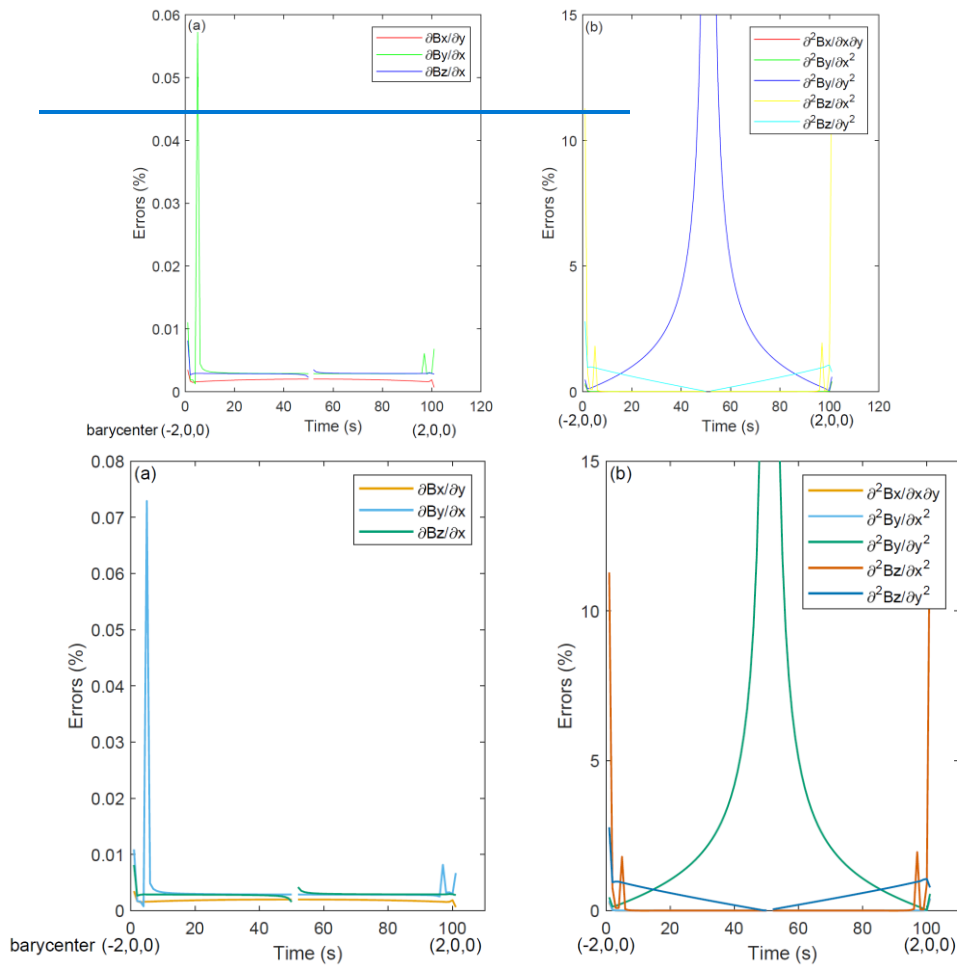


Figure 6. Relative errors in the nonzero components of the linear and quadratic gradients along the crossing path.

340 The relative errors of the gradients at points along the trajectory are shown in Figure. 6. All the relative errors of the linear gradients were less than 0.1%, and the vast majority of the relative errors for the quadratic gradients did not exceed 5%. It should be noted that the barycentre is at (0,0,0) at 50 s and that the nonzero components of the linear and quadratic gradients do not exist at (0,0,0). The barycentre is at (-0.04,0,0) R_E at 49 s, when accurate modelling and algorithm values for the quadratic gradient $\nabla_2 \nabla_2 B_2$ of 0.3 and 0.1570 nT $\cdot R_E^2$, respectively. The relative error approaches 50%; however, the absolute error is just 0.143, which is approaching zero. Symmetrically, the situation described is the same as it would be if the barycentre were at (0.04,0,0).

3.2 Dipole field

350 The proposed method was also tested and verified using a magnetic dipole field. The geomagnetic dipole field is mathematically expressed as:

$$\begin{cases} B_x = -\frac{3xz}{r^5} B_0 \\ B_y = -\frac{3yz}{r^5} B_0 \\ B_z = \frac{r^2 - 3z^2}{r^5} B_0 \end{cases} \quad (25)$$

where B_0 is the magnetic field at the Earth's equator and is defined by $B_0 = \frac{\mu_0 M}{4\pi R_E^3} = 30008 \text{ nT}$; $M = 7.76 \times 10^{22} \text{ A} \cdot \text{m}^2$

is the geomagnetic moment, with its direction set anti-parallel to the z-axis; x , y and z are the coordinates of the field points measured by R_E , and $r = \sqrt{x^2 + y^2 + z^2}$ is the radial distance from the origin measured by R_E .

355 The 7-S/C array was assumed to cross the dipole field in a straight line at constant velocity, with the barycentre parallel to the x -axis and moving from (-5, 0, 5) R_E to (5, 0, 5) R_E over 125 s, as illustrated in Figure. 7. The resolution of the magnetic field measurement was set to 1 s and the characteristic size of the 7-S/C array was set to 0.0256 R_E , which is the same as that of the flux-rope case, for the gradients of the magnetic field at the barycentre along the trajectory to be obtained.

