Electron radiation belt safety indices based on the SafeSpace modelling pipeline and dedicated to the internal charging risk

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Abstract.

In this paper, we present the SafeSpace prototype for a safety warning system, dedicated to the electron-radiation-belt induced internal charging hazard, aboard spacecraft. The space weather tool relies on a synergy of physical models associated in a chain that covers the whole Sun-interplanetary space-Earth's inner magnetosphere medium. With the propagation of uncertainties along the modelling pipeline, the safety prototype provides a global nowcast and forecast (within a 4-day lead time) of the electron radiation belt dynamic as well as tailored indicators for space industry operators. They are meant to inform the users about the severity of the electron space environment via a three colored alarm system, which sorts the indices intensity according to a representative historical distribution of in-situ data. The system was challenged over the St-Patrick 2015 storm in order to assess its performance. It showed overall good nowcasting and forecasting capabilities, due to its broad physics-driven pipeline.

1 Introduction

Since the advent of the space era, space applications (commercial, technical, and scientific) have become progressively more and more important for our societies. Earth observation, telecommunication, and geo-positioning are among the needs covered by the space industry – to which one can add scientific activities (Devezas et al., 2012; George, 2019). These activities involve different types of actors such as spacecraft operators, space agencies, governmental institutions, technology corporations and aerospace manufacturers. With over 5000 orbiting spacecraft in 2022 (Satellite Industry Association) and an expected exponential growth for the next decade due to the implication of the "new space" players (Mann et al., 2018; Kodheli et al., 2020), the need for a well-specified operating environment is essential.

The space radiation environment is one of the most challenging constraints met by human space-based activities (Koons et al., 1999). Indeed, the near-Earth space environment – and, more generally-speaking, the inner magnetosphere – is filled with several types of particle populations, including trapped energetic electrons and protons (shaping the so-called radiation

belts), solar energetic particles resulting from the solar eruptive activity, and cosmic emissions (Bourdarie and Xapsos, 2008; Zheng et al., 2019). These populations have specific multi-scale complex dynamics, some components of which are due to their interaction with the Earth's electromagnetic configuration and the solar wind-magnetosphere coupling. (Russell and Thorne, 1970; Lyon, 2000).

In-orbit spacecraft can suffer temporary failures and/or permanent damages caused by their interaction with such energetic charged particles. These disruptions, that depend on the nature and energy range of the incident particles, can be classified according to the physical nature of the damage caused, among which are:

- surface and internal charging phenomenons, due to the build-up of charge on surfaces and leading to electrostatic discharges affecting sensitive components (Garrett and Whittlesey, 2000);

- single event effects caused by energetic protons and ions that deposit energy inside integrated circuits, causing component upset or even destruction (Petersen, 2011; Caron et al., 2018);

- increased drag due to the heating of the high atmosphere by higher ultraviolet solar emissions during active times, leading to recurrent orbital changes (Doornbos and Klinkrad, 2006);

- impacts on spacecraft communications and navigation accuracy, due to the local disruption of the ionosphere (radio signal propagation medium) leading to scintillation, deviation, attenuation or absorption of radio signals (Roston, 1970);

- cumulative degradation of materials due to energy deposit by incident particles or photons, leading to ionization or atomic displacement (Dever et al., 2005).

Note that the exposure of satellites to these hazards depends strongly on their orbit. For instance, Low-Earth Orbit (LEO) satellites are more likely to be subject to drag and trapped radiation, Medium-Earth Orbit (MEO) satellites undergo strong radiation doses and deep charging, while Geosynchronous Earth Orbit (GEO) satellites go through a highly variable radiation environment and occasional solar energetic protons (McCormack, 1988; Benton and Benton, 2001; Bourdarie and Xapsos, 2008).

Due to the near impossibility of performing in-situ repairs on damaged satellites, manufacturers impose conservative assumptions through the satellite design phase to ensure adequate robustness during the operational life of the spacecraft (Durante, 2002; Durante and Cucinotta, 2011). More precisely, they rely on specification or engineering models derived from statistical and empirical studies associated with the field of space climatology (Vette, 1991; Mursula et al., 2007; Sicard et al., 2018). Quite recently, the growing need for nowcasting and forecasting of small-scale variations in the space environment has led to the development of the field of space weather (Rockville, 2019). Space weather services enable space operators to effectively deploy protective measures during periods of intense geomagnetic activity and to mitigate its effects, in addition to improving current situational awareness, operations scheduling, testing, maintenance, and post-event analysis (Bothmer and Daglis, 2007; Moldwin, 2022). This evolution was made possible by the consequent improvement of the physical and empirical models which describe the dynamics of the various inner magnetosphere's components (plasma waves, electron plasma density, radiation belts...) (Tsyganenko, 2013; Fok, 2020; Ripoll et al., 2020).

In this context, the European Union's Horizon 2020 SafeSpace project aims at improving the space weather nowcasting and forecasting capabilities by building a safety service prototype (Daglis, 2022). The latter relies on a chain of physical and semi-empirical models, covering the whole Sun-to-Earth domain. The synergy of these models allows the propagation of physical information and uncertainties that are used to estimate tailored particle radiation indicators for the space industry. In addition, the SafeSpace prototype provides a 2-to-4 days forecast window, enabling an early warning system for hazardous space weather events. In this paper, we present new developments carried out to construct radiation belt activity indices specifically related to the risk of internal charging due to electron radiation belts. More specifically, these indices must meet specific industrial needs, provided here by Thales Alenia Space, that represents the industrial member in the SafeSpace team (Daglis, 2022).