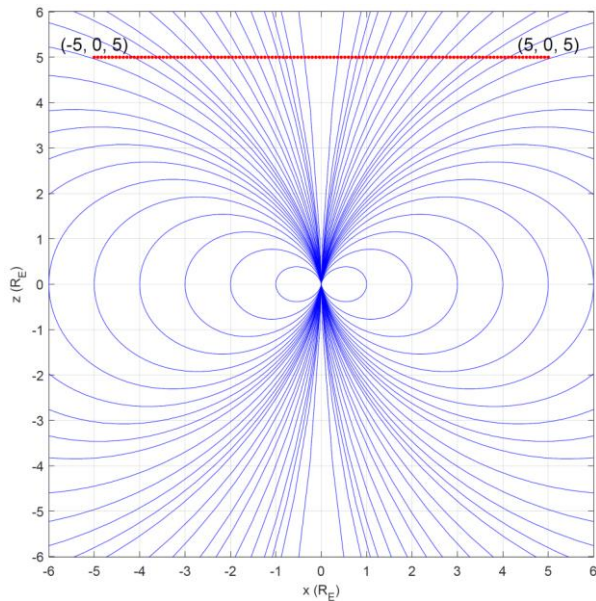
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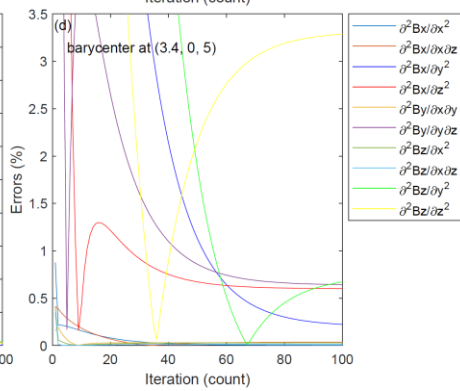
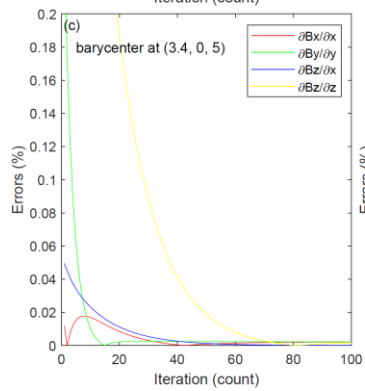
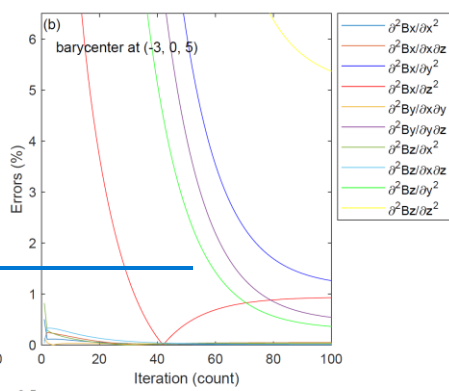
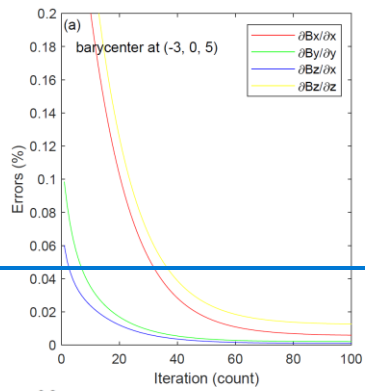


360 **Figure 7.** The magnetic dipole field crossed by the 7-S/C array. Trajectory of the barycentre of the 7-S/C array is from $(-5, 0, 5) R_E$ to $(5, 0, 5) R_E$ over 126 s as shown by the red dotted line. Blue lines represent magnetic field lines, which geometry is demonstrated.

Only nonzero independent components are displayed, similar to the flux rope case. In view of the mathematical expression of the dipole field, ten independent components of the quadratic gradients and four independent components of the linear gradients were nonzero along the crossing path.