This article is organized as follows. First, we present the SafeSpace modelling pipeline and its components (Section 2). Then, in Section 3 we focus on the methodology for constructing the new activity indices. In Section 4 we provide an example of indices nowcast and forecast during a real geomagnetic storm and compare their accuracy against observation-deduced indices. We conclude by briefly discussing these results and highlighting their strengths and weaknesses, which opens the way for future developments.

2 Presentation of the SafeSpace modelling pipeline

The SafeSpace safety service relies on a physical modelling pipeline that enables ensemble nowcast and forecast of the electron outer radiation belt fluxes, as shown in Figure 1. The pipeline can be roughly subdivided into three main segments. The first one is the heliospheric propagation segment, in which the Multi-VP, Helio1D (1D MHD), and EUHFORIA codes are chained together to simulate the development and the propagation of the solar wind and Coronal Mass Ejections (CMEs) from the surface of the Sun (using photospheric magnetograms as inputs) to the interplanetary region (Samara et al., 2021, 2022; Kieokaew et al., 2022). This segment provides estimates of three solar wind parameters at L1 (density, velocity and temperature) as well as the B_y component of the Interplanetary Magnetic Field (IMF), as an ensemble forecast comprising 21 members. These parameters are also fed to OGNN (ONERA Geoeffectiveness Neural Network), whose aim is to estimate the geomagnetic index Kp (Matzka et al., 2021), or, in other words, the geoeffectiveness of the estimated solar wind conditions. OGNN is a neural network based on Long Short Term Memory recurrent layers (Hochreiter and Schmidhuber, 1997), which have been extensively used in the community to forecast geomagnetic indices (Gruet et al., 2018; Chakraborty and Morley, 2020; Bernoux et al., 2021). Note that the B_z component of the IMF, which is notoriously one of the most important components of the IMF for predicting the geoeffectiveness of the solar wind and solar transients (see e.g. Richardson and Cane, 2011), is not reliably estimated by Helio1D, which is why we have to restrict ourselves for B_y in this study. In (Brunet et al., 2022) (currently under review), the authors discuss the implications of the lack of B_z for the SafeSpace modelling pipeline capabilities.

The second segment is the inner magnetosphere modelling segment (for which an exhaustive description is available in (Brunet et al., 2022, submitted and currently under review)). Its purpose is to prepare and estimate the inputs for the Salammbô-EnKF, which is the combination of
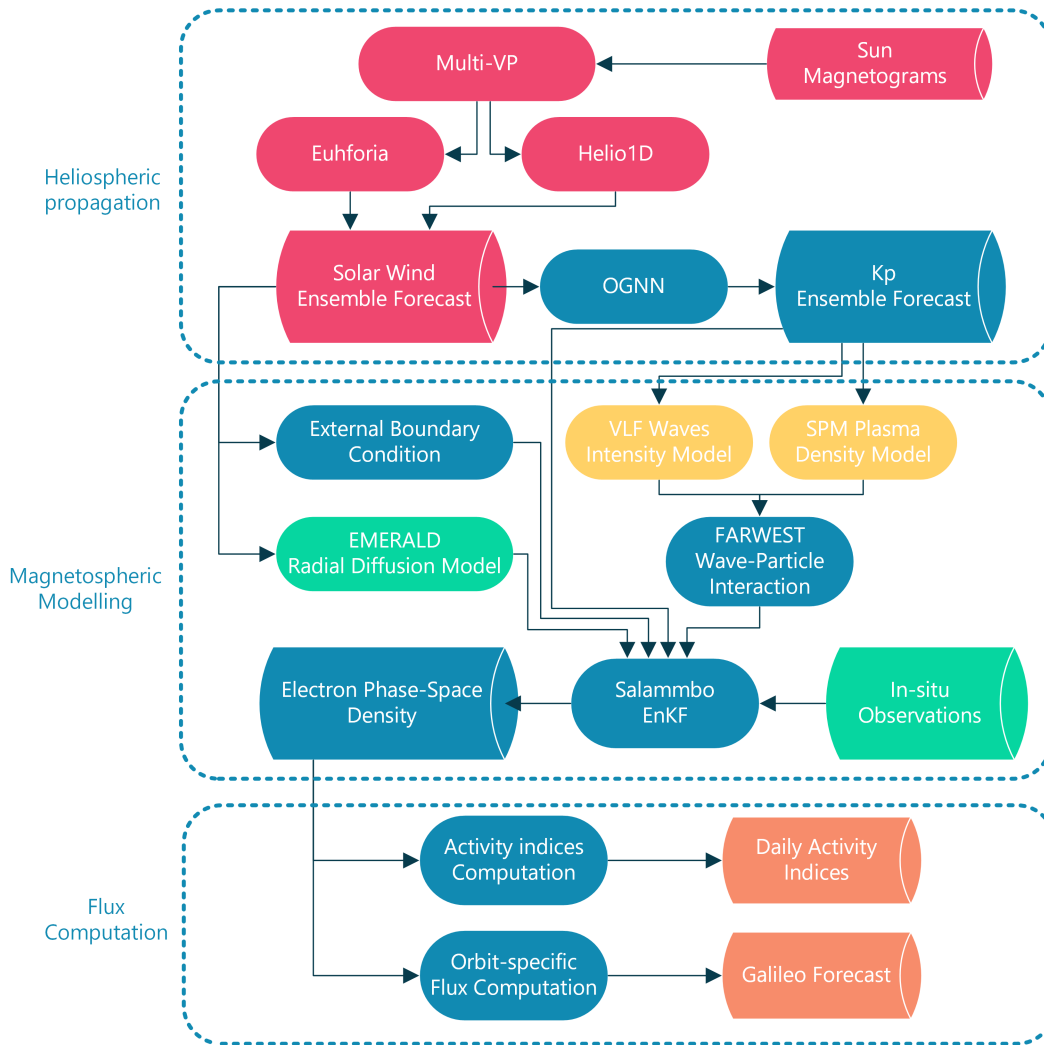
Figure 1. The SafeSpace pipeline, describing the Sun–interplanetary space–Earth's inner magnetosphere interactions.