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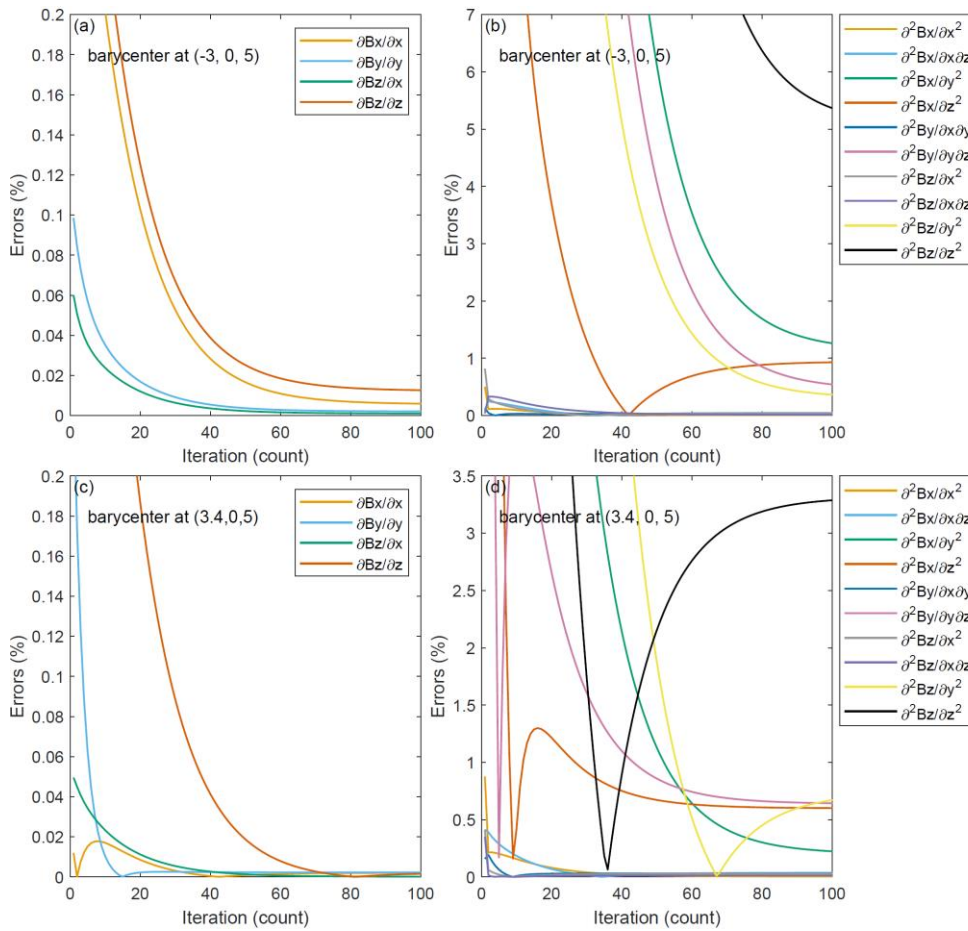
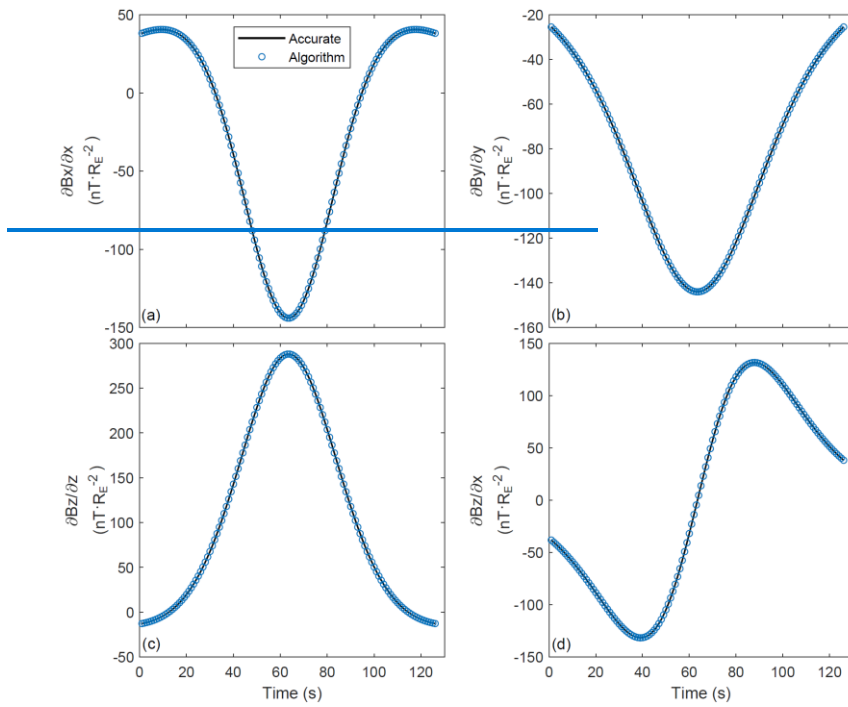


Figure 8. Relative errors in the nonzero components of linear and quadratic gradients with different iteration numbers at different barycentre.

Figure 8 shows the variation in the relative errors under iteration at two different points. As shown in Figure. 8(a) and 8(c), the linear gradients converged to certain values within 60 iterations, with final relative errors of less than 0.02%. Figure 8(b) and 8(d) show that the quadratic gradients also converge to low errors. After 100 iterations, most of the relative errors of

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the quadratic gradients were less than 1%, and the largest relative error was no more than 6%. These results suggest that it is reasonable to set the maximum number of iterations to 100 for the gradients to be derived with good accuracy in this case.



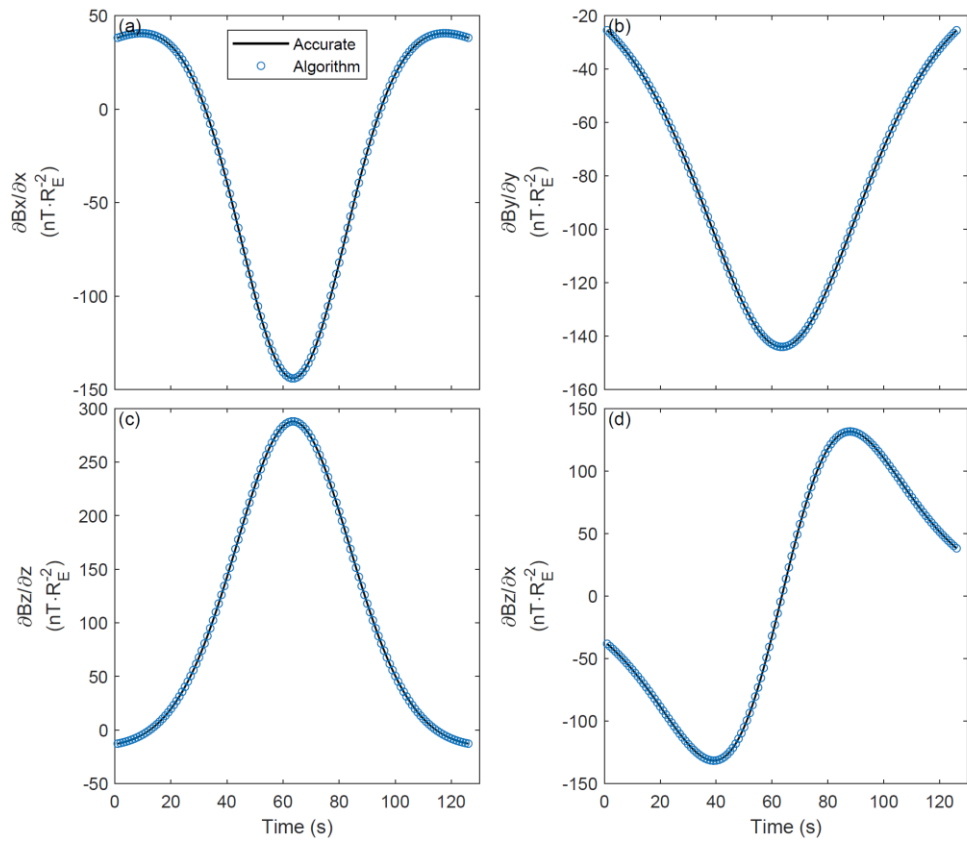
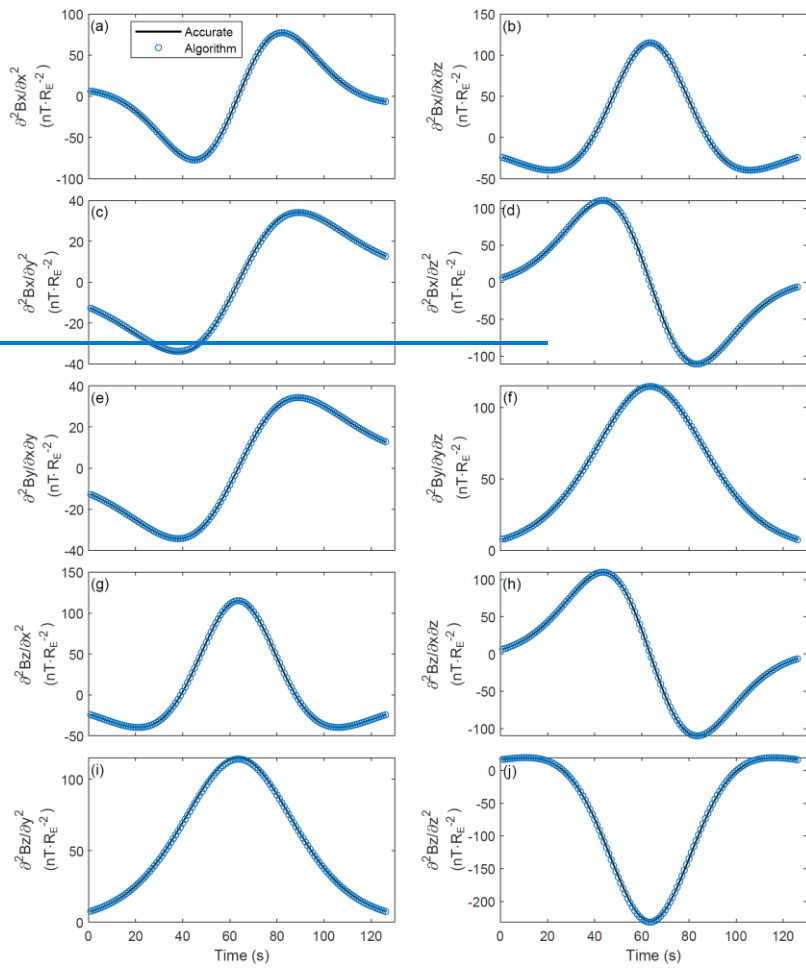
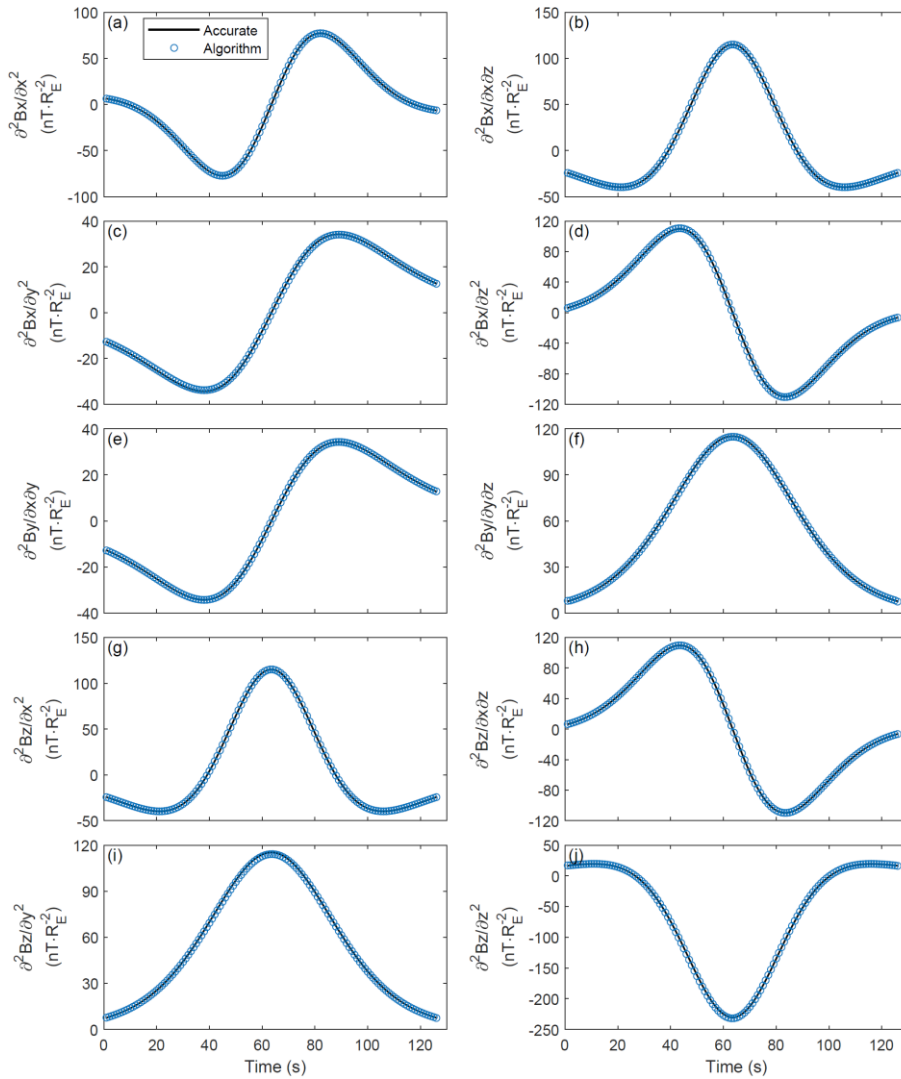