- the Salammbô-Electron 3D code, a physical model that solves the Fokker-Planck equation describing the statistical distribution of magnetically trapped electrons in the (Ec, $y = \sin(\alpha_{eq})$, L^*) phase space (Beutier and Boscher, 1995; Bourdarie and Maget, 2012), with Ec the electron's energy, α_{eq} its equatorial pitch angle with Earth's magnetic field and L^*, the Roederer parameter (Roederer, 2012).

- the Ensemble Kalman Filter (EnKF), a sequential data assimilation tool that combines model forecasts from Salammbô and in-situ data to determine an optimal estimation of the actual conditions (Evensen, 2003). It is expected that the implementation of Salammbô-EnKF in the fully operational SafeSpace pipeline, will operate with observations from the
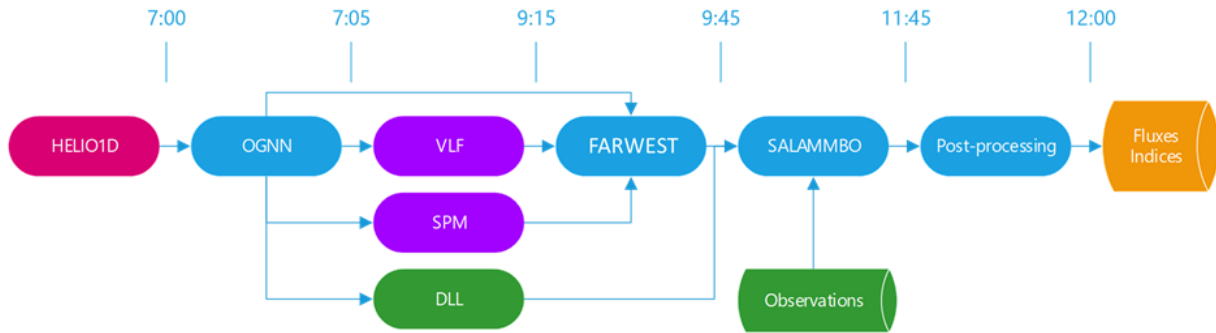
Figure 2. The timeline of the SafeSpace pipeline. The time codes refer to the approximate computation end time of the previous computational step and by extension the time when output data is available for the next computation step.

NOAA GOES-16 electron monitor (Onsager et al., 1996) as well as the EMU instrument onboard Galileo GSAT 0207 and 0215 satellites (Sandberg et al., 2019).

The input data required by Salammbô-EnKF (in the SafeSpace project context) are:

– time-dependent diffusion rates describing the radial diffusion process, using the solar wind-driven neural network-based EMERALD model (Aminalragia-Giamini et al., 2022).

– time-dependent pitch angle and energy diffusion rates, induced by resonant interaction with waves and computed by the FARWEST code (Dahmen et al., 2022). The local diffusion computations in FARWEST are based on the cold plasma cartography estimated by SPM (Pierrard et al., 2021a; Botek et al., 2021) and the VLF wave distributions for the whistler mode hiss, upper and lower chorus as presented in (Santolík et al., 2021).

– boundary conditions provided by the solar wind-driven GEO model (Katsavrias et al., 2021).

For each one of the 21 solar wind ensemble members derived in the first segment, we compute an internal ensemble of magnetospheric parameters consisting of 10 members. Therefore, an ensemble totalling $21 \times 10 = 210$ parameters is fed hourly to Salammbô-EnKF to compute two types of trapped electron flux maps: nowcast flux maps and forecast flux maps with a 4-day lead when possible.

In the SafeSpace pipeline's third segment, the computed flux maps are used to derive electron radiation belt activity indices. These user-driven indices are the core of our paper and are extensively described in the next section.

In terms of computational cost, the SafeSpace pipeline requires approximately 5 hours to provide its estimation of the radiation belt activity indices, after following a timed computational sequence represented in figure 2. It shows that the computational cost is shared almost exclusively between the building of the input data for Salammbô (around 3 hours) and the Salammbô-EnKF simulation (around 2 hours).

3 Radiation belt activity indices

3.1 Description of the indices

The electron radiation belt activity indices are the keystone of the SafeSpace safety service. They are meant to indicate how severe the electron space environment is at a given time. They can be nowcast in real time, and forecast with a 4-day lead. The nowcast indices are meant e.g. for post-event anomaly analysis, and the forecast ones can be used to prevent some anomalies on-board spacecraft. These indices are based on daily averaged electron fluxes. Since the electron population varies greatly according to the considered orbit, one single activity index cannot cover accurately all orbits. Therefore we provide not one, but three indices covering three standard orbits, namely the LEO, MEO (GNSS) and GEO orbits. In addition, space radiative environment effects on spacecraft are greatly dependent on the nature and the energy of the particle fluxes. After consultation and feedback from the space industry stakeholders, we decided to dedicate the indices to the quantification of the internal charging risk.