Figure 9. Time series showing nonzero components of the linear gradients. Circles and solid lines represent the results obtained using the algorithm and accurate modelling, respectively.





380 **Figure 10.** Time series showing nonzero components of the quadratic gradients. Circles and solid lines represent the results obtained using the algorithm and accurate modelling, respectively.

Figure 9 shows a comparison of the nonzero linear gradients derived from our method with those derived from the analytical model. A comparison of the nonzero quadratic gradients derived from the different sources is shown in Figure. 10. Both Figure. 9 and 10 indicate that the algorithm gradients are entirely consistent with those obtained from the accurate model, suggesting that the developed method is effective and precise for use with the dipole field.

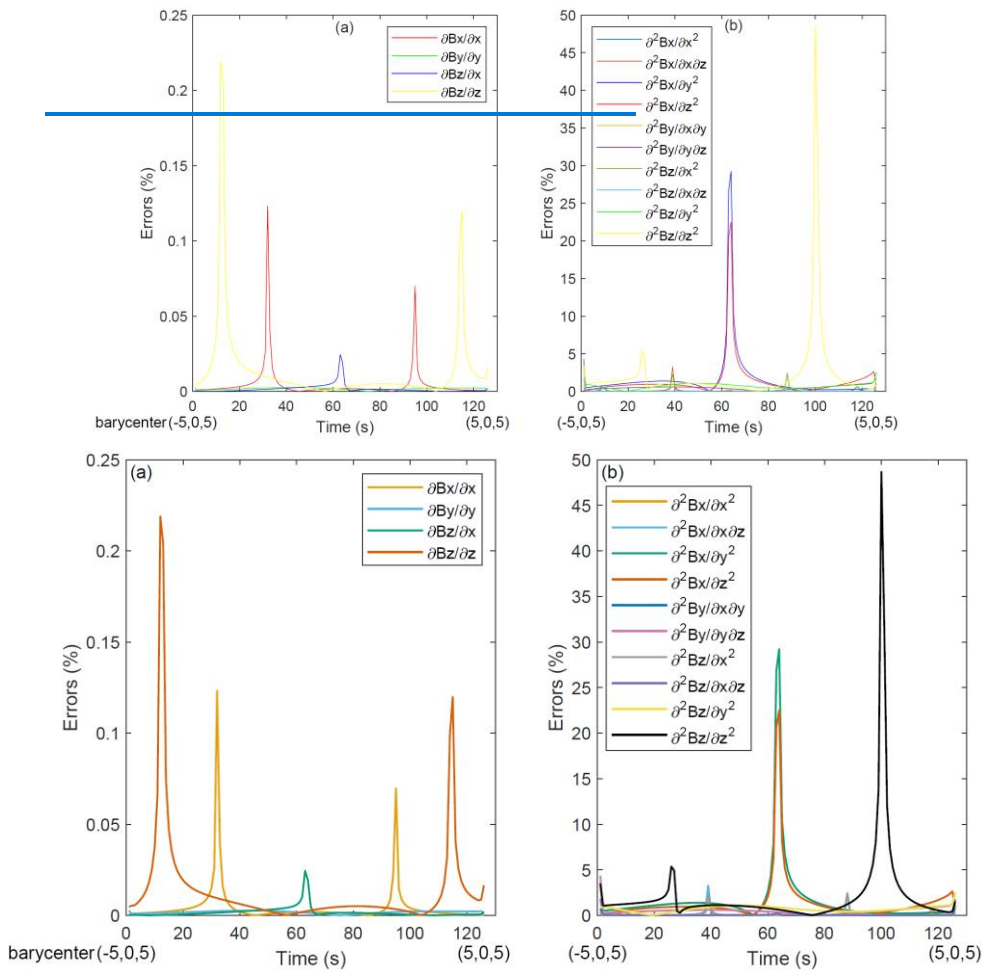


Figure 11. Relative errors in the nonzero components of the linear and quadratic gradients along the crossing path.