It is well established that internal charging events are attributed to electrons with energy greater than 400 keV. As a result, deriving a radiation belt index relevant to internal charging can be done using any electron channel above 400 keV . Hence, we propose nowcast and forecast activity indices computed from the following estimated fluxes:

- an internal charging risk at LEO orbit index, computed from daily averaged electron flux with energies $E > 1.2\,\mathrm{MeV}$, computed on the POES spacecraft orbit.

- an internal charging risk at MEO/GNSS orbit index, computed from daily averaged electron flux with energies $E > 0.8\,\mathrm{MeV}$ computed on the GPS spacecraft orbit.

- an internal charging risk at GEO orbit index, computed from daily averaged electron flux with energies $E > 0.8\,\mathrm{MeV}$ computed on the GOES spacecraft orbit.

The activity indices given here are targeted to spacecraft operators and industry end-users. Therefore, they have to be easily understandable by every user, including those who are not radiative space environment experts. This means that we cannot simply provide average flux values, but rather propose a warning system in the form of a three-colour system. To define these periods, we need to define flux threshold values, which must be representative of the risk under consideration and the needs of the consulted stakeholders. Thus, we specified the levels of risk as follows:

- a period is considered "active" (red) if the daily averaged flux is among the 2% strongest historical fluxes values;

- a period is considered "moderate" (yellow) if the daily averaged flux is among the 20% strongest historical fluxes values;

- the remaining periods (among the 80% weakest historical fluxes values) are considered "quiet" (green).

These statistical limits correspond to threshold values of daily averaged flux, deducted from the statistical distributions of daily energetic electron fluxes values at each of the considered orbits. We will see in 3.2 how we obtain these statistical distributions, from long term measurement data.

 3.2 Historical distributions

The statistical distributions serve at the same time to gauge the intensity of the indices in their presumed statistical range and to define the alarm thresholds needed in the warning system and corresponding to the 20% and 2% limits. They were constructed on each one of the three studied orbits, by gathering historical data of daily averaged electron flux as reported in table 1.

Table 1. Time range of the data used to construct the historical distributions for each orbit.

Orbit	in-situ data source	Historical data time range
LEO	POES satellites	1979-2022
MEO	GPS satellites	2000-2022
GEO	GOES satellites	1996-2022

150 After cleaning and cross calibrating the data (see figure 3), a statistical investigation was operated on the historical distributions and led to a statistical distribution of daily fluxes from which complementary cumulative distribution functions or tail distribution were derived on each orbit. They are presented in figure 4 and table 2, with the demarcation of the three levels of risk as previously defined. The data cleaning step is primordial to minimize bias sources in the historical distribution. For instance, it was stated that measurements of relativistic electrons in the inner belt are subject to consequent proton contamination(Claudepierre et al., 2017; Pierrard et al., 2021b)..
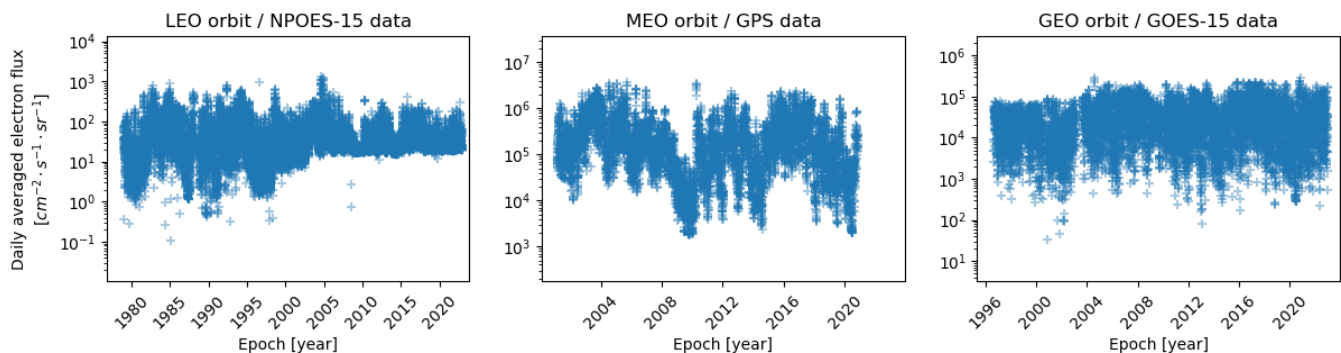
Figure 3. Cleaned and cross-calibrated measurement data used on each orbit to construct the statistical distributions. The energies of the integral flux reported on each panel, are those stated in 3.1 (>1.2 MeV for LEO, >0.8 MeV for MEO and >0.8 for GEO).