390 Figure 11 shows the relative errors of the gradients at the measured points along the crossing path. All the relative
errors for the linear gradients were less than 0.25 %, and most of the relative errors in the quadratic gradients were less than
5%. It should be noted that the barycentre is at (2.92, 0, 5) R_E at 100s, and the accurate and algorithm quadratic gradients
 $\nabla_3 \nabla_3 B_3$ are -1.2584 and -0.6461 $\text{nT} \cdot R_E^2$, respectively. The relative error approaches 50%; however, the absolute error is
low. The barycentre is at (-0.04, 0, 5) R_E at 63 s and (0.04, 0, 5) R_E at 64 s. Similarly, the absolute errors in the quadratic
395 gradients $\nabla_2 \nabla_2 B_1$ and $\nabla_3 \nabla_3 B_1$ were no more than 1 $\text{nT} \cdot R_E^2$, whereas the relative errors were approximately 30%.

3.3 Discussion

The two analytical magnetic field models (cylindrical force-free flux rope and dipole magnetic field) are simplified
and highly symmetrical structures with portion of the number of linear and quadratic gradients. The linear gradient of the
magnetic field has 9 components, while the quadratic gradients comprise 18 independent components due to the symmetry of
400 quadratic gradients. For the flux rope case, only 3 components of linear gradient and 5 components of quadratic gradients have
been assessed. But for dipole field case, 4 components of linear gradient and 10 components of quadratic gradients have been
assessed. The number of assessed parameters has reached half. However, only a subset of the 9+18=27 components can be
assessed. In this study, we have chosen so symmetrical model magnetic field in order to easily compare the simulation results
with the accurate analytic calculations. Nevertheless, these are still somewhat complete tests because the zero components of
405 the magnetic gradients are calculated with the algorithm as well and checked. Accordingly, further evaluations on the algorithm
could be made with the modelled magnetosphere with less symmetry in the future.

4. Errors

In this Section, we consider the diverse sources of errors that can impact the linear and quadratic magnetic spatial
gradient estimation as the following.

410 A spatial scale comes into play through the finite difference representation of local time derivatives, namely, the
product of the intrinsic time scale (time resolution) with the velocity of the magnetic structure in the spacecraft frame. This
will generate discretisation errors. Generally, the separation of the spacecraft in one constellation is several 100km to several
1000km. For the time dimension, the time resolution of the magnetic measurement Δt is about 0.01 sec, i.e. $\Delta t = 0.01 \text{sec}$.
415 Considering that the magnetic structure is moving at a velocity $V < 500 \text{ km/s}$, the spatial resolution along the motion direction
is about $v \Delta t < 5 \text{ km}$, which is much less than the S/C separation. Therefore, it can be expected that the discretisation errors
brought will be much less.

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The convergence properties of the iterative method are discussed in Section 3. A particular type of error considered there is the mismatch between the actual limit of the procedure and the approximation reached after a finite number of iterations. This error may be termed iteration error. It is not associated with the finite resolution of the spacecraft array or the time series. The iterative procedures have made our problem a linear one. This will help reduce the calculation error and make the calculation more stable. Nevertheless, it is not easy to get the formula of the iteration error. The detailed evaluation on the iteration accuracy of this algorithm can be made in the future when the real mission data are available.

Due to noise or measurement inaccuracies of input data, the estimated parameters (first and second derivatives) will be subject to random errors. The noise or disturbances in the data can come from the measurement error or the presence of high-frequency (physical) fluctuations such as those from plasma waves, which can make the calculation of the high order magnetic gradients very difficult (Shen et al. 2021a). When analysing the actual observation data, filtering methods should be employed to remove the high frequency components and avoid the negative effect of the noise. This process would help to extract large-scale magnetic structures under the consideration.

In the following, only truncation error has been evaluated. The discretisation error, iteration error, measurement error and random error will be presented elsewhere when the real measurements become available.

In Section 3, relative error is used to evaluate the truncation error of the proposed method. However, in some cases, evaluation with the relative error is not effective, while the gradient obtained from the accurate model is very small. Furthermore, the truncation error was evaluated under divergence-free magnetic field conditions.

Theoretically in theory, the divergence of the magnetic field and the gradient of the magnetic field divergence are both exactly zero, as shown by $\nabla \cdot \vec{B} = 0$, $\nabla \cdot \mathbf{B} = 0$ and $|\nabla(\nabla \cdot \vec{B})| = 0$, $|\nabla(\nabla \cdot \mathbf{B})| = 0$. To offer a uniform standard for evaluation, the divergence and gradient of divergence were non-dimensionalized with the corresponding characteristic quantity. The length was calibrated with the spatial characteristic scale of the magnetic structure D , and the magnetic field was calibrated with the characteristic magnetic field at the barycentre B_c . Therefore, two evaluation indices were introduced, represented by

$\frac{|\nabla \cdot \vec{B}|}{B_c}$, $\frac{|\nabla(\nabla \cdot \vec{B})|}{B_c}$ and $\frac{|\nabla \cdot \mathbf{B}|}{B_c}$, $\frac{|\nabla(\nabla \cdot \mathbf{B})|}{B_c}$. The values of the two indices can be used to evaluate the accuracies of the

linear and quadratic gradients derived using the proposed method. Nevertheless, this evaluation is not perfect because it cannot include all partial components of the magnetic gradients (the formula $\nabla \cdot \mathbf{B} = 0$ contains 3 of the total 9 components of $\nabla \mathbf{B}$ while $|\nabla(\nabla \cdot \mathbf{B})| = 0$ contains 9 of the total 18 components of $\nabla \nabla \mathbf{B}$). The advantage to use them as the measures of the errors of the magnetic gradients is that they are robust and simple. We still have not found other better ways for evaluating the accuracy of the algorithm because the actual values of the magnetic gradients are unknown when analysing the real observation data.

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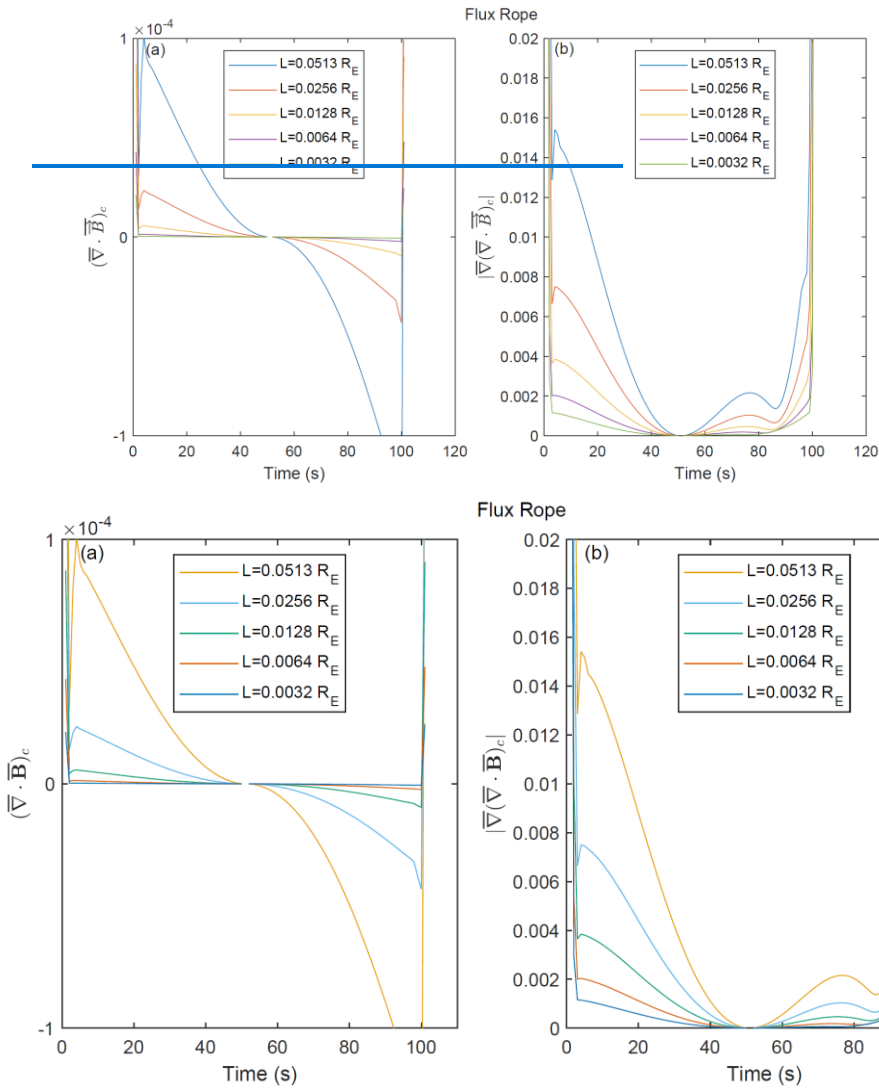


Figure 12. Dimensionless divergence and gradient of divergence for magnetic field along the flux rope crossing path with different characteristic S/C sizes (L).