155 Using an extensive historical distribution is necessary to ensure that the computed statistical distribution is representative of the full-range of the electron radiation belts' dynamics. Indeed, it is notorious that the solar cycle 24 (2008 – 2019) was a weak cycle during which few intense events were observed. Studies dedicated to the analysis of extreme events in the electron radiation belts (such as Meredith et al., 2015; Bernoux and Maget, 2020) show that no extreme events where witnessed during solar cycle 24, whereas multiple ones where observed during solar cycle 23 (1996 – 2008). Therefore, using a short dataset 160 made of measurements from the solar cycle 24 alone would probably bias the statistical distribution, and more importantly its

tail. The "red" threshold would hence be too low, and fluxes above such threshold could be observed more often than simply 2% of the time, which would trigger too many alerts. Using data from at least two solar cycles to derive the statistical distribution lowers the risk of biasing our results. In addition, results from (Bernoux and Maget, 2020, see e.g. Figure 14 in their paper) suggest that 20-year long electron fluxes data sets starting around 1996 are most probably statistically representative of the distribution of extreme events in the electron radiation belts, which comforts our approach.
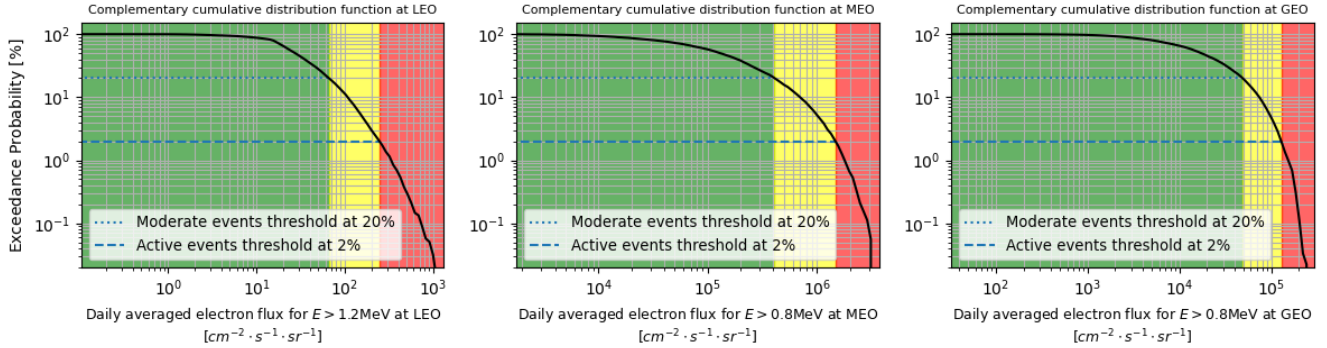
Figure 4. Complementary cumulative distribution functions constructed on each orbit and demarcation of the three activity levels, along with the moderate and active alarm thresholds.

Table 2. Thresholds flux values for the moderate and active alarms expressed in $[\mathrm{cm}^{-2} \cdot \mathrm{s}^{-1} \cdot \mathrm{sr}^{-1}]$.

Orbit	Moderate 20% alarm	Active 2% alarm
LEO	$6.7 \cdot 10^1$	$2.5 \cdot 10^2$
MEO	$4.1 \cdot 10^5$	$1.5 \cdot 10^6$
GEO	$4.8 \cdot 10^4$	$1.3 \cdot 10^5$

4 Example forecast on St Parick's 2015 storm

To assess the reliability of the SafeSpace safety service segment, we decided to test it over the March 2015 period when the CME-driven St-Patrick storm occurred (see figure 5). Precisely, the dynamics of the electron radiation belts during this intense and representative event were replicated with the SafeSpace pipeline. To eliminate the uncertainties linked with the heliospheric part of the modelling pipeline, which are discussed in (Kieokaew et al., 2023), we generated a synthetic forecast from the OMNI2 hourly dataset, by adding a lognormal uncertainty distribution with a standard deviation of 30% on the different solar wind parameters, as well as the Kp index. Salammbô-EnKF assimilated measurements from the Magnetosphere Electron Detector (MAGED) on board the GOES-15 satellite (Hanser, 2011) and the CXD instrument embarked on the GPS-NS54 satellite (Carver et al., 2018), shown in figure 5.
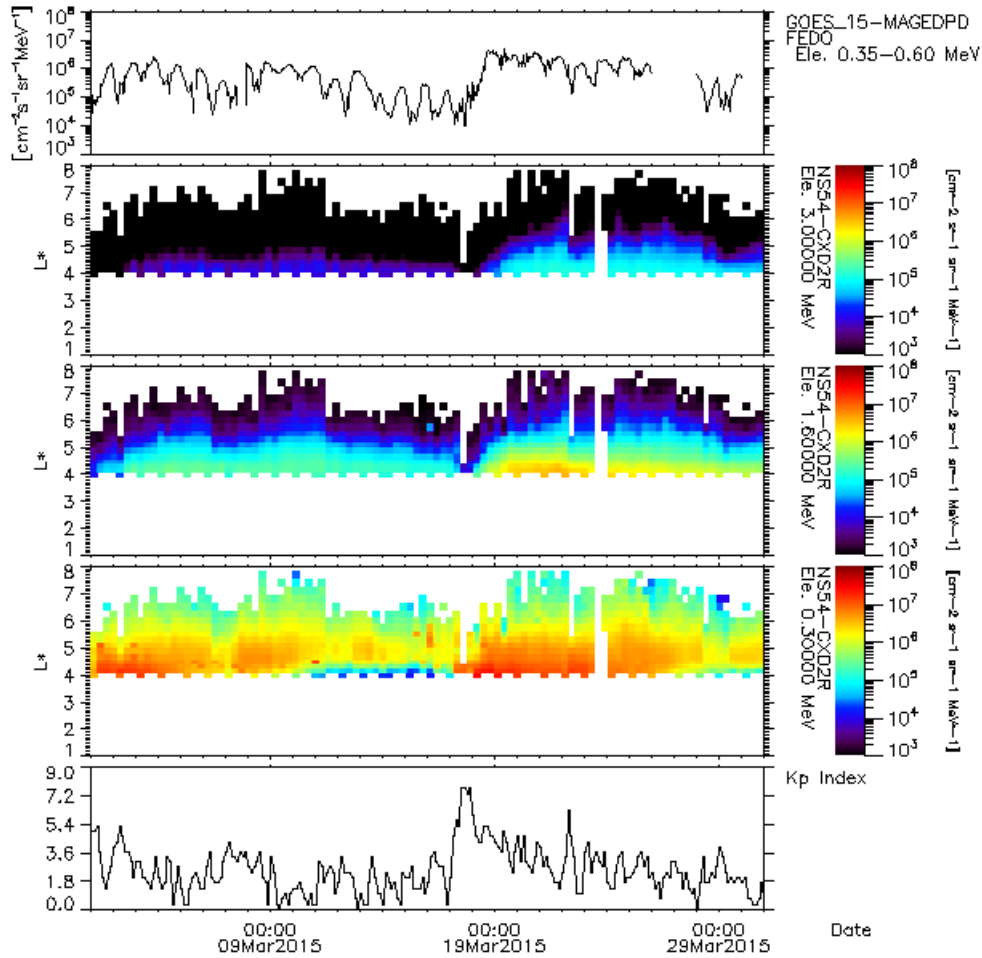
Figure 5. Overview of the assimilated data in Salammbô-EnKF during the March 2015 simulations. From the top to the bottom, GOES-15 MAGED omnidirectional differential flux measurements for electrons from 350 keV to 600 keV, the GPS-NS54 CXD omnidirectional differential flux measurements for electrons at 3 MeV, 1.6 MeV, 300 keV, and the Kp index.

175   In order to measure the nowcast and the forecast capabilities of the SafeSpace pipeline, we conducted two different simulations of the March 2015 period :

    – An ensemble simulation of the whole month with the assimilation of GOES and GPS data during the whole simulation period.

    – An ensemble forecast simulation (no assimilation) of the main phase of the storm (from March 17th to March 20th),
180   considering as an initial state, the assimilated state on March 17th from the previous simulation.

    After generating the appropriate indices for each simulation as defined in section 3, we compared them to reference satellite observations of the same indices on the studied time period and taken from

- NPOES-15/SEM-2 instrument for the LEO orbit (Evans, 2000; Davis, 2007).

- GPS-NS54/CXD detector for the MEO orbit (Carver et al., 2018).

185 - GOES-15/MAGED detector for the GEO orbit (Onsager et al., 1996; Hanser, 2011).

The latter observation-based indices will be considered as perfect references that the simulation-based indices seek to replicate and will serve in the performance evaluation of the SafeSpace warning system. Their relevance to perfectly depict reality will not be treated in this paper.

Figure 6 reports the time evolution of the indices for the nowcast simulation and forecast simulation estimated by the
190 SafeSpace pipeline along with the reference observations of these same indices. On the other hand, tables 3 and 4 report the warning system performance scores for the nowcast and forecast simulations.

We can attest from figure 6, that the nowcast indices perform moderately well with reduced uncertainties most of the time. At the GEO orbit, the SafeSpace nowcast index is very close to the GOES-15 index during the first half of the simulated month. Besides, the latter is often contained in the nowcast index ensemble. From the beginning of the main phase of the storm on
195 March 17th and later, the nowcast index overestimates the GOES-15 index. This is mainly explained by the absence of dropouts in the physical representation of the pipeline and the lack of assimilated data related to highly energetic electrons at the GEO orbit.

At the MEO orbit, the nowcast index also manages to reconstruct accurately the dynamic of the observed GPS index with a slight underestimation (average factor of 0.66). This is expected due to the presence of adequate assimilated data on the studied
200 orbit. At the LEO orbit, the nowcast performs very poorly. Indeed, the nowcast stays almost one order of magnitude below the observed index during the whole simulation. This behavior is explained by two major factors. First, the limited physical description of the inner belt dynamics (still an open subject investigated by the community (Ripoll et al., 2020)), especially the balance between radial diffusion and pitch angle precipitation. Second, the inadequacy of the numerical solver, with the adopted grid refined, to capture the strong gradients observed in the region in the vicinity of the loss cone. One can consider
205 refining the grid to improve the LEO index estimation, but this operation will impose a too intensive computational cost on the daily pipeline.

As regards the forecast results, figure 6 shows encouraging median results yet with relatively large uncertainties. This is particularly the case at MEO and GEO orbits, for which the median forecast matches closely the observed and nowcast indices with definitely larger uncertainties than with the nowcast. This is due to the accurate initial condition and physical
210 representation of the pipeline considered by the forecast before the main phase of the storm on March 2015. At the LEO orbit, the forecast ensemble extends over almost 4 orders of magnitudes and due to the dominant grid refinement constraint, the nowcast and forecast median join and stay way below the observed index.