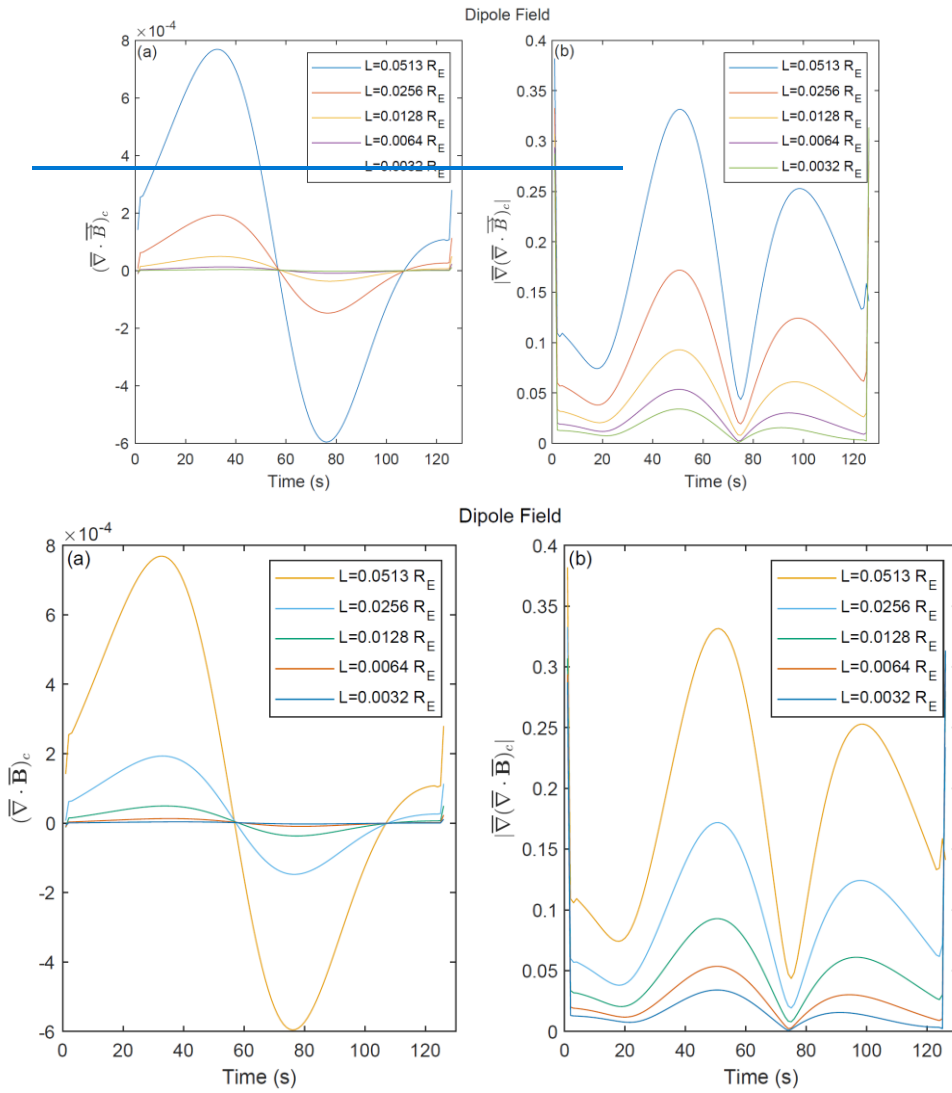


Figure 13. Dimensionless divergence and gradient of divergence for magnetic field along the dipole field crossing path with different characteristic S/C sizes (L).

The algorithm gradients were utilized to calculate the dimensionless divergence and the gradient of divergence for magnetic field at the barycentre along the crossing path with different characteristic S/C sizes, with the results for the flux-rope and dipole-field cases shown in Figure. 12 and 13, respectively. Figure 12 (a) and 13 (a) show that the dimensionless divergence $(\bar{\nabla} \cdot \bar{\mathbf{B}})_c$ at the barycentre is in the order of 10^{-4} , while L varies from 0.0032 to 0.0513 R_E . The dimensionless gradient of the divergence $\left| \bar{\nabla} (\bar{\nabla} \cdot \bar{\mathbf{B}})_c \right|$ for the flux-rope case was less than 0.02 with L=0.0513 R_E , as shown in Figure. 12 (b). Similarly, Figure. 13 (b) shows that $\left| \bar{\nabla} (\bar{\nabla} \cdot \bar{\mathbf{B}})_c \right|$ was less than 0.4, with L=0.0513 R_E , for the dipole field. Meanwhile, that $\left| \bar{\nabla} (\bar{\nabla} \cdot \bar{\mathbf{B}})_c \right|$ decreased with decreasing L in both cases. These results confirm the accuracy of the proposed method. As evidenced in Figures 12 and 13, the errors of the first derivative decrease quadratically with the scale L whereas the errors of the second derivatives decrease linearly with L.

5. Conclusions

In this study, a new algorithm was derived to estimate the linear and quadratic spatial gradients of the magnetic field from simultaneous 7- or 9-point magnetic measurements to obtain the fine structure of the magnetic field and the magnetic field geometry, allowing elucidation of whether the 7-spacecraft Plasma Observatory and the 9-spacecraft HelioSwarm missions could be utilized for such measurements. By inputting simultaneous 7–9 point magnetic measurements and using the reference frame transformation relationships of the magnetic field as well as the least-squares method, the new algorithm performs several iterations to finally derive the convergent magnetic linear and quadratic spatial gradients.

The developed algorithm requires fewer restrictions on the spatial configuration of the constellations and only demands that the constellations are non-planar. Actual operating constellations can easily satisfy this constraint. Only homogeneous-simultaneous magnetic measurements are required, with no other physical measurements needed, and the only physical constraint of the algorithm is the reference-frame transformation relationship of the magnetic field. In this study, simultaneous magnetic measurements from 7 or 9 points were assumed to be able to be obtained from identical instruments onboard mothercraft and daughtercraft of the space mission. We also note that the magnetic field data from different detectors need to be synchronized by the time interpolation. In reality, a homogeneous set of instruments onboard spacecraft may not be achieved, and the temporal measurements at different detectors may not be perfectly synchronized. In addition, other systematic errors including measurement errors of the magnetic field are not considered and they are beyond the scope of this

480 [study](#). In this study, only total truncation error has been evaluated. The discretisation error, iteration error, measurement error
and random error could be evaluated in details in the future. ~~the homogeneous measurement has been adopted in both the
method and tests. For actual multi-satellite measurements, the measurement could be not homogeneous. The magnetic field
data from different detectors need to be synchronized by the time interpolation.~~ Divergence-free magnetic field conditions
485 were not required to calculate the magnetic [spatial](#) gradient. Alternatively, in the algorithm, the magnitudes of the magnetic
divergence and its gradient were used to evaluate the truncation errors of the linear and quadratic magnetic [spatial](#) gradients,
respectively.

The proposed algorithm was verified using a cylindrical force-free flux rope and a dipole magnetic field, with results
showing that the iterations effectively converged and that the magnetic [spatial](#) gradients can reach reasonable accuracy. The
results of this study can thus be applied to the analysis of magnetic field data from multi-spacecraft constellations (e.g., the
490 Plasma Observatory and HelioSwarm) as well as to the design of future constellation missions.

Author contributions

CS designed the method [and RK gives some advice](#). GZ developed the model code and performed the test. CS and GZ wrote
495 the paper. CS and RK
reviewed and edited the paper. [YZ helps to evaluate the method](#).

Competing interests

The authors declare that they have no conflict of interest.

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(ISSI) in Bern through the ISSI International Team project #556 (cross-scale energy transfer in space plasmas). [Work at IRAP
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