As regards the warning system performances, table 3 shows that the nowcast simulation offers overall adequate results for the MEO and GEO orbits, related to their associated indices results (as presented in figure 6). For the GEO orbit index, all the
215 "critical" active time alarms were activated on time. However, 7 out of 8 moderate time alarms were missed. This is mainly due to the overestimation of the nowcast GEO index observed after March 21th that also led to the over-activation of the active
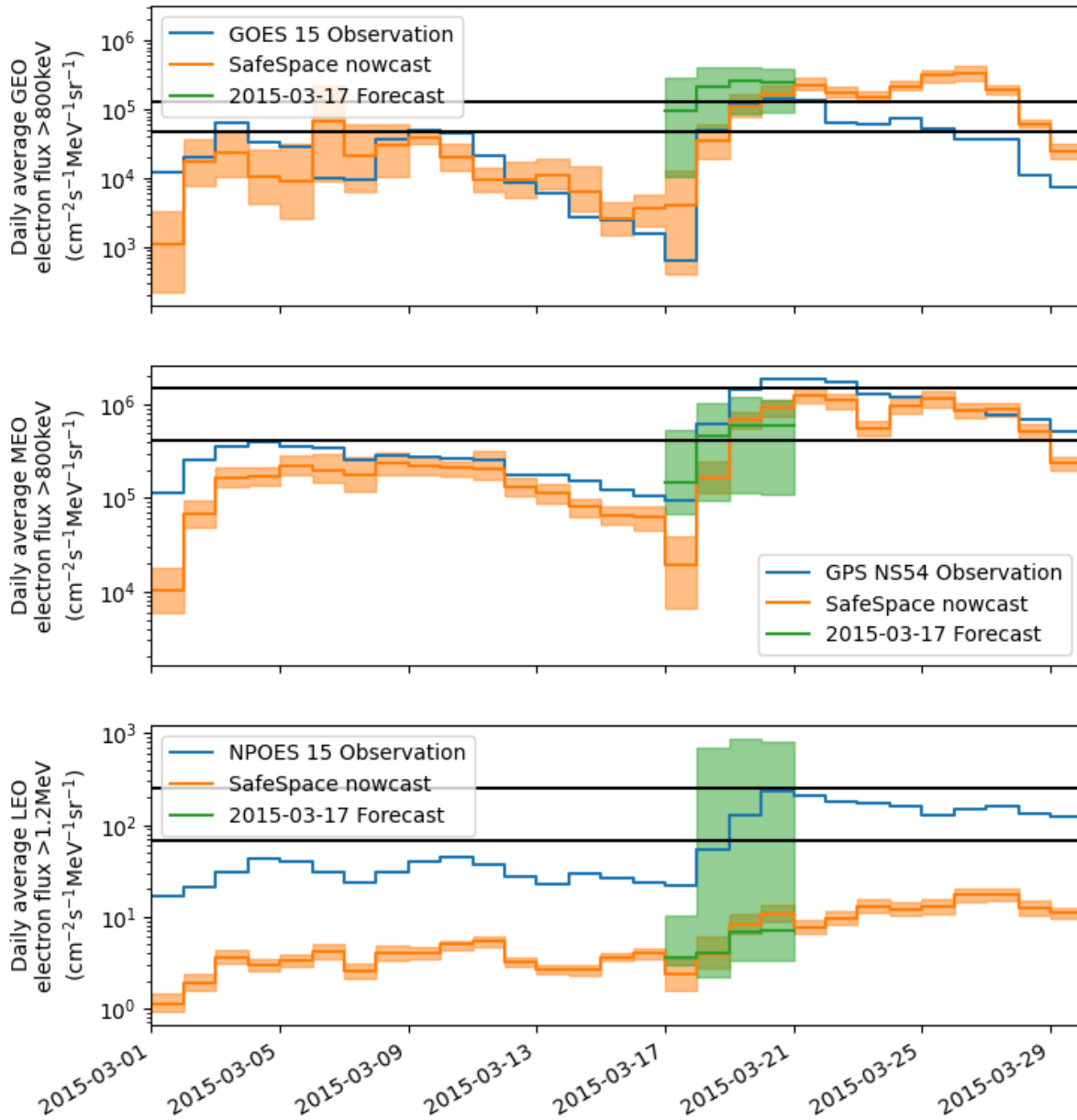
Figure 6. Comparison between the indices derived from the SafeSpace nowcast simulation (in orange for the median, and light orange for the 10th-90th percentile ensemble range), from the SafeSpace forecast simulation (in green for the median, and light green for the 10th-90th percentile ensemble range), with their associated reference observations (from the top to the bottom : GEO, MEO and LEO orbits). The solid black lines represent by increasing order, the moderate and active alarm thresholds.

Table 3. Assessing the performance of the warning system over the one month-long nowcast simulation : Counting of the raised alarms for moderate (M) and active (A) times by observation-based indices (serving as a reference) and by simulation-based indices, along with the missed and false alarms for the latter. Simulation results are formatted as "median (min-max)" with min/max respectively the minimum and maximum values attained by the 10th-90th percentile range of the ensemble simulation.

	GEO index		MEO index		LEO index	
	Observations	Simulation	Observations	Simulation	Observations	Simulation
Raised (M) alarms	8	3 (2-5)	9	10 (10-10)	11	0 (0-0)
Missed (M) alarms	N/A	7 (6-7)	N/A	2 (2-2)	N/A	11 (11-11)
False (M) alarms	N/A	2 (1-3)	N/A	3 (3-3)	N/A	0 (0-0)
Raised (A) alarms	2	8 (8-10)	3	0 (0-0)	0	0 (0-0)
Missed (A) alarms	N/A	0 (0-0)	N/A	3 (3-3)	N/A	0 (0-0)
False (A) alarms	N/A	6 (6-8)	N/A	0 (0-0)	N/A	0 (0-0)

Table 4. Assessing the performance of the warning system over the four-day-long forecast simulation : Counting of the raised alarms for moderate (M) and active (A) times by observation-based indices (serving as a reference) and by simulation-based indices, along with the missed and false alarms for the latter. Simulation results are formatted as "median (min-max)" with min/max respectively the minimum and maximum values attained by the 10th-90th percentile range of the ensemble simulation.

	GEO index		MEO index		LEO index	
	Observations	Simulation	Observations	Simulation	Observations	Simulation
Raised (M) alarms	2	1 (0-3)	2	3 (0-4)	2	0 (0-0)
Missed (M) alarms	N/A	2 (0-2)	N/A	0 (0-2)	N/A	2 (2-2)
False (M) alarms	N/A	1 (0-1)	N/A	1 (0-2)	N/A	0 (0-0)
Raised (A) alarms	1	3 (0-4)	1	0 (0-0)	0	0 (0-3)
Missed (A) alarms	N/A	0 (0-1)	N/A	1 (1-1)	N/A	0 (0-0)
False (A) alarms	N/A	2 (0-3)	N/A	0 (0-0)	N/A	0 (0-3)

time alarms in that same period. Still, the impact of those false positives is contained as the system remains conservative and manages, at least, t discriminate between risky and non-risky states. On the MEO orbit, the system performs better with the moderate time alarm. In fact, only 2 out of 9 moderate alarms were missed and few false positives were raised. However, it missed all the active time alarms raised by the GPS observations. On the LEO orbit, the warning system is totally overtaken and misses all the moderate time alarms due to the substantial underestimation of the LEO index as previously stated. It should be noted from the ensemble's results in 3 that the warning system's performance won't change radically if the whole ensemble range was considered due to the low uncertainty reported on the nowcast indices.

The warning system performance in case of the forecast simulation is adequate and present a similar missed and false alarm distribution as for the nowcast simulation. However, their associated ensemble range is much bigger, due to the higher uncertainties reported on the indices forecast.

5 Conclusions

In this article, we introduced new space industry tailored internal charging dedicated indices, derived from the SafeSpace physical chain of codes that models the outer electron radiation belt dynamic. We presented the different components of the pipeline sun-to-earth, with its heliospheric propagation, inner magnetosphere and safety warning system segments. We focused on the latter part by presenting the dedicated safety indices defined on three standard orbits (LEO,MEO,GEO) and constructed on the daily averaged electron fluxes. We also specified the risk levels for moderate and active alarms in the warning system and showed the data sets used to build the statistical distribution for the indices on each orbit. We discussed the importance of a long and representative historical distribution for a reliable and trust-worthy warning system. Finally, we tested the new implementation on the representative CME-driven St-Patrick storm of 2015. Results show adequate nowcast and forecast of the indices by the SafeSpace pipeline at the GEO and MEO orbits, when compared to observation and a rough estimation of the index at the LEO orbit. Similarly, the warning system performance seem acceptable for the nowcast and forecast simulations. However, to adequately assess the full potential of the SafeSpace pipeline in its current form, the warning system capabilities and its potential sensitivity to geomagnetic events (CME-driven events, or SIR-driven events for example), it is essential to conduct the previous investigation not on only one but on a myriad of pre-selected geomagnetic storms, over a long simulated period.With no doubts, the GEO and MEO indices results demonstrate the relevance of the SafeSpace sun-to-earth pipeline. They also point at the importance of a time-dependent physical description of the outer electron belt which necessarily requires an accurate description of VLF waves and plasma distributions (Ripoll et al., 2023).As a straightforward improvement to the pipeline, we recommend the integration of dropouts modelling (as done in Herrera et al., 2016), after taking care of the Bz parameter estimation. This modification will surely decrease the intensity of the GEO orbit index especially before geomagnetic storms and reduce the number of false active alarms. In front of the poor results of the LEO orbit index, we can only offer as an alternative, the refinement of the grid near the loss cone. However, this operation will introduce an additional computational cost that will degrade the operability of the safety system or even make it obsolete for space weather applications. Therefore, there is no other choice but to take up the challenge of the inner belt electrons modelling. Moreover, we strongly advocate for the adoption by physical codes, of dedicated numerical schemes that would be more suitable to the numerical limitations imposed by the radiation belt modelling (Dahmen et al., 2020) like the strong gradient issue.

Data availability. The SafeSpace safety service website (http://www.safespace-service.eu/) gathers data of the indices computed by the SafeSpace pipeline for the current day (nowcast), the day before and four days ahead in time (forecast) in addition to plots of the indices in colored bar charts, according to their activity state and the location of the indices in their associated cumulative distribution function.The NOAA GOES-15/MAGED omnidirectional electron fluxes data are publicly available at https://satdat.ngdc.noaa.gov/sem/goes/data/full/. The GPS-NS54/CXD electron fluxes are publicly available at https://www.ngdc.noaa.gov/stp/space-weather/satellite-data/satellite-systems/gps/data/ns54.

Author contributions. ND drafted and organized the paper with participation of all co-authors. AB contributed to software developments and the generation of figures. SB and GB revised the manuscript and contributed to the statistical study along with CK, SD, AN, JOF. AB, GB and SB helped with the results interpretation. CK, SD, AN, IS, CP and IAD contributed to the development of the service website.

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